

Tutorial on Control of Non Holonomic Systems by Active Constraints

Franco Rampazzo

▶ To cite this version:

Franco Rampazzo. Tutorial on Control of Non Holonomic Systems by Active Constraints. SADCO Summer School 2011 - Optimal Control, Sep 2011, London, United Kingdom. inria-00629512

HAL Id: inria-00629512 https://inria.hal.science/inria-00629512

Submitted on 6 Oct 2011

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Control of Non Holonomic Systems by Active Constraints

Franco Rampazzo, University of Padova, Italy

SADCO Summer School Imperial College, London, September 5-9, 2011,

FIRST MOVEMENT

—Allegro—

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Introduction

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INVESTIGATE analysis AND geometry RELATED TO THE FOLLOWING PROGRAM:

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INVESTIGATE analysis AND geometry RELATED TO THE FOLLOWING PROGRAM:

Consider a (N + M)-dimensional mechanical system and

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INVESTIGATE analysis AND geometry RELATED TO THE FOLLOWING PROGRAM:

Consider a (N + M)-dimensional mechanical system and let Q be the (N + M)-dimensional configuration manifold, locally parameterized by

$$q=(q^1,\ldots,q^N,q^{N+1},\ldots,q^{N+M})$$

INVESTIGATE analysis AND geometry RELATED TO THE FOLLOWING PROGRAM:

Consider a (N + M)-dimensional mechanical system and let Q be the (N + M)-dimensional configuration manifold, locally parameterized by

$$q = (q^1, \ldots, q^N, q^{N+1}, \ldots, q^{N+M})$$

Assign the "control"

$$u(t) = (u^1, \ldots, u^M) \equiv (q^{N+1}(t), \ldots, q^{N+M}(t))$$

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(i.e. give the evolution of the last *M* coordinates)

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(i.e. give the evolution of the last *M* coordinates)

PROBLEM:

What can we say on the *whole* motion q = q(t)?

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OPTIMIZATION

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Optimization Controllability

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Optimization Controllability Stabilizability

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Optimization Controllability Stabilizability

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Optimization ...

mostly for \mathcal{NC} (=non-centrifugal) systems

CONTROLLABILITY

Stabilizability

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WARNING

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WARNING

In this presentation the words "CENTRIFUGAL" and "NON-CENTRIFUGAL" do not refer to precise definitions.

WARNING

In this presentation the words "CENTRIFUGAL" and "NON-CENTRIFUGAL" do not refer to precise definitions. Rather they have a *pedagogic, allusive* function. For this reason their use will be (hopefully) more acceptable once the presentation is concluded.

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In the conventional applications of Control Theory to Mechanics

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In the conventional applications of Control Theory to Mechanics controls are forces (or powers)

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In the conventional applications of Control Theory to Mechanics controls are forces (or powers)

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In the conventional applications of Control Theory to Mechanics controls are forces (or powers)

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controls are identified with some of the coordinates:

$$u(t) = (q^{N+1}, \ldots, q^{N+M})$$

In the conventional applications of Control Theory to Mechanics controls are forces (or powers)

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controls are identified with some of the coordinates:

$$u(t) = (q^{N+1}, \ldots, q^{N+M})$$

Which is a local way of imposing moving constraints as controls.

In the conventional applications of Control Theory to Mechanics controls are forces (or powers)

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controls are identified with some of the coordinates:

$$u(t) = (q^{N+1}, \ldots, q^{N+M})$$

Which is a local way of imposing moving constraints as controls. We assume that d'Alembert principle is verified, i.e., reaction forces are "orthogonal" to the reduced system Let us begin with

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Let us begin with

SOME EXAMPLES of MECHANICAL SYSTEMS:

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THREE $\underline{CLASS \ C}$ EXAMPLES:

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THREE $\underline{CLASS C}$ EXAMPLES:

(C stands for centrifugal)

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(1C) The angle as control



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(2C) The pendulum with oscillating pivot

Vertically moving pivot



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(3C) The "Roller Racer"



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(The Roller Racer is a well-known toy):





NOW, THREE **CLASS** \mathcal{NC} EXAMPLES:

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NOW, THREE **CLASS** \mathcal{NC} EXAMPLES:

(\mathcal{NC} stands for *non centrifugal*)

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$(1\mathcal{NC})$ The pendulum with length as control



Figure: Length as control

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$(2\mathcal{NC})$ The pendulum with a second pendulum as control



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$(3\mathcal{NC})$ "Shape" as control



q= cylinder's position

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A rigid movement (the shape *u* is unchanged)

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q= cylinder's position

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u= shape

q= cylinder's position

Change of shape (i.e. change of *u*)

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Exposition

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The abstract framework:



The abstract framework:



Given u = u(t), we are interested in the corresponding motion q(t) along the leaves of Λ .

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We shall focus on <u>TWO</u> MAIN KINDS OF CONTROLS *u*:

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(1) <u>RAPIDLY OSCILLATING</u> CONTROLS

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We shall focus on <u>TWO</u> MAIN KINDS OF CONTROLS *u*:

(1) <u>RAPIDLY OSCILLATING</u> CONTROLS

(2) <u>Bounded variation</u> (possibly discontinuous) controls

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Figure: Rapidly oscillating controls

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Figure: Rapidly oscillating controls

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For instance:

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Figure: Rapidly oscillating controls

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For instance: $\epsilon \sin(\mathbf{t}/\epsilon)$



Figure: Control with bounded variation

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Let us consider

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PRODUCE a "FORCE"

PRODUCE a "FORCE"

NOW GUESS:

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PRODUCE a "FORCE"

NOW GUESS:

Which of the previous examples do the job?

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PRODUCE a "FORCE"

NOW GUESS:

Which of the previous examples do the job?The C(=centrifugal)

PRODUCE a "FORCE"

NOW GUESS:

Which of the previous examples do the job?The C(=centrifugal) or the \mathcal{NC} (=non-centrifugal) systems?

(1C) The angle as control



Figure: Oscillations of the angle generate a (centrifugal!) force on the sliding ring

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(2 C) The pendulum with oscillating pivot

Vertically moving pivot



Figure: Oscillations of the pivot stabilize the unstable equilibrium (*Kapiza pendulum*)

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(3C) The "Roller Racer"



Figure: Oscillations of handlebar generate forward motion

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YES,

in the above three examples,

OSCILLATIONS ARE FORCE-GENERATING.

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YES,

in the above three examples, OSCILLATIONS ARE FORCE-GENERATING. Notice that they all are C(=centrifugal)!

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... A SUSPECT ARISES:

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... A SUSPECT ARISES:

Are $\mathcal{NC}(=$ non-centrifugal) systems such that "OSCILLATIONS DO NOT GENERATE FORCES"?

YES,

in the above three examples, OSCILLATIONS ARE FORCE-GENERATING. Notice that they all are C(=centrifugal)!

... A SUSPECT ARISES:

Are $\mathcal{NC}(=$ non-centrifugal) systems such that "OSCILLATIONS DO NOT GENERATE FORCES"?

To feed our intuition let us give one more look to what we have called $\mathcal{NC}(=$ non-centrifugal) systems:

$(1\mathcal{NC})$ The pendulum with length as control



Figure: Almost insensitive to small oscillation of length

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$(2\mathcal{NC})$ The pendulum with a second pendulum as control



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$(3\mathcal{NC})$ "Shape" as control



q= cylinder's position

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A rigid movement (the shape *u* is unchanged)

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Change of shape (i.e. change of *u*)

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q= cylinder's position

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Also in this case,

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q= cylinder's position

Also in this case, the body is almost insensitive to small rapid oscillation of the control (=the shape)

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In fact, in the case of $\mathcal{NC}(=$ non centrifugal) systems, small rapid oscillation of the control-coordinate

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In fact, in the case of $\mathcal{NC}(=$ non centrifugal) systems, small rapid oscillation of the control-coordinate "DO NOT PRODUCE FORCES "

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At this stage, this statement is just an act of faith,

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"DO NOT PRODUCE FORCES "

At this stage, this statement is just an act of *faith*,...not to mention that

In fact, in the case of $\mathcal{NC}(=$ non centrifugal) systems, small rapid oscillation of the control-coordinate

"DO NOT PRODUCE FORCES "

At this stage, this statement is just an act of *faith*,...not to mention that it is **quite tautological**



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NOTICE: the derivative *u* is a measure....

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NOTICE: *the derivative u is a* **measure**....THEY CAN BE IMPLEMENTED ONLY IN "NON CENTRIFUGAL" SYSTEMS



NOTICE: the derivative \dot{u} is a measure....THEY CAN BE IMPLEMENTED ONLY IN "NON CENTRIFUGAL" SYSTEMS (WHERE THEY ARE INTERESTING FOR OPTIMIZATION PURPOSES)

Development

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IT IS TIME FOR SOME MATHEMATICS:

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IT IS TIME FOR SOME MATHEMATICS:

Let us forget mechanics for a while

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IT IS TIME FOR SOME MATHEMATICS:

Let us forget mechanics for a while

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and let us consider a particular class of control systems:

We are interested in control equations having the form

$$\dot{x} = f(x, u) + \sum_{\alpha=1}^{m} g_{\alpha}(x, u) \dot{u}_{\alpha} + \sum_{\alpha, \beta=1}^{m} h_{\alpha\beta}(x, u) \dot{u}_{\alpha} \dot{u}_{\beta}$$

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Notice: the actual "controls" are the derivatives u

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Notice: the actual "controls" are the derivatives \dot{u} As a matter of fact, , we can even neglect the dependence on u just by adding variables

$$\dot{x}^{n+\alpha} = \dot{u}^{\alpha}$$

so obtaining
ODE's depending quadratically on the control derivative \dot{u}

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Notice: the actual "controls" are the derivatives \dot{u} As a matter of fact, , we can even neglect the dependence on u just by adding variables

$$\dot{x}^{n+\alpha} = \dot{u}^{\alpha}$$

so obtaining

$$\dot{x} = f(x) + \sum_{lpha}^{m} g_{lpha}(x) \dot{u}_{lpha} + \sum_{lpha,eta=1}^{m} h_{lphaeta}(x) \dot{u}_{lpha} \dot{u}_{eta}$$

Flashforward to Mechanics:

If we set

$$(q^1,\ldots,q^N,p^1,\ldots,p^n)=x$$

in the mechanical examples above, we obtain control equations of the form

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In view of mechanical applications, we distinguish among:

• The trivial case : $g_{\alpha} = 0$ $h_{\alpha\beta} = 0$

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In view of mechanical applications, we distinguish among:

• The trivial case : $g_{\alpha} = 0$ $h_{\alpha\beta} = 0$ (non-centrifugal!)

$$(q^1,\ldots,q^N,p^1,\ldots,p^n)=x$$

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- The trivial case : $g_{\alpha} = 0$ $h_{\alpha\beta} = 0$ (non-centrifugal!)
- The affine case : $h_{\alpha\beta} = 0$

$$(q^1,\ldots,q^N,p^1,\ldots,p^n)=x$$

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- The trivial case : $g_{\alpha} = 0$ $h_{\alpha\beta} = 0$ (non-centrifugal!)
- The affine case : $h_{\alpha\beta} = 0$ (non-centrifugal!)
- The general case : $h_{\alpha\beta} \neq 0$

$$(q^1,\ldots,q^N,p^1,\ldots,p^n)=x$$

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- The trivial case : $g_{\alpha} = 0$ $h_{\alpha\beta} = 0$ (non-centrifugal!)
- The affine case : $h_{\alpha\beta} = 0$ (non-centrifugal!)
- The general case : $h_{\alpha\beta} \neq 0$... (centrifugal!)

 $u(\cdot)\mapsto x[u(\cdot)](\cdot)$



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 $u(\cdot)\mapsto x[u(\cdot)](\cdot)$



 $u(\cdot)\mapsto x[u(\cdot)](\cdot)$



Of course its *regularity* depends on how the right hand of dynamic equations depend on \dot{u} :

 $u(\cdot)\mapsto x[u(\cdot)](\cdot)$



Of course its *regularity* depends on how the right hand of dynamic equations depend on \dot{u} : linearly?...

 $u(\cdot)\mapsto x[u(\cdot)](\cdot)$



Of course its *regularity* depends on how the right hand of dynamic equations depend on \dot{u} : linearly?... ($\dot{x} = f(x) + \sum_{\alpha=1}^{m} g_{\alpha}(x) \dot{u}^{\alpha}$)

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 $u(\cdot)\mapsto x[u(\cdot)](\cdot)$



Of course its *regularity* depends on how the right hand of dynamic equations depend on \dot{u} : linearly?... ($\dot{x} = f(x) + \sum_{\alpha=1}^{m} g_{\alpha}(x) \dot{u}^{\alpha}$) or

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 $u(\cdot)\mapsto x[u(\cdot)](\cdot)$



Of course its *regularity* depends on how the right hand of dynamic equations depend on \dot{u} : linearly?... ($\dot{x} = f(x) + \sum_{\alpha=1}^{m} g_{\alpha}(x) \dot{u}^{\alpha}$) or quadratically?...

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 $u(\cdot)\mapsto x[u(\cdot)](\cdot)$



Of course its *regularity* depends on how the right hand of dynamic equations depend on \dot{u} : linearly?... ($\dot{x} = f(x) + \sum_{\alpha=1}^{m} g_{\alpha}(x)\dot{u}^{\alpha}$) or quadratically?... ($\dot{x} = f(x) + \sum_{\alpha=1}^{m} g_{\alpha}(x)\dot{u}^{\alpha} + \sum_{\alpha,\beta=1}^{m} h_{\alpha\beta}(x)\dot{u}^{\alpha}\dot{u}^{\beta}$)

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(1) Can discontinuous controls $u(\cdot)$ be implemented?

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(1) Can discontinuous controls $u(\cdot)$ be implemented? *Comment*: Question (1) is important in optimization with no coercivity

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- (1) Can discontinuous controls u(·) be implemented?
 Comment: Question (1) is important in optimization with no coercivity
- (2) What happens with highly oscillatory u(·)?
 Comment: Question (2) is important for controllability and stabilizability questions

- (1) Can discontinuous controls u(·) be implemented?
 Comment: Question (1) is important in optimization with no coercivity
- (2) What happens with highly oscillatory u(·)?
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IN THIS TUTORIAL WE SHALL BE MAINLY CONCERNED WITH (2)

- (1) Can discontinuous controls u(·) be implemented?
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but we are also recalling crucial issues concerning (1),

- (1) Can discontinuous controls u(·) be implemented?
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(i) its geometrical meaning,
(ii) the vast related literature,

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(i) its geometrical meaning,
(ii) the vast related literature,
(iii) its relation with (2)

$$\dot{x} = f(x) + \sum_{lpha}^{m} g_{lpha}(x) \dot{u}^{lpha} + \sum_{lpha,eta=1}^{m} h_{lphaeta}(x) \dot{u}^{lpha} \dot{u}^{eta}$$

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(2) Despite (1), limits of solutions can well be discontinuous.

(see A. Bressan - F.R. (1999) SIAM Control Opt. for the closure of the set of solutions when $\int_0^T |u|^2 dt \le K$, in the case of scalar u)

Definition of STABILIZABILITY

• The control system

(Eq)
$$\dot{x} = f(x) + \sum_{\alpha}^{m} g_{\alpha}(x) \dot{u}_{\alpha} + \sum_{\alpha,\beta=1}^{m} h_{\alpha\beta}(x) \dot{u}_{\alpha} \dot{u}_{\beta}$$

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if, $\forall \varepsilon > 0 \ \exists \delta > 0 \ \text{such that} : \forall \ \hat{x} \in B(\bar{x}, \delta) \ \text{there exists a}$ piecewise smooth control function $t \mapsto u(t) = (u_1, \dots, u_m)(t)$ such that

$$x(t,u) \in B(\bar{x},\varepsilon) \quad \forall t \geq 0$$

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The differential inclusion

The control system

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$$\mathcal{F}(x) \doteq \overline{co} \Big\{ f(x) + \sum_{\alpha=1}^{m} g_{\alpha}(x) w_{\alpha} + \sum_{\alpha,\beta=1}^{m} h_{\alpha\beta}(x) w_{\alpha} w_{\beta}; \quad (w_{1},\ldots,w_{m}) \in \mathbb{R}$$

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• Notice that it is the differential inclusion associated with

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where there is no \dot{u} -linear term;

• $\mathcal{G}(x) \subset \mathcal{F}(x)$

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The point \bar{x} is weakly stable for the differential inclusion

$$\frac{dx}{dt} \in \mathcal{F}(x)$$

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If, in addition,

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the differential inclusion is called weakly asymptoticly stable (at \bar{x}).

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Definition. A scalar function V is a weak Lyapunov function if

- V is continuous on \mathcal{N} , and continuously differentiable on $\mathcal{N}\setminus\{\bar{x}\}.$
- $V(\bar{x}) = 0$ while V(x) > 0 for all $x \neq \bar{x}$.
- For each $\delta > 0$ sufficiently small, the sublevel set $\{x; V(x) \le \delta\}$ is compact.
- At each $x \neq \bar{x}$ one has

$$\inf_{y\in F(x)}\nabla V(x)\cdot y\leq 0\,.$$

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Theorem Assume that the differential inclusion

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Theorem Assume that the differential inclusion

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can be stabilized at \bar{x} .

Corollary.

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Corollary. Assume that the "simmetrized" differential inclusion

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Corollary. Assume that the "simmetrized" differential inclusion

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Then the control system

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$$\frac{dx}{dt}\in\mathcal{G}(x)$$

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without the (= \dot{u} -linear) term $\sum_{\alpha=1}^{m} g_{\alpha}(x)\dot{u}_{\alpha}$,

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while $\mathcal{F}(x)$ is the convexification with the (=*i*-linear) term $\sum_{\alpha=1}^{m} g_{\alpha}(x)\dot{u}_{\alpha}$. (Intuitively: Rapid oscillations of *u* cancel out the *u*-linear term.)

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A main technical device: reparameterization

NOTICE:

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$$\sigma(t) \doteq \frac{\int_0^t (1+|\dot{u}|^2) d\tau}{T+\|\dot{u}\|_2^2}$$

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 $\phi_\alpha(s) \doteq u_\alpha(t(s)) \quad \alpha = 1..., m, \quad v_\alpha \doteq \frac{\phi_\alpha}{ds}$

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Setting y(s) = x(t(s)), the original control system

$$\dot{x} = f(x) + \sum_{\alpha=1}^{m} g_{\alpha}(x) \dot{u}_{\alpha} + \sum_{\alpha,\beta=1}^{m} h_{\alpha\beta}(x) \dot{u}_{\alpha} \dot{u}_{\beta}$$

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is turned into the reparameterized system

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Correspondingly, one has the **REPARAMETERIZED DIFFERENTIAL INCLUSION:**

$$\frac{dy}{ds} \in \mathbf{F}(y)$$

$$\mathbf{F}(y) \doteq \overline{co} \left\{ f(y)v_0^2 + \sum_{\alpha=1}^m g_\alpha(y)v_0v_\alpha + \sum_{\alpha,\beta=1}^m h_{\alpha,\beta}(y)v_\alpha v_\beta; \\ v_0 \in [0,1], v_0^2 + \dots + v_m^2 = 1 \right\}.$$

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$$\dot{x} = \dot{u} + (\dot{u})^2$$
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Consider the controls

$$u_n(t) \doteq \frac{\sin(nt)}{n}$$
 so $\dot{u}_n(t) = \cos(nt)$

Therefore, for each $\tau > 0$,

$$x_n(\tau) = x_n(\tau) - x_n(0) = \int_0^{\tau} \cos(nt) dt + \int_0^{\tau} \cos^2(nt) dt =$$

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$$\frac{1}{n} \frac{n\tau}{2\pi} \int_0^{2\pi} \cos(s) \, ds + \frac{1}{n} \frac{n\tau}{2\pi} \int_0^{2\pi} \cos^2(s) \, ds + O(1/n) = \frac{\tau}{2} + O(1/n)$$

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Therefore, for each $\tau > 0$,

$$x_n(\tau) = x_n(\tau) - x_n(0) = \int_0^{\tau} \cos(nt) dt + \int_0^{\tau} \cos^2(nt) dt =$$

$$\frac{1}{n} \int_0^{n\tau} \cos(s) \, ds + \frac{1}{n} \int_0^{n\tau} \cos^2(s) \, ds =$$

$$\frac{1}{n}\frac{n\tau}{2\pi}\int_0^{2\pi}\cos(s)\,ds + \frac{1}{n}\frac{n\tau}{2\pi}\int_0^{2\pi}\cos^2(s)\,ds + O(1/n) = \frac{\tau}{2} + O(1/n)$$

Hence

$$x_n(\tau)-x_n(0)
ightarrow rac{ au}{2}.$$

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Notice: on each interval $[0, \tau]$, as *n* tends to infinity

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Notice: on each interval $[0, \tau]$, as *n* tends to infinity

(1) the integral of the term \dot{u}_n cancels out.

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Notice: on each interval $[0, \tau]$, as *n* tends to infinity

(1) the integral of the term \dot{u}_n cancels out.

(2) the term \dot{u}_n^2 has the same effect as the constant $\frac{1}{2}$, i.e. it produces the output $t \to \frac{t}{2}$

Recapitulation

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BACK TO MECHANICS

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• assigning the control u = u(t) is nothing but adding the new holonomic constraint

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u = u(t)... to the original system.

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• As is well-known in order to determine equations one needs to specify how the control constraint u = u(t) is physically implemented :

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u = u(t)... to the original system.

• As is well-known in order to determine equations one needs to specify how the control constraint u = u(t) is physically implemented :

Here we assume **D'Alembert Hypothesis**, namely, *the* work *of constraint reactions* **is equal to zero**.

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The Kinetic Energy

$$\mathcal{T} = \mathcal{T}[(q, u)(\dot{q}, \dot{u})]$$

is a quadratic form in (\dot{q}, \dot{u}) ,

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$$\mathcal{T} = \mathcal{T}[(q, u)(\dot{q}, \dot{u})]$$

$$\mathcal{T} = (\dot{q}, \dot{u}) \cdot \mathbf{A} \cdot (\dot{q}, \dot{u})^t \qquad \mathbf{A} = (a_{r,s})_{r,s=1,\dots,N+M}$$

(A is the so-called *kinetic matrix*)

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(A is the so-called *kinetic matrix*) **The Hamiltonian** is nothing but its Legendre transform:

 $\mathcal{H}[(q,u)(p,\pi)] \doteq \mathcal{T}^*$

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(A is the so-called *kinetic matrix*) **The Hamiltonian** is nothing but its Legendre transform:

$$\mathcal{H}[(q,u)(p,\pi)] \doteq \mathcal{T}^*$$

In particular \mathcal{H} is quadratic in the momenta (p, π)

$$\mathcal{H} = (p, \pi) \mathbf{A}^{-1} (p, \pi)^t \qquad \mathbf{A}^{-1} = (a^{r,s})_{r,s=1,...,N+M}$$

$$\begin{cases} \dot{q}_{i} = \frac{\partial \mathcal{H}}{\partial p_{i}} \\ \dot{u}_{\alpha} = \frac{\partial \mathcal{H}}{\partial \pi_{\alpha}} \\ \dot{p}_{i} = -\frac{\partial \mathcal{H}}{\partial q_{i}} + \mathcal{F}_{i} \\ \dot{\pi}_{\alpha} = -\frac{\partial \mathcal{H}}{\partial u_{\alpha}} + \mathcal{F}_{u_{\alpha}} \end{cases}$$

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Now there is an isomorphism between the momentum (p, π) and the velocity (\dot{q}, \dot{u}) .

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$$\begin{aligned} \dot{\mathbf{q}}_{\mathbf{i}} &= \frac{\partial \mathcal{H}}{\partial \mathbf{p}_{\mathbf{i}}} \\ \dot{\boldsymbol{\mu}}_{\alpha} &= \frac{\partial \mathcal{H}}{\partial \pi_{\alpha}} \\ \dot{\mathbf{p}}_{\mathbf{i}} &= -\frac{\partial \mathcal{H}}{\partial \mathbf{q}_{\mathbf{i}}} + \mathcal{F}_{\mathbf{i}} \\ \dot{\boldsymbol{\pi}}_{\alpha} &= -\frac{\partial \mathcal{H}}{\partial \boldsymbol{\mu}_{\alpha}} + \mathcal{F}_{\boldsymbol{\mu}_{\alpha}} \end{aligned}$$

Now there is an isomorphism between the momentum (p, π) and the velocity (\dot{q}, \dot{u}) . By partially inverting this isomorphism we obtain

 π as a linear combination of (p, \dot{u}) . Therefore, we get the control equations for (q, p):

$$\begin{pmatrix} \dot{q} \\ \dot{p} \end{pmatrix} = \overbrace{\begin{pmatrix} 0 \\ F^{u(\cdot)} \end{pmatrix}}^{f} + \phi + \sum_{\alpha=1}^{m} g_{\alpha} \dot{u}_{\alpha} + \sum_{\alpha,\beta=1}^{M} h_{\alpha,\beta} \dot{u}_{\alpha} \dot{u}_{\beta}$$

with suitable vector fields $f(q, p, u), g_{\alpha}(q, p, u), h_{\alpha,\beta}(q, p, u)$ determined by the Kinetic Energy and the applied forces.

$$\begin{pmatrix} \dot{q} \\ \dot{p} \end{pmatrix} = \overbrace{\begin{pmatrix} 0 \\ F^{u(\cdot)} \end{pmatrix}}^{f} + \phi + \sum_{\alpha=1}^{m} g_{\alpha} \dot{u}_{\alpha} + \sum_{\alpha,\beta=1}^{M} h_{\alpha,\beta} \dot{u}_{\alpha} \dot{u}_{\beta}$$

with suitable vector fields $f(q, p, u), g_{\alpha}(q, p, u), h_{\alpha,\beta}(q, p, u)$ determined by the Kinetic Energy and the applied forces. **Notice:**

$$\begin{pmatrix} \dot{q} \\ \dot{p} \end{pmatrix} = \overbrace{\begin{pmatrix} 0 \\ F^{u(\cdot)} \end{pmatrix}}^{f} + \phi + \sum_{\alpha=1}^{m} g_{\alpha} \dot{u}_{\alpha} + \sum_{\alpha,\beta=1}^{M} h_{\alpha,\beta} \dot{u}_{\alpha} \dot{u}_{\beta}$$

with suitable vector fields f(q, p, u), $g_{\alpha}(q, p, u)$, $h_{\alpha,\beta}(q, p, u)$ determined by the Kinetic Energy and the applied forces. **Notice:** Setting x = (q, p) the equations take the same form as the control equations we have considered above, namely

$$\begin{pmatrix} \dot{q} \\ \dot{p} \end{pmatrix} = \overbrace{\begin{pmatrix} 0 \\ F^{u(\cdot)} \end{pmatrix}}^{f} + \phi + \sum_{\alpha=1}^{m} g_{\alpha} \dot{u}_{\alpha} + \sum_{\alpha,\beta=1}^{M} h_{\alpha,\beta} \dot{u}_{\alpha} \dot{u}_{\beta}$$

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$$\dot{x} = f(x) + \sum_{\alpha=1}^{M} g_{\alpha}(x) \dot{u}_{\alpha} + \sum_{\alpha,\beta=1}^{M} h_{\alpha\beta}(x) \dot{u}_{\alpha} \dot{u}_{\beta}$$

The quadratic term:

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The quadratic term:

$$\begin{pmatrix} \dot{q} \\ \dot{p} \end{pmatrix} = \begin{pmatrix} 0 \\ F^{u(\cdot)} \end{pmatrix} + \phi + \sum_{\alpha=1}^{M} g_{\alpha} \dot{u}_{\alpha} + \sum_{\alpha,\beta=1}^{M} h_{\alpha,\beta} \dot{u}_{\alpha} \dot{u}_{\beta}$$

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$$h_{\alpha,\beta} = \begin{pmatrix} 0 \\ \cdot \\ 0 \\ 0 \\ \frac{\partial e_{\alpha,\beta}}{\partial q^{1}} \\ \frac{\partial e_{\alpha,\beta}}{\partial q^{N}} \end{pmatrix} \qquad \text{where}$$

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The quadratic term:

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$$h_{\alpha,\beta} = \begin{pmatrix} 0 \\ \cdot \\ 0 \\ \frac{\partial e_{\alpha,\beta}}{\partial q^{1}} \\ \cdot \\ \frac{\partial e_{\alpha,\beta}}{\partial q^{N}} \end{pmatrix} \quad \text{where}$$
$$E = \begin{pmatrix} e_{11}, \dots, e_{1M} \\ \dots \\ e_{M1}, \dots, e_{MM} \end{pmatrix} \doteq \begin{pmatrix} a^{N+1,N+1}, \dots, a^{N+1,N+M} \\ \dots \\ a^{N+M,1}, \dots, a^{N+M,N+M} \end{pmatrix}^{-1}$$

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The quadratic part $h_{\alpha,\beta}\dot{u}_{\alpha}\dot{u}_{\beta}$ is zero

IF AND ONLY IF

the matrix

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does not depend on q.

Let us assume that the force Γ is a conservative, and let U = U(q, u) be a potential of Γ . Let us fix \bar{u} , and let us assume that there exist positive real coefficients $\lambda_1, \ldots, \lambda_k$ verifying $\sum_{r=1}^k \lambda_r = 1$, and vectors w_1, \ldots, w_s such that the effective potential

$$\widehat{U}(q) \doteq U(\overline{u},q) - rac{1}{2} \sum_{r=1}^{k} \lambda_r \sum_{lpha,eta=1}^{m} e_{lpha,eta}(q,\overline{u}) w_r^{lpha} w_r^{eta}$$

has a strict minimum at $q = \bar{q}$.

Then, the control mechanical system can be stabilized to (\bar{q}, \bar{u}) .

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Idea of the proof: the choice of the λ_r, w_r selects from the corresponding differential inclusion
 x ∈ G a conservative mechanical system with potential energy

equal to \widehat{U}

• If \bar{q} is unstable for the frozen control $u = \bar{u}$, a necessary condition for stabilizability is that the matrix $e_{\alpha,\beta}$ be q-dependent.



Is the state q=0 stabilizable by a vibrational u ?

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$$\begin{cases} \dot{q} = p + (\sin q)\dot{u} \\ \dot{p} = -\frac{\partial U}{\partial q} - p\cos q\dot{u} - (\sin q\cos q)(\dot{u})^2 \,, \end{cases}$$

where $U(q, c) \doteq \cos q$.

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The kinetic matrix:

$$\mathbf{A} = \begin{pmatrix} 1 & -\sin q \\ \\ -\sin q & 2 \end{pmatrix}$$

...so that the 1×1 matrix E is

$$E=e_{11}=1+\cos^2 q$$

Notice: The necessary condition (i.e. "E depends on q") is verified; Moreover: the effective potential

$$U_{\{1\}\{w\}} = \cos q - \frac{1}{2}(1 + (\cos q)^2)w^2.$$

has a minimum at q = 0 as soon as $w^2 \ge 1$. Therefore: The system is vibrationally stabilizable at 0.

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Coda

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In order to give the answer one need not even know the equation of motion...

$$\mathbf{A} = \left(egin{array}{cc} u^2 & 0 \ & \ & \ 0 & 1 \end{array}
ight) \, ,$$

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$$\mathbf{A}=\left(egin{array}{cc} u^2 & 0 \ & \ & \ 0 & 1 \end{array}
ight)\,,$$

so that

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$$oldsymbol{\mathsf{A}}=\left(egin{array}{cc} u^2 & 0 \ & \ & \ 0 & 1 \end{array}
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so that the 1×1 matrix *E* is equal to 1...

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$$\mathbf{A}=\left(egin{array}{cc} u^2 & 0 \ & \ & \ 0 & 1 \end{array}
ight)\,,$$

so that **the** 1×1 **matrix** *E* **is equal to** 1... In particular, *E* **is** <u>independent</u> of *q* !

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$$\mathbf{A}=\left(egin{array}{cc} u^2 & 0 \ & \ & \ 0 & 1 \end{array}
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the 1×1 matrix E is equal to 1...

In particular,

E is independent of q !

i.e. the necessary condition for vibrational stabilizability is not satisfied

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In particular,

E is independent of q !

i.e. the necessary condition for vibrational stabilizability is <u>not satisfied</u>

in other words the control equations are affine in \dot{u} .

Second Movement

—Adagio—

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$$\dot{\mathbf{x}} = \mathbf{f}(\mathbf{x}) + \sum_{lpha = \mathbf{1}}^{\mathbf{m}} \mathbf{g}_{lpha}(\mathbf{x}) \dot{\mathbf{u}}^{lpha}$$

 $(h_{\alpha,\beta}=0)$

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$$\dot{\mathbf{x}} = \mathbf{f}(\mathbf{x}) + \sum_{lpha = \mathbf{1}}^{\mathbf{m}} \mathbf{g}_{lpha}(\mathbf{x}) \dot{\mathbf{u}}^{lpha}$$

 $(h_{\alpha,\beta} = 0)$ These systems are called "impulsive", because one might want to implement discontinuous controls u.

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minimize $\psi(\mathbf{x}(\mathbf{T}))$,

because minimizing sequences u_n could exist with larger and larger derivatives $\dot{\mathbf{u}}_n$)

FACTS:

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(i) A distributional approach does not work.

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FACTS:

- (i) A distributional approach does not work.
- (ii) A robust definition of solution can be given by a density argument (on the graphs)

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FACTS:

- (i) A distributional approach does not work.
- (ii) A robust definition of solution can be given by a density argument (on the graphs)
- (iii) Such a definition allows for jumps of *u*.

WHY?

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WHY?

Because

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WHY?

Because a distributional (=in measure) approach cannot work in a nonlinear equation...

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WHY?

Because a distributional (=in measure) approach cannot work in a nonlinear equation... **More intuitively**:

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WHY?

Because a distributional (=in measure) approach cannot work in a nonlinear equation... **More intuitively**: consider the simple equation

 $\dot{x}(t) = x(t) \dot{u}(t)$

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Because a distributional (=in measure) approach cannot work in a nonlinear equation... **More intuitively**: consider the simple equation

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and take \mathbf{u} to be the Heaveside function, so $\dot{\mathbf{u}}$ is the Dirac measure $\delta_{\mathbf{\bar{t}}}$.

u(t)

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Because a distributional (=in measure) approach cannot work in a nonlinear equation... **More intuitively**: consider the simple equation

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and take **u** to be the Heaveside function, so $\dot{\mathbf{u}}$ is the Dirac measure $\delta_{\overline{\mathbf{t}}}$. Which value should we assign to x at the jump time \overline{t} ?...Simply, there is no reasonable a priori choice

u(t)
$$\dot{\mathbf{x}} = \mathbf{f}(\mathbf{x}) + \sum_{lpha = \mathbf{1}}^{\mathbf{m}} \mathbf{g}_{lpha}(\mathbf{x}) \dot{\mathbf{u}}^{lpha}$$

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$$\dot{\mathbf{x}} = \mathbf{f}(\mathbf{x}) + \sum_{lpha = \mathbf{1}}^{\mathbf{m}} \mathbf{g}_{lpha}(\mathbf{x}) \dot{\mathbf{u}}^{lpha}$$

(a) Approximate a L^1 control u by means of smooth u_n ;

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$$\dot{\mathsf{x}} = \mathsf{f}(\mathsf{x}) + \sum_{lpha = \mathbf{1}}^{\mathsf{m}} \mathsf{g}_{lpha}(\mathsf{x}) \dot{\mathsf{u}}^{lpha}$$

(a) Approximate a L¹ control u by means of smooth u_n;
(b) Consider the solutions x_n corresponding to u_n;

$$\dot{\mathsf{x}} = \mathsf{f}(\mathsf{x}) + \sum_{lpha = \mathbf{1}}^{\mathsf{m}} \mathbf{g}_{lpha}(\mathsf{x}) \dot{\mathsf{u}}^{lpha}$$

- (a) Approximate a L^1 control u by means of smooth u_n ;
- (b) Consider the solutions x_n corresponding to u_n ;
- (c) Let $n \to \infty$ and verify that x_n tends to a unique function x;

$$\dot{\mathsf{x}} = \mathsf{f}(\mathsf{x}) + \sum_{lpha = \mathbf{1}}^{\mathsf{m}} \mathsf{g}_{lpha}(\mathsf{x}) \dot{\mathsf{u}}^{lpha}$$

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- (d) Define x as the solution corresponding to u;

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- (a) Approximate a L^1 control u by means of smooth u_n ;
- (b) Consider the solutions x_n corresponding to u_n ;
- (c) Let $n \to \infty$ and verify that x_n tends to a unique function x;
- (d) Define x as the solution corresponding to u;



$$\dot{\mathbf{x}} = \mathbf{f}(\mathbf{x}) + \sum_{lpha = \mathbf{1}}^{\mathbf{m}} \mathbf{g}_{lpha}(\mathbf{x}) \dot{\mathbf{u}}^{lpha}$$

IT WORKS!

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$$\dot{\mathbf{x}} = \mathbf{f}(\mathbf{x}) + \sum_{lpha = \mathbf{1}}^{\mathbf{m}} \mathbf{g}_{lpha}(\mathbf{x}) \dot{\mathbf{u}}^{lpha}$$

IT WORKS! BUT

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More precisely:

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$$[\mathbf{g}_{\alpha},\mathbf{g}_{\beta}] \doteq \mathbf{D}\mathbf{g}_{\beta} \cdot \mathbf{g}_{\alpha} - \mathbf{D}\mathbf{g}_{\alpha} \cdot \mathbf{g}_{\beta} \neq \mathbf{0} \qquad \alpha \neq \beta$$

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Roughly speaking:at a jump of *u* one has to assign the *trajectory of the jump*

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NOTICE: All graph completions are equivalent if and only if the vector fields g_{α} commute, namely:

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The deep geometrical reason for this is the **Simultaneous Flow-box Theorem:**

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The deep geometrical reason for this is the **Simultaneous Flow-box Theorem:**

All vector fields g_{α} commute IF AND ONLY IF

there exists a system of coordinates where they all are constant

An important property of affine systems

$$\dot{\mathbf{x}} = \mathbf{f}(\mathbf{x}) + \sum_{lpha = \mathbf{1}}^{\mathbf{m}} \mathbf{g}_{lpha}(\mathbf{x}) \dot{\mathbf{u}}^{lpha}$$

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The input-output map

$$\phi: u(\cdot) \to \Phi(u) = x(\cdot)$$

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In any case, do not forget it, thinking in particular to the non-stabilazable double pendulum ... wait for the *Third Movement*

An incomplete list of authors who have given contributions on the subject of impulsive system:

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H. Sussmann, A. Bressan, A.Bressan-F.R, M.Motta- F.R., B. Miller, Dal Maso-F.R., C. Sartori-F.R., J. Dorroh-G.Ferreyra, A. Sarychev, R. Vinter-G. Silva, F.Lobo Pereira, P. Mason, P. Wolenski-S.Zabic, H. Zidani...

THIRD MOVEMENT (Scherzo)

—Con moto—

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As before, we consider a mechanical systems where the last M coordinates $(x^{N+1}, \ldots, x^{N+M}) = (u^1, \ldots, u^M)$ are regarded as a control.

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QUESTION:

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As before, we consider a mechanical systems where the last M coordinates $(x^{N+1}, \ldots, x^{N+M}) = (u^1, \ldots, u^M)$ are regarded as a control.

QUESTION:

Is the fact that the control system is affine in *u* —generally it is quadratic— related to some differential geometric property?



$$\mathcal{T} = \mathcal{T}[(q, u)(\dot{q}, \dot{u})]$$

is a quadratic form in (\dot{q}, \dot{u}) ,

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(A is the so-called *kinetic matrix*)

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(A is the so-called *kinetic matrix*) **The Hamiltonian** is nothing but its Legendre transform:

$$\mathcal{H}[(q,u)(p,\pi)]\doteq\mathcal{T}^*$$

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(A is the so-called *kinetic matrix*) **The Hamiltonian** is nothing but its Legendre transform:

$$\mathcal{H}[(q,u)(p,\pi)] \doteq \mathcal{T}^*$$

In particular \mathcal{H} is quadratic in the momenta (p,π)

$$\mathcal{H} = (p, \pi) \mathbf{A}^{-1}(p, \pi)^t$$
 $\mathbf{A}^{-1} = (a^{r,s})_{r,s=1,...,N+M}$

<u>Remember</u>: The Hamiltonian equations of motion before the imposition of the holonomic constraint (i.e., the control) u = u(t)

$$\begin{cases} \dot{q}_{i} = \frac{\partial \mathcal{H}}{\partial p_{i}} \\ \dot{u}_{\alpha} = \frac{\partial \mathcal{H}}{\partial \pi_{\alpha}} \\ \dot{p}_{i} = -\frac{\partial \mathcal{H}}{\partial q_{i}} + \mathcal{F}_{i} \\ \dot{\pi}_{\alpha} = -\frac{\partial \mathcal{H}}{\partial u_{\alpha}} + \mathcal{F}_{u_{\alpha}} \end{cases}$$

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<u>Remember</u>: the equations of motion <u>after</u> the imposition of the constraint (=control) u = u(t):

$$\begin{pmatrix} \dot{q} \\ \dot{p} \end{pmatrix} = \overbrace{\begin{pmatrix} 0 \\ F^{u(\cdot)} \end{pmatrix}}^{f} + \phi + \sum_{\alpha=1}^{M} g_{\alpha} \dot{u}_{\alpha} + \sum_{\alpha,\beta=1}^{M} h_{\alpha,\beta} \dot{u}_{\alpha} \dot{u}_{\beta}$$

with suitable vector fields $f(q, p, u), g_{\alpha}(q, p, u), h_{\alpha,\beta}(q, p, u)$ determined by the Kinetic Energy and the applied forces.

$$\begin{pmatrix} \dot{q} \\ \dot{p} \end{pmatrix} = \begin{pmatrix} 0 \\ F^{u(\cdot)} \end{pmatrix} + \phi + \sum_{\alpha=1}^{M} g_{\alpha} \dot{u}_{\alpha} + \sum_{\alpha,\beta=1}^{M} h_{\alpha,\beta} \dot{u}_{\alpha} \dot{u}_{\beta}$$

$$h_{\alpha,\beta} = \begin{pmatrix} 0 \\ \cdot \\ \cdot \\ 0 \\ \frac{\partial e_{\alpha,\beta}}{\partial q^{1}} \\ \cdot \\ \frac{\partial e_{\alpha,\beta}}{\partial q^{N}} \end{pmatrix}$$
where
$$\begin{pmatrix} e_{11}, \dots, e_{1M} \\ \dots \\ e_{M1}, \dots, e_{MM} \end{pmatrix} = \begin{pmatrix} a^{N+1,N+1}, \dots, a^{N+1,N+M} \\ \dots \\ a^{N+M,1}, \dots, a^{N+M,N+M} \end{pmatrix}^{-1}$$

The quadratic part $h_{\alpha,\beta}\dot{u}_{\alpha}\dot{u}_{\beta}$ is zero

IF AND ONLY IF

THE MATRIX

$$E = \begin{pmatrix} e_{11}, \dots, e_{1M} \\ \dots \\ e_{M1}, \dots, e_{MM} \end{pmatrix} = \begin{pmatrix} a^{N+1,N+1}, \dots, a^{N+1,N+M} \\ \dots \\ a^{N+M,1}, \dots, a^{N+M,N+M} \end{pmatrix}^{-1}$$

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The question now is:

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IS INDEPENDENT OF q.

The question now is:

Is this property a "geometric property"?

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THEOREM

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THEOREM

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THEOREM

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(2) Geodesics orthogonal to <u>one</u> leaf $\{u = constant\}$ are orthogonal to <u>all</u> leaves they meet.

THEOREM

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THEOREM

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 IF AND ONLY IF
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 IF AND ONLY IF
 (3) Any pair of leaves {u = constant} have keep the same

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The "orthogonal curvature"

 $rac{\partial e_{lpha,eta}}{\partial q^i}$

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$rac{\partial e_{lpha,eta}}{\partial q^i}$

MEASURES

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 $rac{\partial e_{lpha,eta}}{\partial q^i}$

MEASURES how much geodesics which are orthogonal to $\Lambda = \{u = constant\}$ at a point

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 $rac{\partial e_{lpha,eta}}{\partial q^i}$

MEASURES how much geodesics which are orthogonal to $\Lambda = \{u = constant\}$ at a point FAIL to remain orthogonal at the other points.

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 $rac{\partial e_{lpha,eta}}{\partial q^i}$

MEASURES how much geodesics which are orthogonal to $\Lambda = \{u = constant\}$ at a point FAIL to remain orthogonal at the other points.

The orthogonal curvature $\frac{\partial e_{\alpha,\beta}}{\partial q^i}$ is a *tensor* with respect to the coordinate transformations that respect the foliation structure.

EXAMPLES

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The angle as control: quadratic part!

Non-zero

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The angle as control: quadratic part! Hence,



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No quadratic part!

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No quadratic part!

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Hence,



No quadratic part!

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Hence, no vibrational stabilization!



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Non-zero quadratic part!

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Non-zero quadratic part!

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Hence,



Non-zero quadratic part!

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Hence, chance of vibrational stabilization!

The length of the pendulum as control



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The length of the pendulum as control



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No quadratic part!

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No quadratic part!

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Hence,



No quadratic part!

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Hence, no vibrational stabilization!

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Geodesics keep orthogonality to leaves.

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Geodesics keep orthogonality to leaves. Hence no quadratic part!

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Geodesics keep orthogonality to leaves. Hence no quadratic part!Hence,

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Geodesics keep orthogonality to leaves. Hence no quadratic part!Hence, no vibrational stabilization!



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FOURTH MOVEMENT (Finale)

—Allegro molto—

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UP TO NOW EVERYTHING HAS BEEN HOLONOMIC...

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UP TO NOW EVERYTHING HAS BEEN HOLONOMIC...

So, this Final is devoted to "Non holonomic constraints"

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WHAT DOES THIS MEAN?

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WHAT DOES THIS MEAN?

A non holonomic constraint is a (linear) constraint on the velocity \dot{q}

$$\omega_1(\dot{q})=0\ldots\omega_
u(\dot{q})=0$$

which cannot be deduced by differentiation by a constraint on configuration q.

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but $\Delta(q)$ is **not integrable**

i.e., there is no *foliation* of Q whose leaves have $\Delta(q)$ as tangent space at any q.

EXAMPLE in \mathcal{R}^3



$$\Delta = \textit{span} \{ \textit{f}_1, \textit{f}_2 \}$$

where

$$f_1 = \left(\begin{array}{c} 1\\0\\y\end{array}\right), f_2 = \left(\begin{array}{c} 0\\1\\-x\end{array}\right)$$

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("Brocket example"). Δ is not integrable because it is not involutive, i.e the Lie Bracket

 $[f_1, f_2]$

does not belong to Δ .

A mechanical example: The Roller Racer

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A mechanical example: The Roller Racer



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A mechanical example: The Roller Racer



The fact that the velocity of the first body must be directed as the angle q^2 and the analog fact for the velocity of the second body,
A mechanical example: The Roller Racer



The fact that the velocity of the first body must bedirected as the angle q^2 and the analog fact for the velocity of the second body, is **non holonomic constraint** on the system.





If the Roller Racer were on a icy surface, it would be not subject to the holonomic constraints,

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i.e. the control system would be affine in u. In particular, no forces would be produced by rapid small oscillations of u.

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So one could conjecture that

"the system is affine in u if and only if this was true before the imposition of the non holonomic constraint."

IN FACT, THIS IS WRONG.

The non holonomic constraint, strikingly, **adds quadratic terms** (to \dot{u}) in the equations. Physically one can say the non holonomic constraint produces a *centrifugal effect*

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The control equations for the Roller Racer (on a surface with friction):

$$\begin{cases} \dot{q}^{1} = 2\rho \cos u \sin q^{2} \cdot \xi - \frac{J\rho \sin q^{2} \sin 2u}{2\Delta_{0}} \cdot \dot{u} \\ \dot{q}^{2} = 2 \sin u \cdot \xi - \frac{J \sin^{2} u}{\Delta_{0}} \cdot \dot{u} \\ \dot{q}^{3} = 2\rho \cos q^{2} \cos u \cdot \xi - \frac{J\rho \cos q^{2} \sin 2u}{2\Delta_{0}} \cdot \dot{u} \\ \dot{\xi} = -\sin 2u \left(\frac{(I+J-\rho^{2})}{\Delta_{1}} + \frac{1}{2(\rho^{2}/\Delta_{4}+\sin^{2} u)} \right) \cdot \xi \dot{u} - \frac{2J\rho^{2} \cos u}{\Delta_{1}^{2}} \cdot \dot{u}^{2}. \end{cases}$$

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RollerRacers' race ...





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RollerRacers' race ...

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