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Juan P Saldía, Sergio Elaskar, Luis F Gutiérrez Marcantoni, Pascal Bruel. Implementation and validation of a second-moment RANS turbulence model in OpenFOAM ®. MECOM 2021 - 37° Congreso Argentino de Mecánica Computacional, Nov 2021, Resistencia, Argentina. hal-03536915

**HAL Id: hal-03536915**

**<https://inria.hal.science/hal-03536915>**

Submitted on 20 Jan 2022

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# Implementation and validation of a second-moment RANS turbulence model in OpenFOAM®

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MECOM 2021  
XXXVII Congreso Argentino de Mecánica Computacional 1 a 5 de Noviembre de 2021  
Resistencia, Chaco, Argentina

## Background

- The no-slip condition at solid interfaces, creates a layered structure for a near-wall turbulent flow, where in the immediate vicinity called viscous sublayer, viscous effects on turbulence cannot be neglected.
- Moreover, the impermeability condition introduces a non-viscous in nature damping, that particularly dampens wall-normal velocity fluctuations.
- This *wall-blocking* effect along with the consequent reflection of pressure fluctuations in the wall, is also felt outside the viscous layer, in the fully turbulent wall region.
- The modeling of these effects it is important for overcoming deficiencies in nonequilibrium flows scenarios, primarily in strong pressure gradients, impinging flows, separation, reattachment, etc. are present.

## Background

- Simple models of near-wall turbulence usually resort to damping functions that depends on the local turbulence Reynolds number and the wall distance that are not well suited to incorporate turbulence anisotropy effects.
- In the context of an eddy viscosity RANS model Durbin (1991) noted that kinematic constraints due to blocking can be introduced through an elliptic model of pressure-velocity fluctuations.
- In this model a separate transport equation was solved for a scalar surrogate of wall-normal velocity fluctuations (designated  $v^2$ ), in conjunction with an *elliptic relaxation* parameter  $f$ , that has become known since as the  $v^2 - f$  model.
- In Durbin (1993) the elliptic relaxation concept is applied, but now in the context of a full second-moment closure. In this latter method, the elliptic relaxation has a tensorial character, which boundary conditions near a solid wall introduce a source of of numerical stiffness and instability.

## Background

- In Manceau and Hanjalić (2002), the authors proposed a model that reduces the tensorial character of Durbin's second-order model to a single scalar elliptic blending variable.
- This model maintains the solid theoretical basis and modeling properties of Durbin's model, while increasing the robustness and ease of implementation and adaptation to industrial CFD codes.

## Second-order RANS equations

- Consider the Reynolds decomposition of the velocity into a mean  $U_i$  and a fluctuating part  $u_i$ .
- The *Reynolds averaged* momentum equation for an incompressible flow of kinematic viscosity  $\nu$  in the absence of external forces reads as:

$$\frac{DU_i}{Dt} = -\frac{\partial P}{\partial x_i} + \frac{\partial}{\partial x_j} \left[ \nu \left( \frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} - \overline{u_i u_j} \right) \right]$$

- The Reynolds-stress transport equation reads as:

$$\frac{D\overline{u_i u_j}}{Dt} = P_{ij} + D_{ij}^\nu + D_{ij}^T + \phi_{ij}^* - \epsilon_{ij}$$

- ▶  $P_{ij}$ : turbulent production.
- ▶  $D_{ij}^\nu$ : viscous diffusion.
- ▶  $D_{ij}^T$ : turbulent diffusion by fluctuating velocity.
- ▶  $\phi_{ij}^*$ : fluctuating velocity-pressure gradient correlation.
- ▶  $\epsilon_{ij}$ : turbulence dissipation.

# Elliptic Blending Method (EBM)

- Velocity-pressure gradient correlation, Manceau (2015):

$$\phi_{ij}^* = (1 - \alpha^3)\phi_{ij}^w + \alpha^3\phi_{ij}^h$$

- ▶ Homogeneous term:

$$\begin{aligned}\phi_{ij}^h = & - \left( g_1 + g_1^* \frac{P}{\epsilon} \right) \epsilon b_{ij} + \left( g_3 - g_3^* \sqrt{b_{kl}b_{kl}} \right) k S_{ij} \\ & + g_4 k \left( b_{ik} S_{jk} + b_{jk} S_{ik} - \frac{2}{3} b_{lm} S_{lm} \delta_{ij} \right) \\ & + g_5 k \left( b_{ij} W_{jk} + b_{jk} W_{ik} \right)\end{aligned}$$

where

$$\begin{aligned}b_{ij} &= \frac{\overline{u_i u_j}}{2k} - \frac{1}{3} \delta_{ij} \\ S_{ij} &= \frac{1}{2} \left( \frac{\partial U}{\partial x_j} + \frac{\partial U_j}{\partial x_i} \right), \quad W_{ij} = \frac{1}{2} \left( \frac{\partial U_i}{\partial x_j} - \frac{\partial U_j}{\partial x_i} \right)\end{aligned}$$

- ▶ Wall reflection term:

$$\phi_{ij}^w = -5 \frac{\epsilon}{k} \left[ \overline{u_i u_k} n_j n_k + \overline{u_j u_k} n_i n_k - \frac{1}{2} \overline{u_k u_l} n_k n_l (n_i n_j + \delta_{ij}) \right]$$

- Elliptic relaxation equation:

$$\alpha - L^2 \nabla^2 \alpha = 1$$

- ▶ Approximate wall normal vector:

$$\mathbf{n} = \frac{\nabla \alpha}{\|\nabla \alpha\|}$$

- Turbulent diffusion:

$$D_{ij}^T = \frac{\partial}{\partial x_l} \left( \frac{C_\mu}{\sigma_k} \overline{u_l u_m} T \frac{\partial \overline{u_i u_j}}{\partial x_m} \right)$$

- Time and length scales:

$$T = \max \left( \frac{k}{\epsilon}, C_T \left( \frac{\nu}{\epsilon} \right)^{1/2} \right), \quad L = C_L \max \left( \frac{k^{3/2}}{\epsilon}, C_\eta \frac{\nu^{3/4}}{\epsilon^{1/4}} \right)$$



- Dissipation tensor:

$$\epsilon_{ij} = (1 - \alpha^3) \frac{\overline{u_i u_j}}{k} \epsilon + \frac{2}{3} \alpha^3 \epsilon \delta_{ij}$$

- Turbulence dissipation rate equation:

$$\frac{D\epsilon}{Dt} = \frac{C'_{\epsilon_1} P - C_{\epsilon_2} \epsilon}{T} + \frac{\partial}{\partial x_l} \left( \frac{C_\mu}{\sigma_\epsilon} \overline{u_l u_m T} \frac{\partial \epsilon}{\partial x_m} \right) + \nu \frac{\partial^2 \epsilon}{\partial x_k \partial x_k}$$

where

$$C'_{\epsilon_1} = C_{\epsilon_1} \left[ 1 + A_1 (1 - \alpha^3) \frac{P}{\epsilon} \right]$$

- Model constants:

$$g_1 = 3.4; \quad g_1^* = 1.8; \quad g_3 = 0.8; \quad g_3^* = 1.3; \quad g_4 = 1.25; \quad g_5 = 0.4;$$

$$C_\mu = 0.21; \quad \sigma_k = 1.0; \quad C_T = 6.0; \quad C_L = 0.133; \quad C_\eta = 80.0;$$

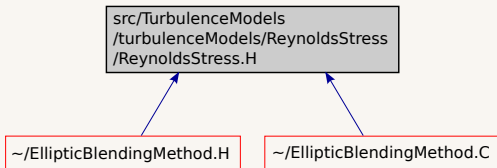
$$C_{\epsilon_1} = 1.44; \quad C_{\epsilon_2} = 1.83; \quad A_1 = 0.065; \quad \sigma_\epsilon = 1.15$$

- Wall boundary conditions:

$$U_i = 0; \quad \overline{u_i u_j} = 0; \quad \epsilon = 2\nu \lim_{y \rightarrow 0} \frac{k}{y^2}; \quad \alpha = 0$$

# Implementation

- Implemented in OpenFOAM<sup>®</sup> v2012 as a class inherited from the ReynoldsStress class:



- We followed guidelines from the implementation that is described by Javadi (2016).
- Code Saturne Archambeau et al. (2004) also provides a publicly available implementation.

## Results

- The implementation has been tested with three cases:
  - ▶ Low Reynolds flow in a plane channel.
  - ▶ Flow over periodic hills.
  - ▶ Flow in an asymmetric diffuser.
- RANS equations are discretized with a collocated second order cell-centered finite volume method and solved through the SIMPLE algorithm.
- Convergent residuals below  $10^{-6}$  were noted in all variables, with the exception of pressure that stalled at  $\approx 10^{-4}$  in most cases.
- With the aim of comparison results with a low Reynolds  $k-\epsilon$  model Launder et al. (1974) were also obtained.

## Plane channel flow

- Fully developed turbulent flow in a plane channel flow at low Reynolds number.
- Results are compared with DNS solutions presented in Kim et al. (1987).
- Reynolds number based on friction velocity corresponds to  $Re_\tau = 180$  and  $Re_\tau = 395$ .
- First wall-normal cell height  $\Delta y^+ \approx 1$ .
- 300 cells in normal direction. One cell in streamwise direction with cyclic boundary conditions.

# Plane channel flow

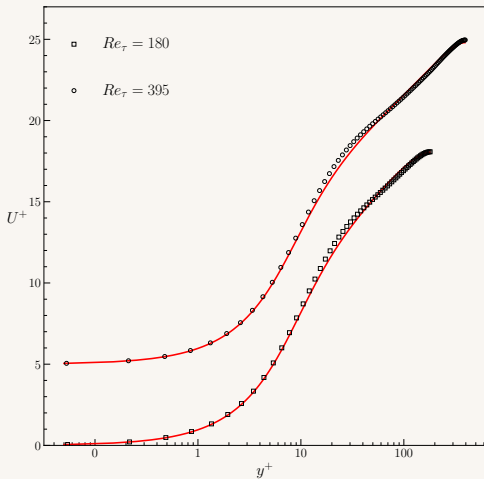
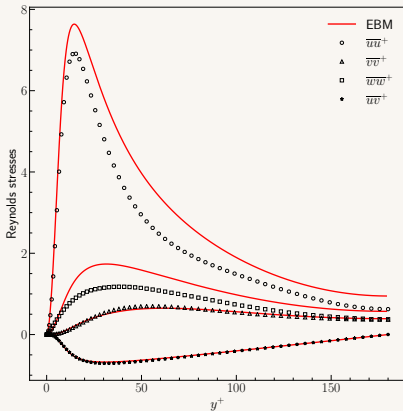
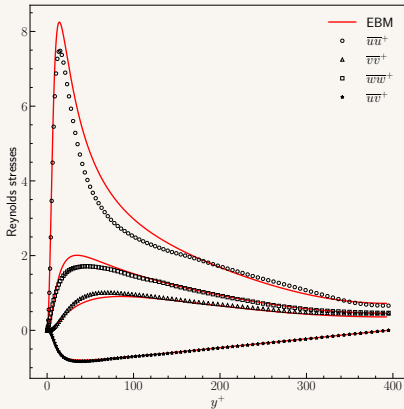


Fig. 1: Velocity profiles for a plane channel flow. Symbols: DNS, Kim et al. (1987); lines: computations with EBM model. Results shifted for clarity.

# Plane channel flow



(a)  $Re_{\tau} = 180$



(b)  $Re_{\tau} = 395$

Fig. 2: Reynolds stresses for a plane channel flow. Symbols: DNS, Kim et al. (1987); lines: computations with EBM model.

## Plane channel flow

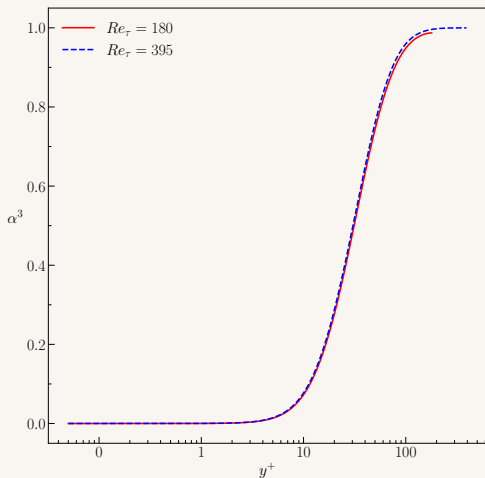


Fig. 3: Elliptic blending function profile for a plane channel flow.

## Flow over periodic hills

- This test corresponds to the 2D turbulent flow over periodic hills numerically investigated by Temmerman et al. (2003)

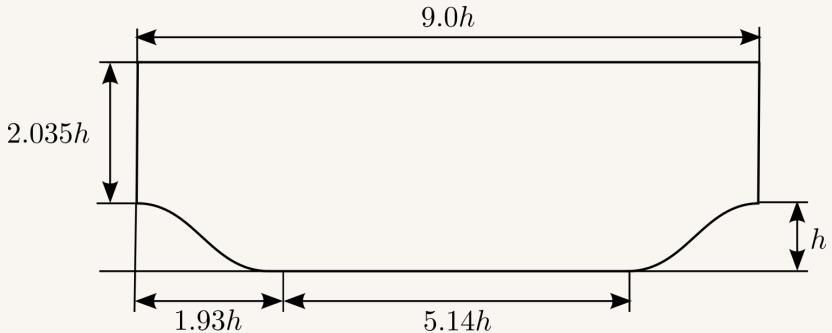


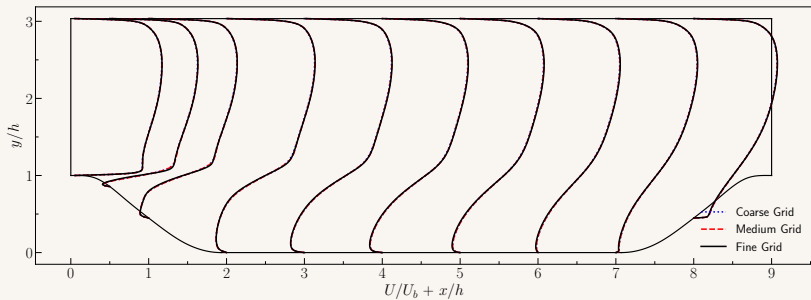
Fig. 4: Periodic hill case description.



## Flow over periodic hills

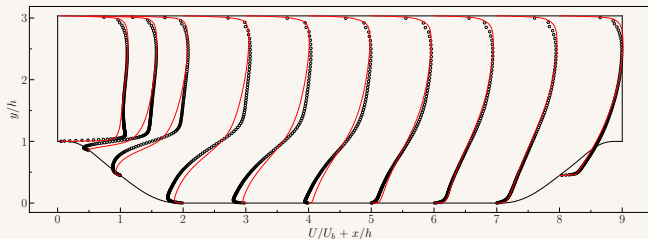
- Fully developed turbulent flow in a plane channel flow at low Reynolds number.
- Results are compared with LES solutions presented in Temmerman et al. (2003).
- Reynolds number based on bulk velocity and the hill height is  $Re_h = 10595$ .
- Mesh convergence was evaluated in three meshes with increasing level of refinement.
- Coarse:  $100$  (streamwise)  $\times$   $70$  (normal). Semi-coarse:  $100 \times 140$ . Fine:  $200 \times 140$ .
- First wall-normal cell height  $\Delta y^+ \approx 1$  was kept constant in all meshes.

## Flow over periodic hills

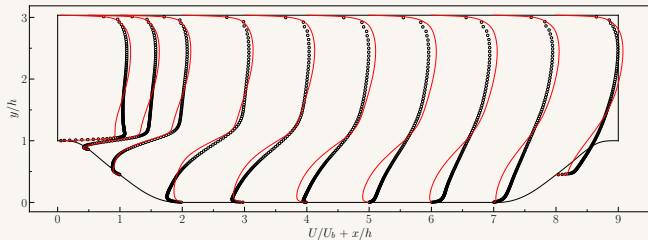


**Fig. 5:** Mesh convergence of flow over periodic hills simulations. Streamwise velocity profiles computed with EBM model.

# Flow over periodic hills



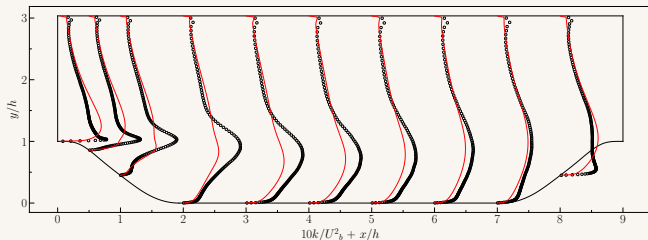
(a) Launder and Sharma low-Reynolds  $k - \epsilon$  model



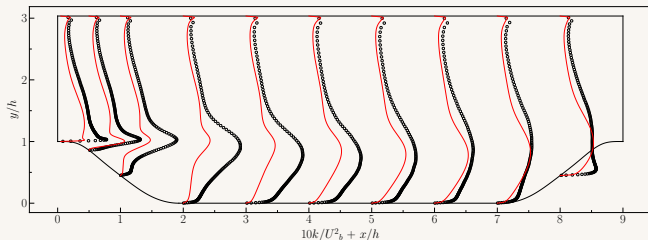
(b) Elliptic blending method

**Fig. 6:** Streamwise velocity profiles for a flow over periodic hills. Symbols: LES Temmerman et al. (2003); lines: computations.

# Flow over periodic hills

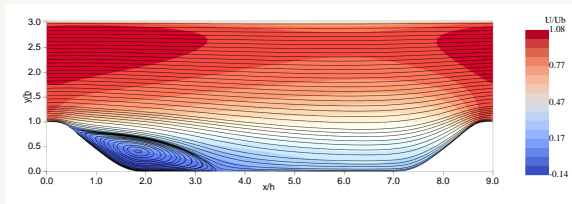


(a) Launder and Sharma low-Reynolds  $k - \epsilon$  model

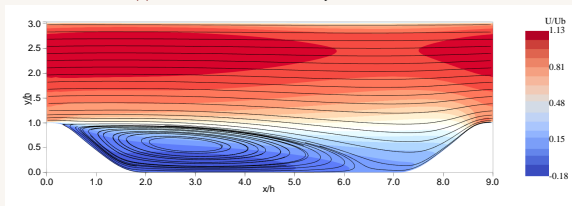


(b) Elliptic blending method

**Fig. 7:** Normal velocity profiles for a flow over periodic hills. Symbols: LES Temmerman et al. (2003); lines: computations.



(a) Launder and Sharma low-Reynolds  $k - \epsilon$  model



(b) Elliptic blending method

**Fig. 8:** Streamline profiles for a flow over periodic hills.

	Separation ( $h$ )	Reattachment ( $h$ )
LES Temmerman et al. (2003)	0.22	4.72
Low Reynolds $k - \epsilon$	0.31	3.51
Elliptic Blending Method	0.26	6.35

## Asymmetric diffuser

- This case corresponds to the separated 2D flow through the axisymmetric diffuser studied experimentally by Buice et al. (2000).

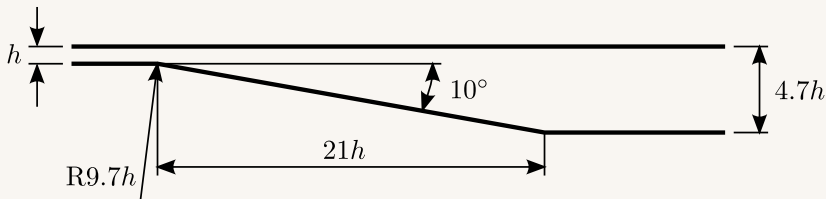
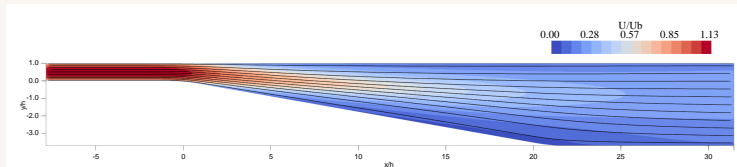


Fig. 9: Asymmetric diffuser geometry

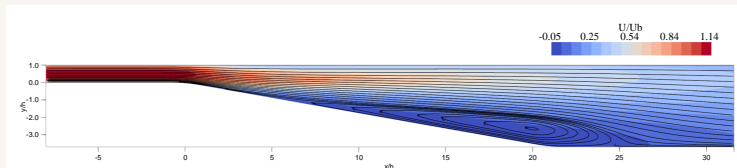
## Flow over periodic hills

- Fully developed turbulent flow in a plane channel flow at low Reynolds number.
- Results are compared with experimental results presented in Buice et al. (2000).
- Reynolds number based on center line velocity and entrance channel height  $h$  is:  $Re_h = 20000$ .
- Mesh convergence was evaluated in three meshes with increasing level of refinement.
- Coarse:  $160$  (streamwise)  $\times$   $140$  (normal). Semi-coarse:  $210 \times 190$ . Fine:  $320 \times 240$ .
- First wall-normal cell height  $\Delta y^+ \approx 0.5$  was kept constant in all meshes.

# Asymmetric diffuser



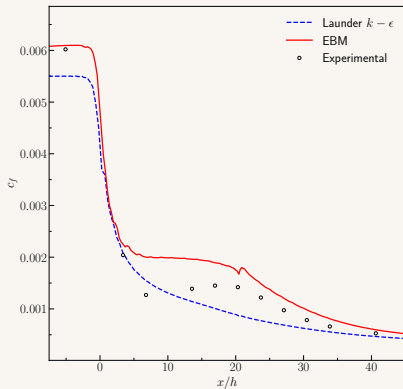
(a) Launder and Sharma low-Reynolds  $k - \epsilon$  model



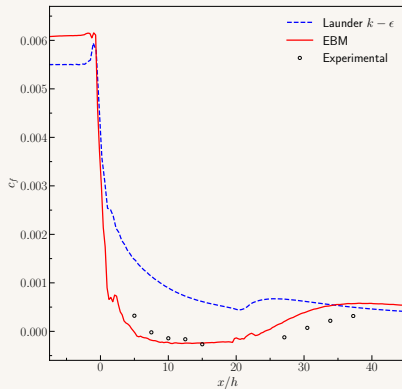
(b) Elliptic blending method

Fig. 10: Streamline profiles for the asymmetric diffuser





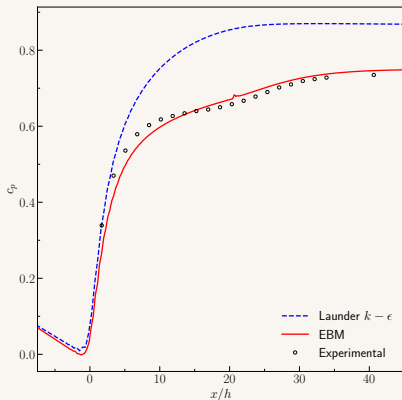
(a) Flat wall



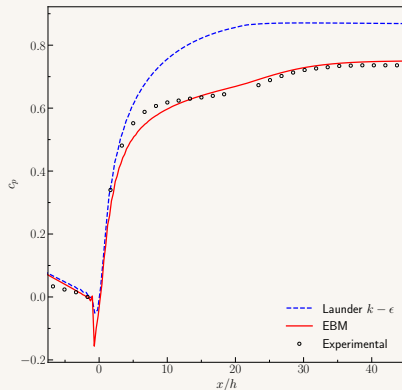
(b) Inclined wall

**Fig. 11:** Friction coefficient for the asymmetric diffuser.  
 Symbols: experiments; lines: computations.

	Separation ( $h$ )	Reattachment ( $h$ )
Experimental Buice et al. (2000)	7.3	29.2
Low reynolds $k - \epsilon$	NA	NA
Elliptic Blending Method	4.56	24.8

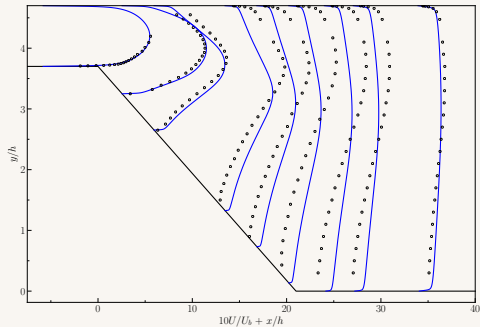


(a) Flat wall

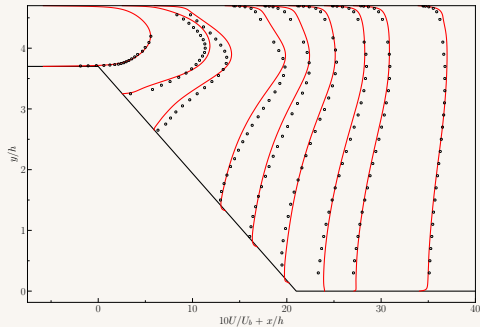


(b) Inclined wall

**Fig. 12:** Pressure coefficient for the asymmetric diffuser.  
 Symbols: experiments; lines: computations.



(a) Launder and Sharma low-Reynolds  $k - \epsilon$  model



(b) Elliptic blending method

**Fig. 13:** Streamwise velocity profiles for the asymmetric diffuser.  
 Symbols: experiments; lines: computations.

## Conclusions

- The implementation and validation of the Elliptic Blending Method in OpenFOAM<sup>®</sup> has been presented.
- Results were compared with high-fidelity numerical solutions corresponding to test cases involving flows with attached and separated boundary layers with streamline curvature effects.
- The performance of the model was also assessed in the separated flow in a 2D asymmetric diffuser where the low-Reynolds  $k - \epsilon$  turbulence model was unable to capture the turbulent separation bubble.
- The implementation is considered satisfactory and as future work further testing in 3D separated flows problems will be considered.

**¡Muchas gracias por su atención!**