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#### A SUFFICIENT CONDITION FOR THE ABSENCE OF 2D 2 INSTABILITIES OF AN ELASTIC PLATE IN A DUCT WITH **COMPRESSIBLE FLOW\*** 3

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#### J-F. MERCIER<sup>†</sup>

5 Abstract. We study the time-harmonic resonance of a finite-length elastic plate in a fluid in 6uniform flow confined in a duct. Although the resonance frequencies are usually real, the combined effects of plate elasticity and of a flow can create complex frequencies, different from the usual so-7 8 called scattering frequencies, corresponding to instabilities. We study theoretically the existence of 9 instabilities versus several problem parameters, notably the flow velocity and the ratio of densities and of sound speeds between the plate and the fluid. A 3D-volume in the parameters space is defined, 10 11 in which no instability can develop. In particular it corresponds to a low enough velocity or a light enough plate. The theoretical estimates are validated numerically. 12

**1.** Introduction. We are interested in an acoustic fluid-structure problem: an 13 elastic finite plate surrounded by a compressible uniform flow confined in a rigid 14 15waveguide. We consider the time harmonic regime: all the quantities vary like  $e^{-i\omega t}$ where  $\omega$  is the frequency. 16

In this paper we focus on the resonance frequencies: the frequencies for which 17there is non-uniqueness in the associated scattering problem, with a solution of the 18homogeneous problem in the  $L^2$ -space. It means that without any acoustic source, 1920 fluid vibrations and plate deformations, periodic in time and with finite energy, can exist. This solution is called a trapped mode since it is confined to the vicinity of 21 the obstacle and do not radiate energy. Therefore, these trapped modes are of great 22 physical importance because at the resonance frequencies, the response to a forced 23 excitation can be quite high. 24

The case of real resonance frequencies  $(\Im m(\omega) = 0)$  has already been studied 25[1, 2]. These frequencies are obtained as eigenvalues of a self-adjoint operator. Due to 26 the fluid-structure coupling, a quadratic eigenvalue problem is involved, in which the 27resonance frequencies  $\omega$  solve the equations  $\lambda(\omega) = \omega^2$  where  $\lambda$  are the eigenvalues 28 of an operator of the form  $A + \omega B$ . Then the method is in two steps. For a fixed 2930  $\omega$  value, the spectrum of the operator is made of a continuous essential spectrum and of eigenvalues below the essential spectrum ( $\omega < \omega_c$  defined in Eq. (6)). We restricted to the discrete spectrum which is studied thanks to the Min-Max principle. 32 In particular we proved the existence of eigenvalues  $\lambda_i(\omega)$ ,  $i = 1, \dots, N$  and we 33 determined theoretical lower bounds for N. Then the fixed-point equations  $\lambda_i(\omega) =$ 34  $\omega^2, i = 1, \cdots, N$  are solved. The number of resonances has been estimated versus 35 various parameters, notably the flow velocity and the rigidity of the plate. 36

37 This study of real resonances has revealed an interesting behavior: when the fluidstructure problem is decoupled (rigid plate or fluid at rest), the resonance frequencies 38 are necessarily real [1] (see also lemma 7). But when both effects are present, fluid 39 in flow and elastic plate, the resonance frequencies can be complex [2]. This is the 40 object of this paper, to study theoretically the resonance frequencies with  $\Im(\omega) > 0$ . 41 These complex frequencies are particularly interesting because they correspond to a 42 stronger transient response to a source of frequency  $\omega$  than for real frequencies: for 43 a real resonance, the response varies like  $te^{-i\omega t}$  whereas the response grows expo-44

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nentially for a complex resonance, like  $e^{\Im m(\omega)t}$ . This latter case corresponds to the 45 flutter instability encountered in aeroelasticity of airfoils [3, 4]. More generally the 46 interaction of a finite flexible panel with a fluid flow with which the plate is aligned 47 is a fundamental problem that has received a large attention because of its practical 48importance [5]. The main application lies in aerospace with the study of the airfoil 49flutter of a flexible plate immersed in an axial flow. This instability is similar to flag 50flutter and results from the competition between fluid forces and elasticity. The flutter 51instability has also applications in paper industry or snoring. Experimental [6] and numerical [7] studies have been carried out to clarify the paper flutter phenomenon, which currently limits the speed of printing machines and paper machines. Besides 54to study human snoring, the stability of a flexible cantilevered plate in channel flow has been studied as a representation of the dynamics of the human upper airway 56 [8]. The focus is on instability mechanisms of the flexible plate that cause airway 57blockage during sleep. A last class of applications concerns nuclear engineering where 58 parallel-plate assemblies are used as core elements in some nuclear research and power 59reactors. References on these subjects may be found in [9]. 60

To our knowledge, this study is new. Most of the works on trapped modes concern 61 62 a rigid body and a compressible fluid at rest [10, 11, 12], the main application being the water-waves trapped by vertical cylinders of various cross-section shapes. Unlike 63 the case of an unbounded flexible wall [13], works on a bounded elastic body and a 64 fluid in flow are rather focused on an incompressible flow [14] or a slightly compressible 65 flow [15, 16, 17, 18]. Some studies consider the case of a compressible flow, but mostly 66 67 for a membrane structure, not an elastic body, with applications to the control of noise [19, 20, 21]. The case of an elastic structure in a compressible flow has been treated but 68 in conditions different from ours [22]: although we consider a rectangular geometry in 69 the time-harmonic regime with an elastic plate clamped-free, in [22] is considered a cylindrical geometry in the time-domain with clamped-clamped boundary conditions. 71The configuration in [22] is a flow in a cylindrical tube, a portion of which is flexible: 7273 it is mostly a membrane and some results are given for a shell structure. Although these differences, a critical velocity  $U_c$  is defined in [22] to which we will compare (see 74paragraph 5.3). The critical velocity is derived using an alternative approach to ours, 75which consists in determining that a positive energy can be defined for  $U < U_c$ . 76

The outline of this paper is as follows. In the second section is presented the 77 problem and the definition of what we call an instability is given. In the third section 78 is derived a general criterion for the existence of instabilities and this criterion is 79 precised in section 4: it is written in terms of the problem parameters, the ratio of 80 densities  $\mu$ , the flow velocity U and the ratio of sound speeds c. In particular, a volume 81 in the space  $(\mu, U, c)$  is defined such that no instabilities can develop in this volume. 82 In section 5, the boundaries of this volume are further investigated. In particular, 83 thanks to an approximation, threshold values  $\mu_c$  and  $U_c$  are explicitly derived such 84 that instabilities can develop only above these thresholds:  $\mu > \mu_c$  and  $U > U_c$ . 85 Finally, numerical validations of the derived theoretical estimates are presented in 86 section 6. 87

# 88 2. The general framework.

**2.1. Geometry and dimensional equations.** The geometry is shown in Fig. 1: we consider the 2D waveguide  $\tilde{\Omega} = \{(X, Y) \in \mathbb{R} \times ] - h, h]$  filled with a compressible fluid. In the centre of the guide is placed an elastic plate  $\tilde{\Gamma} = \{(X, Y), X \in ]0, L[, Y = 0\}$ . The fluid is in flow with a uniform velocity  $u_0 e_X$ . We take  $0 < u_0 < c_0$  (subsonic flow) where  $c_0$  is the uniform fluid sound speed.



FIG. 1. Geometry of the problem

In time-harmonic regime of frequency  $\tilde{\omega}$ , if we send an incident wave  $e^{i\tilde{\omega}(X/c_0-t)}$ from  $X = -\infty$  on the plate and if we look for the scattered field, the problem is found well-posed except for a sequence of so-called resonance frequencies [1, 2]. These resonances correspond to the existence of eigenmodes (called also trapped modes): solutions in  $L^2(\tilde{\Omega})$  of the problem without source. In this paper we focus on these trapped modes and therefore we look for a solution of the following homogeneous problem [1, 2]:

$$\begin{cases}
\left(\frac{\partial^2}{\partial X^2} + \frac{\partial^2}{\partial Y^2}\right)\tilde{\varphi} = \left(\frac{\tilde{D}}{c_0}\right)^2 \tilde{\varphi} \quad \text{in} \quad \tilde{\Omega} \setminus \tilde{\Gamma}, \\
\left(B\frac{d^4}{dX^4} - \rho_p h_p \tilde{\omega}^2\right)\tilde{u} = -[p](X,0) \quad \text{on} \quad \tilde{\Gamma}, \\
p(X,Y) = -\rho_a \tilde{D}\tilde{\varphi}(X,Y) \quad \text{in} \quad \tilde{\Omega} \setminus \tilde{\Gamma}, \\
\frac{\partial\tilde{\varphi}}{\partial Y}(X,0^{\pm}) = \tilde{D}\tilde{u} \quad \text{on} \quad \tilde{\Gamma}, \\
\frac{\partial\tilde{\varphi}}{\partial Y}(X,\pm h) = 0 \quad \text{for} \quad X \in \mathbb{R}, \\
\tilde{u}(0) = 0 = \tilde{u}'(0), \quad \tilde{u}''(L) = 0 = \tilde{u}'''(L),
\end{cases}$$

where p is the fluid pressure,  $\tilde{\varphi}$  the fluid velocity potential and  $\tilde{u}$  the vertical displacement of the plate. The velocity potential satisfies the convected Helmholtz equation whereas the displacement is solution of the linear plate equation. We have introduced  $[p](X,0) = p(X,0^+) - p(X,0^-)$  the pressure jump through the plate and

106 
$$\tilde{D} = u_0 \frac{\partial}{\partial X} - i\tilde{\omega}.$$

107 the convective operator. The densities are  $\rho_a$  for the fluid (typically the air) and  $\rho_p$ 108 for the plate. The plate of width  $h_p$  (supposed very small, the plate is seen as 1D) 109 and of flexural rigidity B, is clamped at its leading edge X = 0 and free at its trailing 110 edge X = L.

**2.2.** Some symmetries considerations. In the following we will restrict to an 111 antisymmetric problem. Indeed, if the obstacle is symmetric and positioned symmetri-112 cally about the waveguide centerline, the solution can be decomposed into a symmetric 113 and an antisymmetric part about that line. Antisymmetric trapped modes are gen-114115erally looked for because the continuous spectrum of the antisymmetric solution has a nonzero lower limit given by the first duct cut-off frequency (see paragraph 2.4.4). 116 117 Then the corresponding antisymmetric trapped modes can be found as discrete eigenvalues below the first cut-off frequency. On the contrary, for symmetric modes, all 118 the spectrum is continuous. 119

120 Trapped modes can be determined if the obstacle is no longer symmetric about 121 the waveguide centerline but this situation is mathematically more complicated. In

this case, possible discrete trapped mode frequencies are embedded in the continuous
spectrum of the relevant operator, and one speaks of embedded trapped modes [23,

124 24, 25].

Since we only study antisymmetric modes, we can restrict to a half guide:  $Y \in [0, h]$ , the boundary condition [p](X, 0) becomes 2p(X, 0) and the boundary condition  $\tilde{\varphi}(X, 0) = 0$  for  $X \notin [0, L]$  has to be added.

128 **2.3.** Dimensionless equations and reduction to a half guide. Following 129 [9], we introduce the parameter  $\gamma$  such that

130 
$$\gamma^2 = \frac{B}{\rho_p h_p}$$

131 and the following dimensionless parameters:

132 
$$\alpha = \frac{h}{L}, \quad \mu = \frac{\rho_a L}{\rho_p h_p}, \quad c = c_0 \frac{L}{\gamma}.$$

133  $\alpha$  is an aspect ratio,  $\mu$  compares the fluid and plate densities and c can be seen 134 as the ratio of the fluid and plate sound speed. Following [9] we also introduce the 135 dimensionless variables x = X/L, y = Y/h and the following dimensionless quantities: 136 velocity U, velocity potential  $\varphi$ , vertical plate displacement u and frequency  $\omega$  defined 137 by

138 
$$u_0 = \frac{\gamma}{L}U, \quad \tilde{\varphi} = \gamma\varphi, \quad \tilde{u} = Lu, \quad \tilde{\omega} = \frac{\gamma}{L^2}\omega.$$

139 We note  $\Omega = \{(x, y) \in \mathbb{R} \times ]0, 1[\}$  the half wave guide,  $\Gamma = \{(x, y), x \in ]0, 1[, y = 0\}$ 140 the plate and  $\Sigma = \{(x, y), x \in \mathbb{R}, y = 1\}$  the upper wall. Also we introduce  $\Gamma^- = \{(x, y), x < 0, y = 0\}$  and  $\Gamma^+ = \{(x, y), x > 1, y = 0\}$  (see Fig. 2) such that  $\Gamma^- \cup \Gamma \cup \Gamma^+$ 142 is the lower wall of the half-guide.

$$\begin{array}{cccc} y=1 & \underline{\Sigma} & & \frac{\partial \varphi}{\partial y}=0 \\ & \Omega & & \\ y=0 & \underline{\Gamma^{-}} & \underline{\Gamma} & \underline{\Gamma^{+}} & & \varphi=0 \\ & \varphi=0 & & \frac{1}{\alpha} \frac{\partial \varphi}{\partial y} = \left(U \frac{\partial}{\partial x} - i\omega\right) u \end{array}$$



Finally, the problem (1) written in a half waveguide in dimensionless form is

143

144 (2) 
$$\begin{cases} \left(\frac{\partial^2}{\partial x^2} + \frac{1}{\alpha^2}\frac{\partial^2}{\partial y^2}\right)\varphi &= \frac{1}{c^2}\left(U\frac{\partial}{\partial x} - i\omega\right)^2\varphi \quad \text{in} \quad \Omega,\\ \left(\frac{d^4}{dx^4} - \omega^2\right)u &= 2\mu\left(U\frac{\partial}{\partial x} - i\omega\right)\varphi(x,0) \quad \text{on} \quad \Gamma,\\ \frac{1}{\alpha}\frac{\partial\varphi}{\partial y}(x,0) &= \left(U\frac{\partial}{\partial x} - i\omega\right)u \quad \text{on} \quad \Gamma,\\ \frac{\partial\varphi}{\partial y}(x,1) &= 0 \quad \text{on} \quad \Sigma,\\ \psi(x,0) &= 0 \quad \text{on} \quad \Sigma,\\ u(0) = 0 = u'(0) \quad \text{and} \quad u''(1) = 0 = u'''(1). \end{cases}$$

145 This is a non-linear eigenvalue problem in which  $\omega$  is the eigenvalue. As we will see,

146 these eigenvalues can be divided in real and complex eigenvalues.

147 **2.4.** Some generalities on the trapped modes.

148 **2.4.1. Definition of a resonance frequency.** We are interested in the exis-149 tence, for a complex frequency, of an acoustic field trapped around the plate, that we 150 call a resonance:

151 DEFINITION 1. We say that

152  $\omega \in \mathbb{C}$  is a resonance frequency  $\Leftrightarrow \exists (\varphi, u) \in L^2(\Omega) \times L^2(\Gamma)$  solution of (2).

153  $(\varphi, u)$  is called the associated trapped mode.

154 In practice it is more convenient to express the eigenproblem in a variational form.

155 Thanks to the introduction of the functional spaces

156 (3) 
$$V = \{ \varphi \in H^1(\Omega), \varphi = 0 \text{ on } \Gamma^{\pm} \},$$

157 and

16

158 (4) 
$$W = \{ u \in H^2(\Gamma), u(0) = 0 = u'(0) \},\$$

159 and of the sequilinear form on  $V \times W$ :

160 
$$b[(\varphi, u), (\psi, v)] = \int_{\Omega} (1 - M^2) \frac{\partial \varphi}{\partial x} \frac{\overline{\partial \psi}}{\partial x} + \frac{1}{\alpha^2} \frac{\partial \varphi}{\partial y} \frac{\overline{\partial \psi}}{\partial y} - i \frac{\omega}{c} M\left(\frac{\partial \varphi}{\partial x} \overline{\psi} - \varphi \frac{\overline{\partial \psi}}{\partial x}\right) + \frac{1}{2\mu\alpha} \int_{\Gamma} \frac{d^2 u}{dx^2} \frac{d^2 \overline{v}}{dx^2} dx^2 \overline{v}$$

$$+ \frac{1}{\alpha} \int_{\Gamma} \overline{\psi} \left( U \frac{\partial u}{\partial x} - i \omega u \right) + \varphi \left( U \frac{\partial \overline{v}}{\partial x} + i \omega \overline{v} \right) - \omega^2 \left( \frac{1}{c^2} \int_{\Omega} \varphi \overline{\psi} + \frac{1}{2\mu\alpha} \int_{\Gamma} u \overline{v} \right),$$

where we have introduced the Mach number  $M \equiv U/c$ , we can write the following equivalent variational formulation of the eigenvalue problem (2):

164 DEFINITION 2.  $\omega \in \mathbb{C}$  is a resonance frequency  $\Leftrightarrow \exists (\varphi, u) \in V \times W$  such that 165  $\forall (\psi, v) \in V \times W, b[(\varphi, u), (\psi, v)] = 0.$ 

166 We have three remarks:

• The space V plays the role of radiation conditions. Indeed away from the plate 167 the velocity potential is decomposed on the duct modes, see Eq. (27) and 168 more generally see Appendix A for the definition of the duct modes. Since the 169propagating modes are not in  $H^1(\Omega)$ , only the decreasing evanescent modes 170are selected when restricting to solutions in V, which leads to a solution 171trapped around the plate. In complement to the definition 2 corresponding 172to an unbounded domain, in Appendix A are defined the radiation conditions 173at finite distance which enables us to define in Appendix B the resonance 174problem set in a bounded domain. The equivalence between the two problems 175is also proved. 176

• The integration by parts leading to  $b[(\varphi, u), (\psi, v)]$  involves the boundary term

178 
$$-\left[\frac{U}{\alpha}\varphi\bar{v}\right](1,0)$$

179 We have imposed a Kutta condition at the trailing edge of the plate: we have 180 set this term to zero by assuming that the condition  $\varphi(x > 1, 0) = 0$  can 181 be extended by continuity to x = 1. Another classical Kutta condition is to 182 impose the fluid velocity to be finite at the plate trailing edge. This requires 183 to use sophisticated tools like a Wiener-Hopf approach [26, 27] and makes 184 our analysis method inapplicable. The choice of the right Kutta condition is

important since such choice has been proved to have strong influence on the 185 186far-field, in particular on the upstream radiation pattern [26]. But it seems to have a reduced influence on the unstable frequencies we are studying. Indeed 187 we have compared our results to the results in [27], in which only a rigid 188 plate in a flow is considered. We have found for different flow velocities that 189 when changing the Kutta condition from ours to the finite velocity one, the 190 resonance frequencies are just slightly moved in the complex plane. Even 191 though only the case of a rigid plate and a particular  $\alpha$  value have been 192tested, we assume that it is a general behavior: when changing of Kutta 193condition, the resonance frequencies only slightly move in the complex plane. 194written back with dimensional quantities,  $M = u_0/c_0$  and thus M < 1 since 195 • the flow is supposed subsonic. 196

197 **2.4.2.** Properties of the resonance frequencies. The search of  $\omega$  in the 198 complex plane can be restricted to a fourth of the complex plane since the spectrum 199 of resonance frequencies has some nice properties:

LEMMA 3. If  $\omega$  is an eigenvalue of (2), then  $-\omega$  and  $\pm \overline{\omega}$  are also eigenvalues.

201 Proof. By taking the complex conjugated of (2), we deduce immediately that if 202  $\omega$  is an eigenvalue associated to the eigenmode  $(\varphi, u)$  then so is  $-\overline{\omega}$  simply associated 203 to  $(\overline{\varphi}, \overline{u})$ .

The other property that if  $\omega$  is an eigenvalue then so is  $\overline{\omega}$  is mode delicate to obtain. In Appendix B we give the proof for the continuous mathematical problem. The proof is rather technical and requires to restrict the resonance problem to a bounded domain to get compacity properties in order to use the Fredholm alternative.

An alternative proof is given here, easier but restricted to the discretized problem. We have determined numerically the resonance frequencies thanks to a method coupling Finite Elements and a modal expansion. The velocity potential is discretized on a Finite Element basis while the plate displacement is expanded on the modes of the plate (details in Appendix C). This leads to a matricial problem to solve of the form

214 (5) 
$$(A+i\omega B+\omega^2 C)X=0.$$

where A, B and C are Finite Element matrices and X is a vector concatenating 215 216 the nodal values of  $\varphi$  and the modal components of u. A, B and C are real valued matrices with A and C symmetric and B antisymmetric. Then we can prove that if 217 $\omega$  is an eigenvalue,  $-\omega$  is also an eigenvalue. Indeed  $\omega$  is an eigenvalue is equivalent 218to  $\det(A + i\omega B + \omega^2 C) = 0$ . Then using that for any matrix M we have  $\det(M) =$ 219  $\det(M^T)$ , we get that  $\det(A - i\omega B + \omega^2 C) = 0$ . Finally, we have proved that if  $\omega$  is an 220 eigenvalue then  $-\overline{\omega}$  and  $-\omega$  are also eigenvalues. But then, since  $-\omega$  is an eigenvalue, 221 $\overline{\omega}$  and  $\omega$  and are also eigenvalues which ends the proof. 222

223 **2.4.3.** An example of resonances. To show the complexity of the behaviors 224 of the resonance frequencies versus the various parameters of the problem, we present 225 here a typical example of such behavior. The non-linear eigenvalue problem (5) can 226 be written as a classical (but generalized) eigenvalue problem: noting  $Y = \omega X$ , we 227 get:

228  $\begin{pmatrix} 0 & 1 \\ A & iB \end{pmatrix} \begin{pmatrix} X \\ Y \end{pmatrix} = \omega \begin{pmatrix} 1 & 0 \\ 0 & -C \end{pmatrix} \begin{pmatrix} X \\ Y \end{pmatrix}.$ 

As proved in the previous paragraph, the search of  $\omega$  in the complex plane can be restricted to a fourth of this plane: we just represent the eigenvalues in the quarter



FIG. 3. Three lowest resonance frequencies  $\omega$  when  $\mu$  varies for  $\alpha = 0.15$ , c = 1 and U = 0.9.  $\Re e(\omega)$  with dots and  $\Im m(\omega)$  with circles. (a): zoom on the small  $\mu$  values, up to 2.5, (b):  $0 < \mu < 25$ 

plane  $\Re e(\omega) \ge 0$ ,  $\Im m(\omega) \ge 0$ . In Fig. 3 are plotted the evolution of the three lowest 231(in the sense  $|\omega|$  small) resonance frequencies when  $\mu$  varies, the other parameters 232 being fixed: c = 1 and U = 0.9. In all the figures of this paper, we fix the geometry 233234 and we take  $\alpha = 0.15$  which corresponds to a plate of length  $L \approx 7h$ . The real parts of  $\omega$  are with dots and the imaginary parts are with circles. Fig. 3(b) is for  $0 < \mu < 25$ 235whereas Fig. 3(a) corresponds to a zoom on the small  $\mu$  values, up to 2.5. We see 236 in Fig. 3(a) that for small values of  $\mu$  (here  $\mu < \mu_c \equiv 0.55$ ) all the frequencies are 237real. At the threshold  $\mu_c$ , the first frequency  $\omega_1$  vanishes and is found to turn from 238 purely real to purely imaginary. As we will detail latter, it becomes an instability. 239240 In Fig. 3(b), we see that at  $\mu = 16.4$ , the second real frequency  $\omega_2$  becomes purely imaginary, leading to a second instability for  $\mu > 16.4$ . Moreover, an interesting 241 event happens at  $\mu = 19.9$ , illustrating the complexity of the resonances behavior and 242 showing also that the complex frequencies are not necessarily associated to a zero real 243 part: the two instabilities, purely imaginary, merge to create for  $\mu > 19.9$  a complex 244 245frequency  $\omega_{1+2}$  with non-zero real and imaginary parts. In this paper, we will focus on the first threshold  $\mu_c$  and we will characterize it: we will derive a lower bound of 246this threshold value and also we will find an upper bound of the modulus of the first 247 unstable frequency (an upper bound of the red curves in Fig. 3). 248

From Fig. 3, it appears that the resonances can be divided in two classes:

- 250 DEFINITION 4. Let  $\omega \in \mathbb{C}$  be a resonance frequency (definition (2)).
- $\Im m(\omega) = 0 \Rightarrow \omega$  is called a real resonance,
- $\Im m(\omega) > 0 \Rightarrow \omega$  is called an instability.

253 We will now give some properties of these two kind of resonances.

254 **2.4.4. The real resonance frequencies.** The study of real frequencies has 255 already been done [1, 2]. They have been looked for below the cut-off frequency  $\omega_c$ 256 of the half wave-guide (with a Dirichlet boundary condition at y = 0, see Appendix 257 A)

258 (6) 
$$\omega < \omega_c \equiv \frac{\pi}{2\alpha} \sqrt{c^2 - U^2}.$$

Indeed, only a finite number of eigenfunctions, so-called propagating modes, propagate undamped along the waveguide while all other so-called evanescent modes decay [28]. Only the propagating modes can radiate energy to infinity and below the cut-off frequency  $\omega_c$ , all the guided modes are evanescent (see Appendix A). Then the velocity

potential  $\varphi$ , which away from the plate can be decomposed on the guided modes, naturally decreases exponentially when  $|x| \to \infty$ : it is localized around the plate and thus called a trapped mode.

Above the cut-off frequency, some resonance frequencies may exist, but they are embedded in the continuous spectrum and their mathematical study is much more involved [23, 24, 25].

269 **2.4.5. The instabilities.** We consider causal solutions and thus we focus on 270 the asymptotic solutions behavior only when  $t \to \infty$ . The second case  $\Im m(\omega) > 0$ 271 corresponds to an unstable behavior since  $e^{-i\omega t} = e^{-i\Re e(\omega)t}e^{\Im m(\omega)t}$  and  $e^{\Im m(\omega)t} \to \infty$ 272 when  $t \to \infty$  (we do not restrict when considering that  $\Im m(\omega) > 0$  since if  $\omega$  is a 273 resonance frequency,  $\overline{\omega}$  is also a resonance frequency).

274Note that these unstable frequencies are not the usual "complex resonances" that have been studied for rigid bodies [29, 30] or elastic bodies [31, 32]. The complex 275resonances are defined as the poles of the analytical continuation of the resolvent 276scattering operator and are also called scattering frequencies. These poles are very 277278 different from the unstable frequencies: the associated modes are not trapped but on the contrary they grow far from the scattering body. These modes are called leaky 279modes since they have radiation losses. Moreover the analytical continuation is made 280 in the half plane  $\Im m(\omega) < 0$  which corresponds to a decreasing behavior in time. 281Therefore the leaky modes cannot correspond to instabilities. 282

### 283 To sum up:

• a complex resonance is bounded in time but not bounded in space,

• an instability is bounded in space but not bounded in time.

In the rest of the paper, to avoid confusion, we will talk about instabilities and not complex resonances and we will focus on the existence conditions of these instabilities.

**3. General results on the instability frequencies.** First we give general properties on the imaginary part of a resonance frequency.

290 LEMMA 5. If  $\omega$  is a resonance frequency and if  $(\varphi, u)$  is the associated trapped 291 mode, then  $\Im(\omega)a(\omega; \varphi, u) = 0$  where

292 (7) 
$$a(\omega;\varphi,u) = \int_{\Omega} (1-M^2) \left| \frac{\partial \varphi}{\partial x} \right|^2 + \frac{1}{\alpha^2} \left| \frac{\partial \varphi}{\partial y} \right|^2 + \frac{1}{2\mu\alpha} \int_{\Gamma} \left| \frac{d^2u}{dx^2} \right|^2$$

293

294 
$$+\frac{2U}{\alpha}\Re e\int_{\Gamma}\overline{\varphi}\frac{\partial u}{\partial x}+|\omega|^{2}\left(\frac{1}{c^{2}}\int_{\Omega}|\varphi|^{2}+\frac{1}{2\mu\alpha}\int_{\Gamma}|u|^{2}\right)$$

295 *Proof.* From definition 2, taking  $(\psi, v) = (\varphi, u)$ , the result is deduced using

$$\Im m \left\{ -\frac{1}{\omega} b[(\varphi, u), (\varphi, u)] \right\} = 0.$$

296

This lemma has a simple consequence: a general non-existence result for instabilities, that will be used in all the following of the paper.

299 LEMMA 6. If  $\forall (\varphi, u) \in V \times W$ ,  $a(\omega; \varphi, u) > 0$ , then  $\omega$  cannot be an instability.

300 Proof. This is a consequence of lemma 5 and definition 4:  $\Im m(\omega)a(\omega;\varphi,u) = 0$ 301 and  $a(\omega;\varphi,u) > 0$  implies  $\Im m(\omega) = 0$ .

REMARK 1. In this paper we derive non-existence results for instabilities and we do not know how to derive an existence result for instabilities. But since in practice

we wish to avoid instabilities, it is useful to determine in which conditions we are sure they do not develop.

REMARK 2. We do not discuss if the instabilities we look for are physical (in the 306 sense causal) solutions. The boundary conditions are assumed to be decay at infinity 307 for any  $\Im m(\omega)$ , and this is definitely correct for the case we are interested in of 308  $\Im m(\omega) > 0$ , but may not be physically correct or causal for the other solutions with 309 310  $\Im m(\omega) < 0$ , in particular if  $\Im m(\omega)$  is sufficiently negative. This causality question is complicated when considering complex frequencies but hopefully not essential for us 311 since we focus on cases for which no instabilities can exist, whether they are causal or 312 not. If we wanted to describe how to excite an instability, then the causality question 313 would become essential. 314

In the rest of the paper, we will derive simple conditions on the parameters of the problem under which lemma 6 is fulfilled.

**4.** Criteria of non-existence of instabilities. Following lemma 6, for a fixed frequency  $\omega$  we look now for conditions, as simple as possible, for which  $a(\omega; \varphi, u) > 0$ ,  $\forall (\varphi, u) \in V \times W$ .

4.1. A simple criterion. A first result is, as announced in the introduction, that only the joint effects of the elasticity of the plate and of the flow can create instabilities:

LEMMA 7. If the fluid is at rest (U = 0) or the plate rigid (u = 0), there is no instability.

Proof. In the expression of  $a(\omega; \varphi, u)$  in (7), all the terms are strictly positive for  $(\varphi, u) \neq (0, 0)$  excepting the term

327

$$\frac{2U}{\alpha} \Re e \int_{\Gamma} \overline{\varphi} \frac{\partial u}{\partial x}.$$

Therefore, if this term is equal to zero, there is no instability using lemma (6). This is the case if U = 0 or u = 0.

As already said, the trapped modes for these uncoupled problems (fluid at rest or rigid plate) have already been studied [1]. In this paper we focus on instabilities.

4.2. A general criterion. We will now state the main theorem of this paper,
 giving an explicit sufficient condition to have no instability. For this theorem, we need
 to introduce some notations. We introduce

335 (8) 
$$\lambda_1 = \inf_{\varphi \in V \setminus \{0\}} R(\varphi),$$

336 where we have introduced the Rayleigh quotient

337 (9) 
$$R(\varphi) = \frac{\int_{\Omega} (c^2 - U^2) \left| \frac{\partial \varphi}{\partial x} \right|^2 + \frac{c^2}{\alpha^2} \left| \frac{\partial \varphi}{\partial y} \right|^2}{\int_{\Omega} \left| \varphi \right|^2}.$$

338 Also we introduce

339 (10) 
$$\alpha_1^4 = \inf_{u \in W \setminus \{0\}} \frac{\int_{\Gamma} \left| \frac{d^2 u}{dx^2} \right|^2}{\int_{\Gamma} \left| u \right|^2}.$$

The values  $\lambda_1$  and  $\alpha_1^4$  are the first eigenvalues of problems that are specified in Appendix D. The value  $\lambda_1$  is sometimes called the first eigenvalue of vibration for the fluid [22]. Its value is determined numerically and a useful approximation of  $\lambda_1$  is derived in Appendix D. In Eq. (10),  $\alpha_1^2 = \omega_1^{\text{plate}}$  where  $\omega_1^{\text{plate}}$  is the first eigenfrequency of the plate alone in vacuum. Numerically it is found that  $\alpha_1 = 1.88$ .

345 Finally we introduce

346 (11) 
$$\hat{\sigma} \equiv \min\left(\lambda_1, \alpha_1^4\right),$$

347 and

348 (12) 
$$\beta = \frac{\alpha}{c} \sqrt{\hat{\sigma}}.$$

349 With these notations, we have the

350 THEOREM 8. Let

351 (13) 
$$T(\omega) \equiv \hat{\sigma} \left[ 1 - U \left( \frac{2}{\beta} + \beta \right) \sqrt{\alpha \mu} \right] + |\omega|^2,$$

where  $\hat{\sigma}$  and  $\beta$  are defined in (11) and (12). If  $T(\omega) > 0$ , then  $\omega$  cannot be an instability.

Proof. The idea is to show that  $T(\omega)$  is a lower bound of  $a(\omega; \varphi, u)$  and then to use lemma 6. In this aim, first we derive a general lower bound for a, depending on some parameters obtained only numerically. Second, we derive another lower bound, a little less accurate, but given in a closed-form and thus easier to utilise.

• a general lower bound for  $a(\omega; \varphi, u)$ 

We decompose a defined in (7) in three parts:  $a = a_1 + a_2 + a_3$ , with

$$\begin{cases} a_1(\varphi, u) = \int_{\Omega} (1 - M^2) \left| \frac{\partial \varphi}{\partial x} \right|^2 + \frac{1}{\alpha^2} \left| \frac{\partial \varphi}{\partial y} \right|^2 + \frac{1}{2\mu\alpha} \int_{\Gamma} \left| \frac{d^2 u}{dx^2} \right|^2, \\ a_2(\varphi, u) = \frac{2U}{\alpha} \Re e \int_{\Gamma} \overline{\varphi} \frac{\partial u}{\partial x}, \\ a_3(\varphi, u) = |\omega|^2 ||(\varphi, u)||^2, \end{cases}$$

361 with

362

$$||(\varphi, u)||^{2} = \frac{1}{c^{2}} \int_{\Omega} |\varphi|^{2} + \frac{1}{2\mu\alpha} \int_{\Gamma} |u|^{2}.$$

We look for a positive lower bound for a and in this aim, we determine a small upper bound for  $|a_2|$ . Using the Poincaré's inequality it is found that  $\forall u \in W$  defined in (4)

366 (15) 
$$\int_{\Gamma} \left| \frac{du}{dx} \right|^2 \le \frac{1}{2} \int_{\Gamma} \left| \frac{d^2u}{dx^2} \right|^2.$$

367 We also use the trace inequality:  $\forall \tau > 0$  and  $\forall \varphi \in V$ ,

368 (16) 
$$\int_{\Gamma} |\varphi|^2 \le (1+\tau^2) \int_{\Omega} |\varphi|^2 + \frac{1}{\tau^2} \int_{\Omega} \left| \frac{\partial \varphi}{\partial y} \right|^2$$

369 This trace inequality is derived by using an alternative of the Poincaré's inequality

<sup>370</sup> for a fixed value of x, and then by integrating along the x axis on the plate  $\Gamma$ .

### A SUFFICIENT CONDITION FOR THE ABSENCE OF 2D INSTABILITIES OF AN ELASTIC PLATE IN A DUCT WITH COMPRESSIBLE FL

371 Then, writing  $a_2$  under the form

372 
$$a_2 = 2\eta \int_{\Gamma} \frac{\tau}{\alpha} |\varphi| \frac{\left|\frac{du}{dx}\right|}{\sqrt{\mu\alpha}},$$

with  $\eta = U\sqrt{\mu\alpha}/\tau$ , we deduce with Young's inequality and then with (15) and (16) that:

375 
$$\frac{|a_2|}{\eta} \leq \int_{\Gamma} \frac{\tau^2}{\alpha^2} \left|\varphi\right|^2 + \frac{1}{\mu\alpha} \left|\frac{du}{dx}\right|^2,$$

$$\eta = \int_{\Gamma} \alpha^{2} dx + \mu \alpha |dx|^{2}$$

$$\leq \frac{\tau^{2}(1+\tau^{2})}{\alpha^{2}} \int_{\Omega} |\varphi|^{2} + \frac{1}{\alpha^{2}} \int_{\Omega} \left|\frac{\partial\varphi}{\partial y}\right|^{2} + \int_{\Gamma} \frac{1}{2\mu\alpha} \left|\frac{d^{2}u}{dx^{2}}\right|^{2}$$

377 Then using (14) leads to

378 
$$\frac{|a_2|}{\eta} \le \tau^2 (1+\tau^2) \frac{c^2}{\alpha^2} ||(\varphi, u)||^2 + a_1$$

Finally, using  $a \ge a_1 - |a_2| + a_3$ , we deduce  $\forall (\varphi, u) \in V \times W$ ,  $(\varphi, u) \neq (0, 0)$  and  $\forall \omega \in \mathbb{C}$ :

381 
$$\frac{a(\omega;\varphi,u)}{||(\varphi,u)||^2} \ge (1-\eta)\sigma^* - \eta\tau^2(1+\tau^2)\frac{c^2}{\alpha^2} + |\omega|^2,$$

382 where

383

$$\sigma^{\star} = \inf_{(\varphi, u) \in V \times W} \frac{a_1(\varphi, u)}{||(\varphi, u)||^2}.$$

This lower bound is more accurate than T defined in (13) but requires the use of some numerics to determine  $\sigma^*$ . To find the closed-form formula (13), we need to evaluate a more explicit lower bound for  $\sigma^*$ , noted  $\hat{\sigma}$  and also to choose the value of  $\tau$ .

• an explicit lower bound for  $a(\omega; \varphi, u)$ 

388 • Value of  $\hat{\sigma}$ 

The value of  $\hat{\sigma}$  is obtained by using the inequality for any positive real numbers a, b, c and d:

$$\frac{a+b}{c+d} \ge \min\left\{\frac{a}{c}, \frac{b}{d}\right\},$$

from which is deduced  $\forall (\varphi, u) \in V \times W, (\varphi, u) \neq (0, 0)$ 

$$\frac{a_1(\varphi, u)}{||(\varphi, u)||^2} \ge \min\left\{\frac{\int_{\Omega} (c^2 - U^2) \left|\frac{\partial \varphi}{\partial x}\right|^2 + \frac{c^2}{\alpha^2} \left|\frac{\partial \varphi}{\partial y}\right|^2}{\int_{\Omega} |\varphi|^2}, \frac{\int_{\Gamma} \left|\frac{d^2 u}{dx^2}\right|^2}{\int_{\Gamma} |u|^2}\right\},$$

394 which leads to

$$\sigma^* \ge \hat{\sigma} \equiv \min\left(\lambda_1, \alpha_1^4\right)$$

396 using the definitions (8) and (10). Finally we get

397 
$$\frac{a(\omega;\varphi,u)}{||(\varphi,u)||^2} \ge T$$

398 with T defined by

399 
$$T = \hat{\sigma} - \frac{U}{\tau}\sqrt{\mu\alpha} \left[\hat{\sigma} + \tau^2(1+\tau^2)\frac{c^2}{\alpha^2}\right] + |\omega|^2.$$

400 • Value of  $\tau$ 

401 To get (13), the value of  $\tau$  has to be chosen. If U = 0,  $T(\omega) = \hat{\sigma} + |\omega|^2$  is always 402 positive and we recover that all the resonances are real for a fluid at rest (lemma 7). 403 For U > 0, we choose  $\tau$  such that T remains positive as long as possible when U404 increases. It means that we need

405 
$$\frac{\hat{\sigma} + \tau^2 (1 + \tau^2) \frac{c^2}{\alpha^2}}{\tau},$$

406 to be small. It is straightforward to check that this quantity is minimum for  $\tau = \hat{\beta}$ 407 with

408 (17) 
$$\hat{\beta}^2 = \frac{1}{6} \left( -1 + \sqrt{1 + 12\frac{\hat{\sigma}\alpha^2}{c^2}} \right).$$

409 If  $\hat{\sigma}\alpha^2/c^2$  is large, we get the approximation  $\hat{\beta}^2 \simeq \hat{\sigma}\alpha^2/c^2$ . We have checked numeri-410 cally that  $\hat{\sigma}\alpha^2/c^2$  remains a good approximation of  $\hat{\beta}^2$ , even when it is not large (see 411 Fig. 4). Therefore we choose to define  $\tau$  by the value of  $\beta$  in(12) which leads to the 412 expression (13) of T.

A nice consequence of Theorem (8) is that we can establish a closed-form criterion for the appearance of instabilities. This is done in the next paragraph.

415 **4.3.** A threshold for instabilities existence. Utilizing theorem 8, we can 416 deduce a threshold for the possible existence of an instability and also a bounding 417 of the modulus of the unstable frequencies above the threshold. If we introduce the 418 function

419 (18) 
$$f(\mu, U, c) = U\left(\frac{2}{\beta} + \beta\right)\sqrt{\alpha\mu},$$

420 we have the

430

421 COROLLARY 9. A resonance frequency  $\omega$  is necessarily real if

422 •  $f(\mu, U, c) < 1$ ,

423 •  $f(\mu, U, c) \ge 1$  and  $|\omega|^2 > \hat{\sigma} [f(\mu, U, c) - 1].$ 

424 In other words, an instability  $\omega$  can exist only if  $f(\mu, U, c) > 1$  and  $|\omega|^2 \leq \hat{\sigma} [f(\mu, U, c) - 1].$ 

426 REMARK 3. When it exists, an instability  $\omega$  is necessarily of small modulus  $|\omega|$ 427 since  $\omega$  is located in the complex plane ( $\Re e(\omega), \Im m(\omega)$ ) inside a disc of bounded radius 428  $R = \sqrt{\hat{\sigma} [f(\mu, U, c) - 1]}.$ 

429 *Proof.* This corollary is a direct consequence of theorem 8:

$$T(\omega) > 0 \Rightarrow \Im m(\omega) = 0.$$

The improvement of corollary 9, compared to theorem 8, is that the conditions leading to T > 0 have been written in a closed-form.

Although this corollary is rather explicit, it is not so easy to use in practice because it describes the joint effects of  $\mu$ , U and c. Before presenting numerical validations of the theoretical estimates, in the next section we will simplify corollary 9 by considering separately the different effects.

Π

437 5. Analysis of the stability area. The previous corollary has introduced a
438 "stability area", defined by

439 (19) 
$$f(\mu, U, c) < 1.$$

It corresponds to an unbounded volume in the space  $(U, \mu, c)$  in which no instability can exist and we will now characterize this volume. This volume can be obtained numerically (see Fig. 7) but we wish here to get closed-form results, easier to utilise. To do so, an approximation is required: in Appendix D we prove that  $\lambda_1$  defined in (8) can be approximated by  $\lambda_1 \simeq (c^2 - U^2)\pi^2$ . Thus  $\hat{\sigma}$  defined in (11) can be approximated by

446 (20) 
$$\sigma = \min\left[(c^2 - U^2)\pi^2, \alpha_1^4\right]$$

Since  $\beta$  defined in (12) depends on U and c, the dependence of the volume defined in (19) versus U or c is rather involved. On the contrary, the dependence versus  $\mu$  is explicit: f is an increasing function of  $\mu$  and the relation (19) is obviously not fulfilled for large  $\mu$  values. For its simplicity, we start by analyzing the dependence versus  $\mu$ .

451 **5.1. Influence of**  $\mu$ .  $\mu$  becomes large when the fluid is heavy and/or the plate 452 is light and this is necessary for the possible development of instabilities as stated in 453 the following lemma:

454 LEMMA 10. For U and c fixed, an instability  $\omega$  can exist only if 455 •  $\mu > \mu_c$ ,

456 • 
$$|\omega|^2 < \hat{\sigma}U\left(\frac{2}{\beta}+\beta\right)\sqrt{\alpha}\left(\sqrt{\mu}-\sqrt{\mu_c}\right)$$
, where the critical threshold  $\mu_c$  is defined  
457 by

458 (21) 
$$\mu_c = g(U,c) \equiv \frac{1}{\alpha U^2 \left(\frac{2}{\beta} + \beta\right)^2}$$

459 *Proof.* f defined in (18) is obviously an increasing function of  $\mu$ . Therefore, if we 460 define  $\mu_c$  by  $f(\mu_c, U, c) = 1$  which is equivalent to (21), we deduce that

461 
$$\mu < \mu_c \Leftrightarrow f(\mu, U, c) < 1.$$

462 Then the lemma is just a consequence of Corollary (9).

The validity of this lemma is illustrated in Fig. 4, where are represented the three first resonance frequencies when  $\mu$  varies for  $\alpha = 0.15$ , c = 1 and U = 0.7.  $\Re e(\omega)$ is represented with dots and  $\Im m(\omega)$  with circles. Using (6), the cut-off frequency is  $\omega_c = 7.48$ . Two additional curves represent the function

467 (22) 
$$h: \begin{cases} \mu \to 0 \quad \text{for} \quad \mu < \mu_c, \\ \mu \to \sqrt{\hat{\sigma}U\left(\frac{2}{\beta} + \beta\right)\sqrt{\alpha}\left(\sqrt{\mu} - \sqrt{\mu_c}\right)} \quad \text{for} \quad \mu > \mu_c, \end{cases}$$

468 where  $\mu_c = 0.35$  is determined with (21). The difference between the two curves is the 469 value of  $\sigma$  and of  $\beta$ : the straight line is an approximated curve evaluated with (20) 470 and (12) whereas the dashed line is exact and corresponds to the use of (11) and (17). 471 As announced previously, (20) and (12) are confirmed to be very good approximations 472 of (11) and (17).

Fig. 4 validates Lemma 10: the instability develops for values of  $\mu > 2$ , larger than  $\mu_c$  and the modulus of the unstable frequency (here  $|\Im m(\omega_1)| = |\omega_1|$  since  $\Re e(\omega_1) = 0$ for the instability) is below the solid and dashed curves.



FIG. 4. Three lowest resonance frequencies  $\omega_n$ , n = 1, 2 and 3 when  $\mu$  varies for  $\alpha = 0.15$ , c = 1 and U = 0.7.  $\Re e(\omega)$  with dots and  $\Im m(\omega)$  with circles. Upper boundary  $h(\mu)$  for  $|\omega|$ , defined in (22): solid line for the approximated boundary and dashed line for the exact one

**5.2.** Influence of U and c. We study now how varies f versus U and c. To 476 simplify the study and get monotonous behaviors, we restrict ourselves to the case 477  $\alpha \pi < \sqrt{2}$  which means that the plate is long enough  $(\sqrt{2}L > \pi h)$ . 478

LEMMA 11. For  $\alpha \pi < \sqrt{2}$ ,  $f(\mu, U, c)$  is an increasing function of U and for values 479of c such that  $c < \alpha_1^2/\pi$ , f is a decreasing function of c. In other words, an instability 480can occur only for U large enough or c small enough. 481

482 *Proof.* We note 
$$\theta(x) = \frac{2}{x} + x$$
 such that  $f$  in (18) is defined by

$$f(\mu, U, c) = U\theta(\beta)\sqrt{\alpha\mu}.$$

Recalling that there is no instability if 
$$f(\mu, U, c) < 1$$
, we need to evaluate  $\beta$  from (12)  
to conclude. The results depend on the value of  $\sigma = \min\left[(c^2 - U^2)\pi^2, \alpha_1^4\right]$  and thus

on the sign of  $U^2 + \frac{\alpha_1^4}{\pi^2} - c^2$ : • If  $c^2 < U^2 + \frac{\alpha_1^4}{\pi^2}$ , then  $\sigma = (c^2 - U^2)\pi^2$  and thus  $\beta = \alpha \pi \sqrt{1 - M^2}$ 486 487 • If  $c^2 > U^2 + \frac{\alpha_1^4}{\pi^2}$ , then  $\sigma = \alpha_1^4$  and thus  $\beta = \frac{\alpha_1^2 \alpha}{c} < \alpha \pi \sqrt{1 - M^2}$ 488

Therefore, in all cases,  $\beta \leq \alpha \pi$ . Now we use the fact that  $\theta$  is decreasing for 489  $x < \sqrt{2}$ . Therefore, if  $\alpha \pi < \sqrt{2}$ ,  $\theta(\beta)$  is a decreasing function of  $\beta$ . The conclusion is 490then easy to deduce:  $f \nearrow$  if 491 ∕>,

492 • 
$$U$$
 ,

483

493 • 
$$c^2 < U^2 + \frac{\alpha_1^4}{\pi_A^2}$$
 and  $c \searrow$ 

494 • 
$$c^2 > U^2 + \frac{\alpha_1}{\pi^2}$$
 and  $c \nearrow$ 

495 In particular, if 
$$c < \alpha_1^2/\pi$$
,  $f \nearrow$  when  $c \searrow$ .

.

To illustrate this lemma, we have represented in Fig. 5 isovalues of the function 496g(U,c) defined in (21). The black thick line is the curve  $c^2 = U^2 + \alpha_1^4/\pi^2$ , which 497 corresponds to the change of definition for  $\sigma$  (see (20)). We recall that we have two 498equivalent definitions for the stability area: 499

500 
$$f(\mu, U, c) < 1 \Leftrightarrow \mu < g(U, c).$$



FIG. 5. Isovalues of the function g(U,c) defined in (21) for  $\alpha = 0.15$ . For  $\mu = \mu_0$ , the area  $g(U,c) > \mu_0$  in the (U,c) plane is stable. Curve  $c^2 = U^2 + \alpha_1^4/\pi^2$  in black thick line

Therefore, if  $\mu = \mu_0$ , the stability zone is defined by the area in the (U, c) plane such that  $g(U, c) > \mu_0$ . Values of  $\mu_0$  between 1 and 5 are indicated in Fig. 5. We see that this area reduces when  $\mu$  increases. We see also that we go out the stability zone when increasing U or  $\mu$  or decreasing c if  $c < \alpha_1^2/\pi$ , in accordance with lemma 11.

Contrary to the simple dependence versus  $\mu$ , it is not possible to define, starting from (19), closed-form critical threshold  $U_c$  and  $c_c$  such that an instability can occur only for  $U > U_c$  or  $c < c_c$ . However, by approximating (19), it is possible to derive an approximate value of  $U_c$ , noted  $U_c^{\text{app}}$ , which compares with the one derived in [22].

509 **5.3. Simplifications for the influence of** U**.** Still for  $\alpha \pi < \sqrt{2}$ , since  $f(\mu, U, c)$ 510 is an increasing function of U, we can introduce the threshold frequency  $U_c$  defined 511 by

512 (23) 
$$f(\mu, U_c, c) = 1.$$

513 Let us recall that thanks to lemma 11 we have the following alternative:

• If  $U < U_c \Rightarrow$  no instabilities exist

• If  $U > U_c \Rightarrow$  instabilities can exist

The aim of this paragraph is to look for a closed-form expression of  $U_c$ , as obtained in [22]. We have

LEMMA 12. For 
$$\alpha \pi < \sqrt{2}$$
 and  $c < \alpha_1^2/\pi$ ,  $U_c < U_c^{app}$  where

519 (24) 
$$(U_c^{app})^2 = \frac{\alpha \pi^2 c^2}{4\mu c^2 + \alpha \pi^2},$$

520 and  $U_c$  is defined in (23).

524

521 Proof. The condition on c implies, from (20), that  $\hat{\sigma} = (c^2 - U^2)\pi^2$ . Then we 522 introduce an approximation of f, deduced by cancelling the term  $\beta$  in (18). It is thus 523 defined by

$$f^{\rm app}(\mu, U, c) = U \frac{2}{\beta} \sqrt{\alpha \mu} = \frac{2}{\pi} \sqrt{\frac{\mu U^2 c^2}{\alpha (c^2 - U^2)}},$$

and it is such that  $f^{\text{app}} < f$ . To this approximated function is associated an approximated threshold velocity  $U_c^{\text{app}}$ , defined by  $f^{\text{app}}(\mu, U_c^{\text{app}}, c) = 1$ . It leads to (24) which is similar to the value in [22] (4 is replaced by 1 in [22], leading to a less precise upper bound, certainly due to a 1D approximation of the acoustic propagation introduced



FIG. 6. Three lowest resonance frequencies  $\omega_n$ , n = 1, 2, 3 when U varies for  $\alpha = 0.15$ .  $\Re e(\omega)$  with dots and  $\Im m(\omega)$  with circles. Upper boundary v(U) for  $|\omega|$ , defined in (25), in solid line. (a) c = 1 and  $\mu = 10$ , (b) c = 5 and  $\mu = 1$ 

in [22]). Finally,  $U_c^{\text{app}}$  is less precise than  $U_c$  since it is an upper bound for  $U_c$ : indeed  $f^{\text{app}} < f$  and  $f^{\text{app}}$  is an increasing function of U.

6. Numerical validation of the estimates. The aim of the numerical study is mostly to validate corollary 9. First we look the influence of the velocity U. Then we illustrate the validity of the stability area defined in (19).

**6.1. Threshold for the velocity.** In Fig. 6 are plotted the three first resonance frequencies when U varies between 0 and c for  $\alpha = 0.15$  and for two values of  $(c, \mu)$ .  $\Re e(\omega)$  is represented with dots and  $\Im m(\omega)$  with circles. The green curves represent the function

538 (25) 
$$v: U \to \sqrt{\max\left\{0, \sigma\left[U\left(\frac{2}{\beta} + \beta\right)\sqrt{\alpha\mu} - 1\right]\right\}},$$

involved in corollary 9 with  $\sigma$  and  $\beta$  evaluated thanks to (20) and (12). Corollary 9 asserts that an instability  $\omega$  can not exist if v(U) = 0 and that, when it exists, it must satisfy

542  $|\omega| \le v(U).$ 

These results are validated since the circles are found below the continuous solid curves. Note that in Fig. 6(b), at  $\mu = 4.3$ , two purely imaginary frequencies merge to produce a frequency  $\omega_{1+2}$  with non-zero real and imaginary parts for  $\mu > 4.3$  (as in Fig. 3(b)). The modulus of the complex frequency  $\omega_{1+2}$  is not represented but it is clearly below the values of the green curve.

**6.2.** Stability area. Now we focus on the shape of the stability area, defined by 548 (19),  $f(\mu, U, c) < 1$ , versus  $\mu, U$  and c. Lemmas 10 and 11 indicate that the stability 549area corresponds to  $\mu$  or U small enough (the behavior versus c is more involved). In other words, we expect instabilities to appear when increasing  $\mu$  or U. We will see numerically that it is the case and that instabilities only develop outside the stability 552553 area, confirming the validity of the theoretical estimates. In Fig. 7 are plotted two isovalues of the function  $f(\mu, U, c)$  defined in (18) for  $\alpha = 0.15$  and with one parameter 554fixed: Fig. 7 (a) is for c = 1 and Fig. 7 (b) for  $\mu = 1$ . The isovalue  $f(\mu, U, c) = 1$  is 555represented to show the borderline of the theoretical stability area. The isovalue f = 2556is included to clarify the location of the theoretical stability area  $f(\mu, U, c) < 1$  (it is



FIG. 7. Isovalues of  $f(\mu, U, c)$  defined in (18) for  $\alpha = 0.15$ . (a) c = 1, (b)  $\mu = 1$ .

enough since f has monotonous variations). The black squares represent the couple of values obtained numerically,  $(\mu, U)$  or (c, U), for which an instability appears. It means that the area above the black squares is an unstable area associated with the effective existence (found numerically) of instabilities. Fig. 7 confirms that the lower area  $f(\mu, U, c) < 1$  (in dark) corresponds to a stability area since the black squares are outside this area. Note that the "true" stable area, found numerically, is larger and in particular it contains the light area located between f = 1 and f = 2.

7. Conclusion. We have studied the trapped modes localized around an elastic 565placed in the center of a rigid waveguide and surrounded by a compressible 566 fluid in a uniform flow. The case of real resonance frequencies was already known 567 and we have extended the study to the search of complex frequencies corresponding 568 569 to the development of instabilities. We have mainly proved two results: in the space  $(\mu, U, c)$ , there is a "stability" area in which no instability can develop. In particular it corresponds to  $\mu < \mu_c$  or  $U < U_c$  where  $\mu_c$  and  $U_c$  are threshold values. Moreover, 571outside this stability area, if an instability of frequency  $\omega$  develop, necessarily  $\omega$  lies 572in a disc of bounded radius in the complex plane. A numerical study has confirmed 573these results. 574

A point that has not been discussed in the paper is the quality of the threshold estimates. In practice it appears that the theoretical threshold are rather far from the 576"real" thresholds found numerically, as shown in Fig. 7. For instance, instabilities 577 are found numerically to develop for  $U > U_c^{num}$  and the theoretical threshold  $U_c$ , 578obtained in closed form, is not a good approximation of  $U_c^{num}$  (see Fig. 6 (a) where 579 $U_c = 0.18$  and  $U_c^{num} = 0.4$ ). We do not know how to improve our estimates since we 580have already used Poincaré's inequalities with adjustable parameters (see Eq. (16) 581and (17) to be the most accurate as we could. However, our results are useful since 582they define a stability area which, although it could be larger, gives a security area 583 to avoid instabilities, which is what we wish the most in practice. 584

# 585 Appendix A. Modes of the duct.

A.1. Characterization of the modes. The acoustic propagation in the half duct without plate, in presence of a uniform flow, can be completely described by computing the modes of the duct. These modes are the solutions with separated 589 variables  $(\varphi(x, y) = e^{i\beta x}\theta(y))$  of:

590

$$\begin{cases} \left(\frac{\partial^2}{\partial x^2} + \frac{1}{\alpha^2}\frac{\partial^2}{\partial y^2}\right)\varphi &= \frac{1}{c^2}\left(U\frac{\partial}{\partial x} - i\omega\right)^2\varphi \quad \text{in} \quad \Omega\\ \frac{\partial\varphi}{\partial y}(x,1) &= 0 \quad \text{on} \quad \Sigma,\\ \varphi(x,0) &= 0 \quad \text{for} \quad x \in \mathbb{R}. \end{cases}$$

591 Introducing the transverse functions

592 
$$\theta_n(y) = \sqrt{2} \sin\left[\pi \left(n - \frac{1}{2}\right)y\right], n \ge 1,$$

forming a basis of  $\{\theta(y) \in H^1(0,1)/\theta(0) = 0\}$ , the modes read

594 
$$\varphi_n^{\pm}(x,y) = e^{i\beta_n^{\pm}x}\theta_n(y),$$

595 with

596 
$$\beta_n^{\pm} = \frac{-kM \pm \gamma_n}{1 - M^2} \text{ where } k = \frac{\omega}{c}$$

597 (26) 
$$\gamma_n = \sqrt{k^2 - (1 - M^2)\zeta_n^2},$$

598 
$$\zeta_n = \frac{\pi}{\alpha} \left( n - \frac{1}{2} \right), n \ge 1.$$

To define  $\gamma_n$ , the chosen definition of the complex square root is:  $\sqrt{z} = \sqrt{r}e^{i\theta/2}$  for  $z = re^{i\theta}, \theta \in [0, 2\pi[$ . This definition of the complex square root leads to  $\Im m(\gamma_n) \ge 0$ for any  $\omega \in \mathbb{C}$ .

602 The characteristics of the modes depend on the nature of  $\omega: \omega \in \mathbb{R}$  or  $\omega \in \mathbb{C} \setminus \mathbb{R}$ . 603 • For real values of  $\omega \in \mathbb{R}$ , we get

604

$$\gamma_n = \sqrt{k^2 - (1 - M^2)\zeta_n^2} \text{ for } n - \frac{1}{2} < \frac{k\alpha}{\pi\sqrt{1 - M^2}},$$
$$\gamma_n = i\sqrt{(1 - M^2)\zeta_n^2 - k^2} \text{ for } n - \frac{1}{2} > \frac{k\alpha}{\pi\sqrt{1 - M^2}},$$

605

 $\gamma_n = i\sqrt{(1-M^2)\zeta_n^2 - k^2} \text{ for } n - \frac{1}{2} > \frac{k\alpha}{\pi\sqrt{1-M^2}}.$ 

Therefore a finite number of propagative modes  $(\beta_n^{\pm} \in \mathbb{R})$  exist and an infinite number of evanescent modes  $(\beta_n^{\pm} \in i\mathbb{R} \text{ with } \pm \Im m(\beta_n^{\pm}) > 0)$  exist. The + modes "propagate" downstream: when they are propagative, they correspond to a positive group velocity and when they are evanescent, they decrease when  $x \to \infty$ . Symmetrically, the – modes propagate upstream.

For low frequencies,  $k < k_c \equiv \pi \sqrt{1 - M^2}/(2\alpha)$ , the so-called cut-off wave number of the half-guide, all the modes are evanescent and this is a good framework when looking for trapped modes.

• For complex values of  $\omega, \omega \in \mathbb{C} \setminus \mathbb{R}$ , all the modes are evanescent since it is straightforward to verify that  $\pm \Im m(\beta_n^{\pm}) > 0$  and the + modes are downstream modes whereas the - modes are upstream modes.

617 **A.2. Transparent boundary conditions.** Let us point out that the duct 618 modes are useful since every fluid vibration propagating outside the duct part in-619 cluding the plate can be decomposed on these modes. More precisely, if we introduce 620 the vertical boundaries

621 
$$\Sigma_R^{\pm} = \{(x, y), x = \pm R, y \in ]0, 1[\},$$

and remembering that the plate is of length 1, we have the

623 LEMMA 13. For any R > 1, for any  $\pm x \ge R$ , a solution of (2) belonging to 624  $H^1(\Omega)$  can be written:

625 (27) 
$$\varphi(x,y) = \sum_{n \ge 1} a_n^{\pm} e^{i\beta_n^{\pm}(x \mp R)} \theta_n(y),$$

626 with

627

637

642

643

$$a_n^{\pm} = (\varphi, \theta_n)_{\Sigma_R^{\pm}} \equiv \int_0^1 \varphi(\pm R, y) \theta_n(y) dy.$$

628 *Proof.* For any  $x \notin [0, 1]$ , which corresponds to be located in the duct outside the 629 plate, the velocity potential can be decomposed on the duct modes:

630 
$$\varphi(x,y) = \sum_{n \ge 1} a_n^{\pm} e^{i\beta_n^{\pm}(x \mp R)} \theta_n(y).$$

631 The coefficient  $a_n^{\pm}$  are deduced, using the orthogonality of the transverse functions

$$(\theta_n, \theta_m)_{\Sigma_R^{\pm}} = \delta_{mn} \text{ for any } n, m \ge 1.$$

Using (27), we can define (exact) transparent boundary conditions thanks to DtN (Dirichlet to Neumann) operators  $T^{\pm}$ : for any R > 1,

635 (28) 
$$\begin{array}{rcl} \frac{\partial \varphi}{\partial x}(x=\pm R,y) &=& T^{\pm}\varphi(x=\pm R,y) \text{ with} \\ T^{\pm}: & H^{1/2}(\Sigma_R^{\pm}) \to H^{-1/2}(\Sigma_R^{\pm}), \\ \varphi(\pm R,y) \to \pm i\beta_n^{\pm} \ (\varphi,\theta_n)_{\Sigma_R^{\pm}} \theta_n(y) \end{array}$$

# 636 Appendix B. Proof of lemma 3 for the continuous resonance problem.

To prove that if  $\omega$  is a resonance frequency, then  $\overline{\omega}$  is also a resonance frequency, we proceed in different steps:

- 640 for the resonance problem (2) set in an unbounded domain, we define an 641 equivalent formulation (29) set in a bounded domain,
  - in the bounded domain, thanks to a compacity property we prove that the resonance problem is of Fredholm type,
- we conclude using the fact that  $A^*(\omega) = A(\overline{\omega})$  and the property Ker  $A^*(\omega) =$ (Ran  $A(\omega)$ )<sup> $\perp$ </sup>.

**B.1. Resonance problem set in a bounded domain.** Thanks to the definition of the radiation conditions at finite distance (28), we can introduce two equivalent ways to define the trapped modes: in an unbounded domain or in a domain bounded by radiation conditions. The unbounded problem has been defined in (2). The problem can also be set in a bounded domain, using the DtN operator (28) to write transparent boundary conditions on the boundaries of the domain. In this aim we note for any R > 1 (to include the whole plate)

653 
$$\Omega_R = \{(x, y) \in \Omega, |x| < R\} \text{ and } \Gamma_R^{\pm} = \{(x, y) \in \Gamma^{\pm}, |x| < R\},\$$

654 the bounded domains and we introduce the new solution space

655 
$$V_R = \{ \varphi \in H^1(\Omega_R), \varphi = 0 \text{ on } \Gamma_R^{\pm} \}.$$

Then the problem of finding trapped mode takes the form: Find  $(\varphi, u) \in V_R \times W$ 656 657 such that

$$(29) \begin{cases} \left(\frac{\partial^2}{\partial x^2} + \frac{1}{\alpha^2}\frac{\partial^2}{\partial y^2}\right)\varphi &= \frac{1}{c^2}\left(U\frac{\partial}{\partial x} - i\omega\right)^2\varphi \quad \text{in} \quad \Omega_R, \\ \left(\frac{d^4}{dx^4} - \omega^2\right)u &= 2\mu\left(U\frac{\partial}{\partial x} - i\omega\right)\varphi(x,0) \quad \text{on} \quad \Gamma, \\ \frac{1}{\alpha}\frac{\partial\varphi}{\partial y}(x,0) &= \left(U\frac{\partial}{\partial x} - i\omega\right)u \quad \text{on} \quad \Gamma, \\ \frac{\partial\varphi}{\partial x} &= \pm i\beta_n^{\pm} (\varphi,\theta_n)_{\Sigma_R^{\pm}}\theta_n(y) \quad \text{on} \quad \Sigma_R^{\pm}, \\ \frac{\partial\varphi}{\partial y}(x,1) &= 0 \quad \text{on} \quad \Sigma_R, \\ u(0) = 0 = u'(0) \quad \text{and} \quad u''(1) = 0 = u^{(3)}(1). \end{cases}$$

To justify this problem we will prove that (29) is equivalent to (2). 659

B.2. Equivalence between the bounded and unbounded problems. Prob-660 661 lems (2) and (29) are equivalent. One implication is easy: (2) implies (29) since (29)is obtained by restricting the solution of (2) to  $\Omega_R$ . The reverse is less obvious but 662is true since a solution of (29) can be extended in a solution of (2), using (27). The 663 only difficulty is to check that the extended solution is in  $H^1(\Omega)$ . When  $\omega \in \mathbb{C} \setminus \mathbb{R}$ , 664665 the fact that all the duct modes are evanescent insures that the extension belongs to  $H^1(\Omega)$ . When  $\omega \in \mathbb{R}$ , we have first to prove that the solution of (29) does not excite 666 the propagating modes. In this aim, we write the variational form of (29): 667

668 (30) Find 
$$(\varphi, u) \in V_R \times W$$
 such that  $a_R[\omega; (\varphi, u), (\psi, v)] = 0$  for all  $(\psi, v) \in V_R \times W$ ,

669 with

670 
$$a_R[\omega;(\varphi,u),(\psi,v)] = a_{fluid}[\omega;(\varphi,u),(\psi,v)] + a_{plate}[\omega;(\varphi,u),(\psi,v)],$$

671 and

$$a_{fluid}[\omega;(\varphi,u),(\psi,v)] =$$

$$\int_{\Omega_R} \left[ (1 - M^2) \frac{\partial \varphi}{\partial x} \frac{\overline{\partial \psi}}{\partial x} + \frac{1}{\alpha^2} \frac{\partial \varphi}{\partial y} \frac{\overline{\partial \psi}}{\partial y} - i \frac{\omega}{c} M \left( \frac{\partial \varphi}{\partial x} \overline{\psi} - \varphi \frac{\overline{\partial \psi}}{\partial x} \right) - \frac{\omega^2}{c^2} \varphi \overline{\psi} \right]$$

672

675 
$$+\frac{1}{\alpha}\int_{\Gamma}\overline{\psi}\left(U\frac{\partial u}{\partial x}-i\omega u\right)-i\sum_{n\geq 1}\gamma_{n}(\omega) \ (\varphi,\theta_{n})_{\Sigma_{R}^{\pm}}(\theta_{n},\psi)_{\Sigma_{R}^{\pm}},$$

with  $\gamma_n$  defined in (26) and 676

677 
$$a_{plate}[\omega;(\varphi,u),(\psi,v)] = \frac{1}{2\mu\alpha} \int_{\Gamma} \left( \frac{d^2u}{dx^2} \frac{d^2\overline{v}}{dx^2} - \omega^2 u\overline{v} \right) + \frac{1}{\alpha} \int_{\Gamma} \varphi \left( U \frac{\partial\overline{v}}{\partial x} + i\omega\overline{v} \right).$$

Then, choosing  $(\psi, v) = (\varphi, u)$ , we take the imaginary part of (30) 678

679 
$$\Im m\left(a_{R}[\omega;(\varphi,u),(\varphi,u)]\right) = 0, \text{ which leads to } \sum_{n \ge 1} \Re e\left(\gamma_{n}\right) |(\varphi,\theta_{n})_{\Sigma_{R}^{\pm}}|^{2} = 0.$$

Since  $\Re e(\gamma_n) = 0$  for the evanecent modes for  $\omega \in \mathbb{R}$ , it leads to  $(\varphi, \theta_n)_{\Sigma_R^{\pm}} = 0$  for 680 all the propagating modes. Therefore the extended solution (27) is purely evanescent 681

and belongs to  $H^1(\Omega)$  and we have proved that (29) implies (2). 682

683 **B.3. Fredholm alternative.** Thanks to the Riesz theorem we can introduce 684 the operator  $A(\omega)$  such that for all  $(\varphi, u)$  in  $V_R \times W$  and all  $(\psi, v)$  in  $V_R \times W$ ,

$$(A(\omega)(\varphi, u), (\psi, v))_{V_R \times W} = a_R[\omega; (\varphi, u), (\psi, v)],$$

686 where we use the scalar product on  $V_R \times W$ :

$$((\varphi, u), (\psi, v))_{V_R \times W} =$$

$$\int_{\Omega_R} \left[ \frac{\partial \varphi}{\partial x} \frac{\overline{\partial \psi}}{\partial x} + \frac{\partial \varphi}{\partial y} \frac{\overline{\partial \psi}}{\partial y} + \varphi \overline{\psi} \right] dx dy + \int_{\Gamma} \left( \frac{d^2 u}{dx^2} \frac{d^2 \overline{v}}{dx^2} + \frac{du}{dx} \frac{d\overline{v}}{dx} + u \overline{v} \right) dx$$

690 Then the problem (30) can be written in an operator form: find  $\omega \in \mathbb{C}$  such that 691 exists  $(\varphi, u)$  in  $V_R \times W$  satisfying  $A(\omega)(\varphi, u) = 0$ . The operator A is decomposed as 692 A = B + C where

693 • 
$$(B(\omega)(\varphi, u), (\psi, v))_{V_R \times W} = \int_{\Omega_R} \left[ (1 - M^2) \frac{\partial \varphi}{\partial x} \frac{\partial \psi}{\partial x} + \frac{1}{\alpha^2} \frac{\partial \varphi}{\partial y} \frac{\partial \psi}{\partial y} + \varphi \overline{\psi} \right],$$

694

695 
$$\frac{1}{2\mu\alpha}\int_{\Gamma}\left(\frac{d^2u}{dx^2}\frac{d^2\overline{v}}{dx^2}+u\overline{v}\right)-i\sum_{n\geq 1}\gamma_n(\omega) \ (\varphi,\theta_n)_{\Sigma_R^{\pm}}(\theta_n,\psi)_{\Sigma_R^{\pm}},$$

696 • 
$$(C(\omega)(\varphi, u), (\psi, v))_{V_R \times W} = \int_{\Omega_R} \left[ -i\frac{\omega}{c}M\left(\frac{\partial\varphi}{\partial x}\overline{\psi} - \varphi\overline{\frac{\partial\psi}{\partial x}}\right) - \left(1 + \frac{\omega^2}{c^2}\right)\varphi\overline{\psi} \right],$$

697

$$+\frac{1}{\alpha}\int_{\Gamma}\overline{\psi}\left(U\frac{\partial u}{\partial x}-i\omega u\right)-\frac{1+\omega^2}{2\mu\alpha}\int_{\Gamma}u\overline{v}+\frac{1}{\alpha}\int_{\Gamma}\varphi\left(U\frac{\partial\overline{v}}{\partial x}+i\omega\overline{v}\right).$$

699 The sesquilinear form  $b_R[\omega; (\varphi, u), (\psi, v)] = (B(\omega)(\varphi, u), (\psi, v))_{V_R \times W}$  is coercive on 700  $V_R \times W$  which proves that  $B(\omega)$  is an isomorphism of  $V_R \times W$ . Indeed

$$\exists C_R > 0, \forall (\varphi, u) \in V_R \times W, \Re e\left(b_R[\omega; (\varphi, u), (\varphi, u)]\right) \ge C_R ||(\varphi, u)||^2_{V_R \times W}.$$

702 The key point is that

703 
$$\Re e\left(-i\sum_{n\geq 1}\gamma_n(\omega) \ (\varphi,\theta_n)_{\Sigma_R^{\pm}} (\theta_n,\varphi)_{\Sigma_R^{\pm}}\right) = \sum_{n\geq 1} \Im m\left(\gamma_n\right) \ | \ (\varphi,\theta_n)_{\Sigma_R^{\pm}} |^2 \ge 0,$$

since  $\Im(\gamma_n(\omega)) \ge 0$ . It is also straightforward to prove that  $C(\omega)$  is a compact operator on  $V_R \times W$ , the key point being the Rellich theorem which applies since  $\Omega_R$ is bounded.

The resonance problem  $(B + C)(\varphi, u) = 0$  can be written  $(I + B^{-1}C)(\varphi, u) = 0$ because B is an isomorphism. And finally since  $B^{-1}C$  is a compact operator, the Fredholm alternative holds for this equation.

710 **B.4. Characteristics of the eignevalues.** The Fredholm property obtained 711 in the previous paragraph implies in particular, for the operator  $S = I + B^{-1}C$  but 712 also for the operator  $A \equiv BS = B + C$  (since B is an isomorphism), that

713 • Ker A( $\omega$ ) = {0}  $\Leftrightarrow$  Ran A( $\omega$ ) = V<sub>R</sub> × W,

• Ker  $A^{\star}(\omega) = (\operatorname{Ran} A(\omega))^{\perp}$  where  $A^{\star}(\omega)$  is the adjoint of  $A(\omega)$ .

715 Moreover it is easy to prove  $A^*(\omega) = A(\overline{\omega})$ . This is in particular due to  $\gamma_n(\overline{\omega}) = 716 - \overline{\gamma_n(\omega)}$ .

Therefore, combining together the previous results, we get  $\operatorname{Ker} A(\omega) = \{0\} \Leftrightarrow$ Ran  $A(\omega) = V_R \times W \Leftrightarrow \operatorname{Ker} A^*(\omega) = \{0\}$ . In other words  $\operatorname{Ker} A(\omega) \neq \{0\} \Leftrightarrow$ Ker  $A(\overline{\omega}) \neq \{0\}$  and thus if  $\omega$  is a resonance frequency, then  $\overline{\omega}$  is also a resonance frequency.

Finally, we have proved that if  $\omega$  is an eigenvalue then  $\pm \overline{\omega}$  are also eigenvalues. But then, since  $\overline{\omega}$  is an eigenvalue,  $\pm \omega$  are also eigenvalues which ends the proof of lemma 3.

# 724 Appendix C. Numerical method.

We start from the formulation of the trapped modes problem set in a bounded domain (30). Noting  $(w_j)_{j=1...N_{dof}}$  the finite element basis (P1 elements) with  $N_{dof}$ the number of degrees of freedom, the velocity potential is sought under the form

728 
$$\varphi = \sum_{j=1}^{N_{\text{dof}}} \varphi_j w_j(x, y),$$

where the nodal values  $\varphi_j$  are unknowns. Also using the modal basis of the plate  $(\xi_m)_{m\geq 1}$  Eq. (33), the displacement of the plate is sought under the form

731 
$$u = \sum_{n=1}^{N_{\rm pl}} u_n \xi_n(x),$$

where the modal components  $u_n$  are unknowns and  $N_{\rm pl}$  is the number of plate modes taken into account. Then, developing

734 
$$a_{fluid}\left[\omega; \left(\sum_{j=1}^{N_{dof}} \varphi_j w_j, \sum_{n=1}^{N_{pl}} u_n \xi_n\right), (w_i, \xi_m)\right] = 0,$$

735 and

736 
$$a_{plate}\left[\omega; \left(\sum_{j=1}^{N_{dof}} \varphi_j w_j, \sum_{j=1}^{N_{pl}} u_n \xi_n\right), (w_i, \xi_m)\right] = 0,$$

for all i = 1 to  $N_{\text{dof}}$  and all m = 1 to  $N_{\text{pl}}$ , we are lead to define the matrices

738 
$$D_{ij} = (1 - M^2) \left( \frac{\partial w_j}{\partial x}, \frac{\partial w_i}{\partial x} \right)_{\Omega_R} + \frac{1}{\alpha^2} \left( \frac{\partial w_j}{\partial y}, \frac{\partial w_i}{\partial y} \right)_{\Omega_R},$$

739 
$$E_{ij} = \frac{M}{c} \left[ \left( w_j, \frac{\partial w_i}{\partial x} \right)_{\Omega_R} - \left( \frac{\partial w_j}{\partial x}, w_i \right)_{\Omega_R} \right],$$

740 
$$M_{ij}(\omega) = -i \sum_{p=1}^{N_{\rm DtN}} \gamma_p(\omega) \zeta_p (w_j, \theta_p)_{\Sigma_R^{\pm}} (\theta_p, w_i)_{\Sigma_R^{\pm}}$$
741

742 and

743 
$$F_{ij} = \left(-\frac{1}{c^2}\right) (w_j, w_i)_{\Omega_R}, \quad G_{in} = \frac{U}{\alpha} \left(\frac{d\xi_n}{dx}, w_i\right)_{\Gamma}, \quad H_{in} = \left(-\frac{1}{\alpha}\right) (\xi_n, w_i)_{\Gamma},$$
744

745 
$$I_{mn} = \frac{1}{2\mu\alpha} \left( \frac{d^2\xi_n}{dx^2}, \frac{d^2\xi_m}{dx^2} \right)_{\Gamma}, \quad J_{mn} = \left( -\frac{1}{2\mu\alpha} \right) (\xi_n, \xi_m)_{\Gamma}.$$

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 $_{746}$   $N_{\rm DtN}$  is the number of term taken to describe the DtN operators. Thanks to the

747 defined matrices, the discretized resonance problem is

748 (31) 
$$\begin{pmatrix} D+i\omega E+\omega^2 F+M(\omega) & G+i\omega H\\ G^T-i\omega H^T & I+\omega^2 J \end{pmatrix} \begin{pmatrix} \Phi\\ U \end{pmatrix} = 0,$$

749 where we have introduced the vectors of unknowns

750 
$$\Phi = (\varphi_1, \cdots, \varphi_{N_{\text{dof}}})^T \text{ and } U = (\xi_1, \cdots, \xi_{N_{\text{pl}}})^T.$$

This is a non-linear eigenvalue problem because of the matrix  $M(\omega)$  which includes  $\gamma_n(\omega)$  defined in Eq. (26) as a square root of  $\omega$ . To simplify the study, we have approximated the matrix M to get a quadratic eigenvalues problem. Several approaches are possible and we have checked that they give very similar results. The simplest possibility is to replace the transparent boundary conditions (28) by homogeneous Neumann or Dirichlet boundary conditions:

757 
$$\frac{\partial \varphi}{\partial x} = 0 \text{ or } \varphi = 0 \text{ on } \Sigma_R^{\pm}$$

Since the velocity potential decreases exponentially away from the plate, these approximations are very good if  $\Sigma_R^{\pm}$  is far enough from the plate. In both cases, we get that the matrix M = 0. The Dirichlet case requires an extra element: the finite element basis  $(w_j)_{j=1...N_{dof}}$  must satisfy the Dirichlet condition.

We have chosen another approach which consists in replacing  $\gamma_n = \sqrt{k^2 - (1 - M^2)\zeta_n^2}$ by the approximation  $\tilde{\gamma}_n = i\sqrt{1 - M^2}\zeta_n$ . This is the natural approximation when  $n \to \infty$ . It leads to the approximation  $\tilde{M}$  of the matrix M, independent of the frequency and defined by

766 
$$\tilde{M}_{ij} = \sqrt{1 - M^2} \sum_{p=1}^{N_{\text{DtN}}} \zeta_p \left( w_j, \theta_p \right)_{\Sigma_R^{\pm}} \left( \theta_p, w_i \right)_{\Sigma_R^{\pm}}.$$

We have chosen this approximation because it is close from the true radiation conditions and it becomes exact for large p values. Moreover it is easy to implement numerically.

To write the quadratic eigenvalue problem as a classical eigenvalue problem, we introduce the matrices:

772 
$$A = \begin{pmatrix} D+M & G \\ G^T & 0 \end{pmatrix}, \quad B = \begin{pmatrix} E & H \\ -H^T & 0 \end{pmatrix}, \quad C = \begin{pmatrix} F & 0 \\ 0 & J \end{pmatrix},$$

where A, B and C are real matrices with  $A^T = A$ ,  $B^T = -B$  (since  $E^T = -E$ ) and  $C^T = C$ . Then (31) reads

$$(A + i\omega B + \omega^2 C)X = 0,$$

776 where X is the vector  $(\Phi, U)^T$ .

The Finite Element matrices are built using a Finite Elements code [33] and the eigenvalues are determined using the Matlab software. For a domain of size  $2 \times 1$ , we use an unstructured triangular mesh, we introduce P1 finite elements and in practice we take  $N_{dof} = 2309$ ,  $N_{pl} = 6$  and  $N_{DtN} = 10$ . We have checked that taking larger values doesn't change significantly the results.

## 782 Appendix D. Simplification of the theoretical estimates.



FIG. 8. Three first eigenvalues  $\lambda_n$  of problem (32) versus U for  $\alpha = 0.15$  and c = 1. Circles: exact values, Solid lines: approximations  $\lambda_n^{app}$ 

Here we show that  $\hat{\sigma}$  defined in (11) can be approximated by (20) that we recall

784 
$$\sigma = \min\left[(c^2 - U^2)\pi^2, \alpha_1^4\right]$$

Indeed the values of  $\lambda_1$  and  $\alpha_1^4$ , introduced in (8) and (10), can be obtained in closed form, as explained now.

• Approximation of  $\lambda_1$ :

Since

788

789

$$\lambda_1 = \inf_{\varphi \in V} R(\varphi),$$

with R defined in (9), using the test field  $\varphi_0 \in V$  defined by

791 
$$\begin{cases} \varphi_0(x,y) = \sin(\pi x) & \text{for} \quad x \in ]0,1[,\\ \varphi_0(x,y) = 0 & \text{for} \quad x \notin ]0,1[, \end{cases}$$

792 we find the upper bound

793 
$$\lambda_1 < R(\varphi_0) = (c^2 - U^2)\pi^2.$$

We have determined numerically that this upper bound  $(c^2 - U^2)\pi^2$  is in fact a good approximation of  $\lambda_1$ . To do so and to find the numerical value of  $\lambda_1$ , we notice that  $\lambda_1$  is also the first eigenvalues of the problem: find  $\varphi \in V$  and  $\lambda \in \mathbb{R}$  such that

797 (32) 
$$\begin{cases} -(c^2 - U^2)\frac{\partial^2\varphi}{\partial x^2} - \frac{c^2}{\alpha^2}\frac{\partial^2\varphi}{\partial y^2} = \lambda\varphi \quad \text{in} \quad \Omega,\\ \frac{1}{\alpha}\frac{\partial\varphi}{\partial y}(x,0) = 0 \quad \text{on} \quad \Gamma,\\ \frac{\partial\varphi}{\partial y}(x,1) = 0 \quad \text{on} \quad \Sigma. \end{cases}$$

Indeed the Min-Max principle indicates that  $\lambda_1$  defined by (8) thanks to the Rayleigh quotient (9) is the first eigenvalue if  $\lambda_1 < (\pi c/2\alpha)^2$  (below the essential spectrum).

For  $\alpha = 0.15$  and c = 1, in Fig. 8 are plotted with solid lines the three first eigenvalues  $\lambda_n$ , n = 1, 2 and 3, of problem (32) versus U below the essential spectrum  $((\pi c/2\alpha)^2 \simeq 110)$ , obtained numerically thanks to a Finite Element method. The circles represent the values  $\lambda_n^{\text{app}} \equiv n^2 \pi^2 (c^2 - U^2)$  versus U for n = 1, 2 and 3. We see that for  $n = 1, \lambda_1^{\text{app}} = \pi^2 (c^2 - U^2)$  approximates very accurately  $\lambda_1$ . On the

contrary, the approximations  $\lambda_n^{\text{app}}$  of  $\lambda_n$  for n = 2 and 3 are not satisfactory (they 805 become better approximations when  $\alpha$  is decreased, result not reported here). 806

• Approximation of  $\alpha_1$ : 807

To determine  $\alpha_1$ , we introduce the following eigenvalue problem: find  $u \in W$  and 808  $\lambda^{\text{plate}} \in \mathbb{R}$  such that 809

810 
$$\frac{d^4u}{dx^4} = \lambda^{\text{plate}} u \quad \text{for} \quad x \in ]0,1[,$$

811 
$$u''(1) = 0 = u^{(3)}(1).$$

This is the equation of the plate in vacuum and the eigenvalues  $\lambda_n^{\text{plate}}$  are equal to the square of the eigenfrequencies  $\omega_n^{\text{plate}}$  of the plate. A simple explicit calculation shows that  $\lambda_n^{\text{plate}} = \alpha_n^4$  with  $\alpha_n$  satisfying [1, 9] 812 813 814

815 
$$\cos \alpha_n \cosh \alpha_n = -1.$$

In particular,  $\alpha_1 = 1.88$ . The eigenmodes are 816

817 (33) 
$$\xi_n(x) = A_n \left[\sinh\left(\alpha_n x\right) - \sin\left(\alpha_n x\right)\right] + B_n \left[\cosh\left(\alpha_n x\right) - \cos\left(\alpha_n x\right)\right],$$

where 818

819 
$$B_n = -A_n \frac{\sinh(\alpha_n) + \sin(\alpha_n)}{\cosh(\alpha_n) + \cos(\alpha_n)},$$

and  $A_n$  can be chosen such that  $\int_{\Gamma} |\xi_n|^2 = 1$ . Finally the Min-Max principle indicates 820

that  $\lambda_1^{\text{plate}}$  satisfies 821

$$\lambda_1^{\text{plate}} = \inf_{u \in W} \frac{\int_{\Gamma} \left| \frac{d^2 u}{dx^2} \right|^2}{\int_{\Gamma} \left| u \right|^2}$$

in accordance with (10). 823

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