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## ▶ To cite this version:

Maxime Tournier, Matthieu Nesme, François Faure, Benjamin Gilles. Velocity-based Adaptivity of Deformable Models. Computers and Graphics, 2014, 45, pp.75-85. 10.1016/j.cag.2014.08.004 . hal-01076338

# HAL Id: hal-01076338 https://inria.hal.science/hal-01076338

Submitted on 5 Dec 2014

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### Velocity-based Adaptivity of Deformable Models

Maxime Tournier<sup>1</sup>, Matthieu Nesme<sup>2</sup>, Francois Faure<sup>2</sup>, Benjamin Gilles<sup>1</sup>

#### Abstract

A new adaptive model for viscoelastic solids is presented. Unlike previous approaches, it allows seamless transitions, and simplifications in deformed states. The deformation field is generated by a set of physically animated frames. Starting from a fine set of frames and mechanical energy integration points, the model can be coarsened by attaching frames to others, and merging integration points. Since frames can be attached in arbitrary relative positions, simplifications can occur seamlessly in deformed states, without returning to the original shape, which can be recovered later after refinement. We propose a new class of velocity-based simplification criterion based on relative velocities. Integration points can be merged to reduce the computation time even more, and we show how to maintain continuous elastic forces through the levels of detail. Such meshless adaptivity allows significant improvements of computation time during simulations. It also provides a natural approach to coarse-to-fine deformable mesh registration.

Keywords: Computer Animation, Physically-based Animation, Deformable Solids, Adaptive Kinematics

The stunning quality of high-resolution physically-based an-2 imations of deformable solids requires complex deformable mod-3 els with large numbers of independent Degrees Of Freedom 4 (DOF) which result in large equation systems for solving dy-5 namics, and high computation times. On the other hand, the 6 thrilling user experience provided by interactive simulations can 7 only be achieved using fast computation times which preclude 8 the use of high-resolution models. Reconciling these two con-<sup>9</sup> tradictory goals requires adaptive models to efficiently manage <sup>10</sup> the number of DOFs, by refining the model where necessary 11 and by coarsening it where possible. Mesh-based deformations 12 can be seamlessly refined by subdividing elements and interpo-13 lating new nodes within these. However, seamless coarsening 14 can be performed only when the fine nodes are back to their 15 original position with respect to their higher-level elements, <sup>16</sup> which only happens in the locally undeformed configurations 17 (i.e. with null strain). Otherwise, a popping artifact (i.e. an in-18 stantaneous change of shape) occurs, which not only violates <sup>19</sup> the laws of physics, but it is also visually disturbing for the user. <sup>20</sup> Simplifying objects in deformed configurations, as presented in <sup>21</sup> Fig. 1c, has thus not been possible with previous adaptive ap-<sup>22</sup> proaches, unless the elements are small or far enough from the 23 user. This may explain why extreme coarsening has rarely been 24 proposed, and adaptive FEM models typically range from mod-25 erate to high complexity.

We introduce a new approach of adaptivity to mechanically r simplify objects in arbitrarily deformed configurations, while exactly maintaining their current shape and controlling the velocity discontinuity, which we call seamless adaptivity. It extends a frame-based meshless method and naturally exploits the ability to attach frames to others in arbitrary relative positions, as illustrated in Fig. 2. In this example, a straight beam is ini-

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Figure 2: Seamless coarsening in a deformed state. Left: reference shape, one active frame in black, and a passive frame in grey attached using a relative transformation (dotted line). Middle: activating the frame let it to move freely and deform the object. Right: deactivated frame in a deformed configuration using an offset  $\delta X$ .

33 tially animated using a single moving frame, while another con-<sup>34</sup> trol frame is attached to it. We then detach the child frame to 35 allow the beam to bend as needed. If the beam deformation 36 reaches a steady state, the velocity field can again be obtained 37 from the moving frame alone, and the shape can be frozen in 38 the deformed state by applying an offset to the child frame ref-<sup>39</sup> erence position relative to the moving frame. Setting the offset 40 to the current relative position removes mechanical DOFs with-41 out altering the current shape of the object. This deformation is 42 reversible. If the external loading applied to the object changes, 43 we can mechanically refine the model again (*i.e.* activate the 44 passive frame) to allow the object to recover its initial shape or 45 to undergo new deformations. The ability to dynamically adapt <sup>46</sup> the deformation field even in non-rest configuration is the spe-47 cific feature of our approach, which dramatically enhances the 48 opportunities for coarsening mechanical models compared with 49 previous methods.

Our specific contributions are (1) a deformation method based on a generalized frame hierarchy for dynamically tuning the complexity of deformable solids with seamless transitions; (2) a novel simplification and refinement criterion based on veloctity, which allows us to simplify the deformation model in deformed configurations, and (3) a method to dynamically adapt the integration points and enforce the continuity of forces across of changes of resolution.



(a) Rest state

(b) Compressing

(c) Equilibrium

(d) Decompressing

Figure 1: Deformable Christmas tree with our proposed adaptive deformation field. (1a): One frame is sufficient in steady state. (1b): When ornaments are attached, additional frames are activated to allow deformations. (1c): The velocity field can be simplified again when the equilibrium is reached. Note that our method can simplify locally deformed regions. (1d): Once the branches are released, the velocity field is refined again to allow the branches to recover their initial shape.

The present article extends an earlier conference version [1] by adding new results on deformable mesh registration (6.4), more derivations concerning metrics (4.1), and providing more details based on reviewer comments. The remainder is organized as follows. We summarize the original frame-based simusal lation method and introduce notations in Section 2. An overview of our adaptive framework is presented in Section 3. We formalize and discuss different criteria for nodal adaptivity in Section 4. The adaptivity of the integration points is then introduced in Section 5. Results obtained with our method are presented and discussed in Section 6, including an application to of deformable mesh registration, and we conclude in Section 7 with future work.

#### 71 1. Related Work

The simulation of viscoelastic solids is a well-studied prob-73 lem in computer graphics, starting with the early work of Ter-74 zopoulos et al. [2]. A survey can be found in [3]. Frame-based 75 models have been proposed [4, 5, 6, 7], and the impressive ef-76 ficiency of precomputed reduced models has raised a growing 77 interest [8, 9, 10, 11, 12, 13], but run-time adaptivity remains a 78 challenge. The remainder of this review focuses on this issue.

<sup>79</sup> Hutchinson et al. [14] and Ganovelli et al. [15] first com-<sup>80</sup> bined several resolutions of 2D and 3D solids dynamically de-<sup>81</sup> formed by mass-springs. Cotin et al. [16] combined two me-<sup>82</sup> chanical models to simulate various parts of the same object. <sup>83</sup> Most adaptive methods are based on meshes at multiple resolu-<sup>84</sup> tions. Mixing different mesh sizes can result in T-nodes that are <sup>85</sup> mechanically complex to manage in the Finite Element Method <sup>86</sup> (FEM). Wu et al. [17] chose a decomposition scheme that does <sup>87</sup> not generate such nodes. Debunne et al. [18] performed the <sup>89</sup> al. [19] showed that hierarchical shape functions are a generic <sup>90</sup> way to deal with T-nodes. Sifakis et al. [20] constrained T-<sup>91</sup> nodes within other independent nodes. Martin et al. [21] solved <sup>92</sup> multi-resolution junctions with polyhedral elements. Several

<sup>93</sup> authors proposed to generate on the fly a valid mesh with dense <sup>94</sup> and fine zones. Real-time remeshing is feasible for 1D ele-<sup>95</sup> ments such as rods and wires [22, 23, 24] or 2D surfaces like <sup>96</sup> cloth [25]. For 3D models, it is an elegant way to deal with <sup>97</sup> cuttings, viscous effects and very thin features [26, 27, 28]. A <sup>98</sup> mesh-less, octree-based adaptive extension of shape matching <sup>99</sup> has been proposed [29]. Besides all these methods based on <sup>100</sup> multiple resolutions, Kim and James [30] take a more alge-<sup>101</sup> braic approach, where the displacement field is decomposed <sup>102</sup> on a small, dynamically updated, basis of orthogonal vectors, <sup>103</sup> while a small set of carefully chosen integration points are used <sup>104</sup> to compute the forces. In constrast to these works, our method <sup>105</sup> relies on velocity field analysis and a meshless discretization.

Numerous error estimators for refinement have been pro-107 posed in conventional FEM analysis. For static analysis, they 108 are generally based on a precomputed stress field. This is not 109 feasible in real-time applications, where the current configura-110 tion and corresponding stress must be used. Wu et al. [17] pro-111 posed four criteria based on the curvature of the stress, strain or 112 displacement fields. Debunne et al. [18] considered the Lapla-113 cian of the displacement. Lenoir et al. [22] refined parts in 114 contact for wire simulation. These approaches refine the ob-115 jects where they are the most deformed, and they are not able to 116 save computation time in equilibrium states different from the 117 rest state. The problems relative to the criterion thresholds are 118 rarely discussed, even though potential popping artifacts can be <sup>119</sup> problematic: the smaller the thresholds, the smaller the popping 120 artifacts, but also the more difficult to simplify and thus the less efficient. 121

While our adaptive scheme is primarily targeted at physicallybased animation, it can also be interesting to improve the robustness of deformable mesh registration schemes. Finding correspondences between a source (template) mesh and a target mesh or point cloud is a fundamental task in shape acquitize global correspondence search is generally regularized using a global correspondence search is generally regularized using a deformation method, that constrains the displacement of the 131 point (ICP) algorithm [34] is the most common procedure to 132 align a source mesh to a target mesh. At each iteration, source 133 point correspondences are locally found by an optimized clos-134 est point search [35]. In the original ICP algorithm, the best 135 global linear transformation is found by minimizing distances 136 between source points and their corresponding points. Instead, <sup>137</sup> elastic ICP [36, 37] can be easily performed by treating distance 138 gradients as external forces (*i.e.* springs) applied to a given deformable model. Deformable registration can be more accurate 140 but is however less robust, because a higher number of DOFs 141 makes it more sensitive to local extrema. In contrast, coarse-to-142 fine registration strategies improve robustness, computational 143 speed, and accuracy. In this paper, we propose to use our adap-144 tive scheme to automatically tune the number of DOFs required 145 during the registration process, and show robustness improve-146 ment.

#### 147 2. Frame-based Simulation Method

In this section we summarize the method that our contribu-149 tion extends, and we introduce notations and basic equations. 150 The method of [7] performs the physical simulation of vis-151 coelastic solids using a hyperelastic formulation. The control nodes are moving frames with 12 Degrees of Freedom (DOF) whose positions, velocities and forces in world coordinates are 153 stored in state vectors  $\mathbf{x}$ ,  $\mathbf{v}$  and  $\mathbf{f}$ . A node configuration is de-154 155 fined by an affine transformation, represented in homogeneous coordinates by the  $4 \times 4$  matrix  $\mathbf{X}_i$ . The world coordinates  $\mathbf{x}_i$  of <sup>157</sup> node *i* are simply the entries in  $X_i$  corresponding to affine trans-158 formations. Relative coordinates are obtained similarly from <sup>159</sup> relative transformations:  $\mathbf{X}_{i}^{j} = \mathbf{X}_{j}^{-1}\mathbf{X}_{i}$ . A collection of nodes 160 generates a deformation field using a Skeleton Subspace Defor-161 mation (SSD) method, also called *skinning* [38]. We use Linear 162 Blend Skinning (LBS), though other methods could be suitable 163 (see e.g. [39] for a discussion about SSD techniques). The po-164 sition of a material point i is defined using a weighted sum of 165 affine displacements:

$$\mathbf{p}_i(t) = \sum_{j \in \mathcal{N}} \phi_i^j \mathbf{X}_j(t) \mathbf{X}_j(0)^{-1} \mathbf{p}_i(0)$$
(1)

166 <sup>167</sup> of the shape function of node *j* at material position  $\mathbf{p}_i(0)$ , com-<sup>168</sup> puted at initialization time using distance ratios as in [7]. Spa-169 tially varying shape functions allow more complex deforma-170 tions. Similar to nodes, the state of all skinned points are stored <sup>171</sup> as vectors:  $\mathbf{p}$ ,  $\dot{\mathbf{p}}$ , and  $\mathbf{f}_{\mathbf{p}}$ . Eq. (1) is linear in node coordinates, <sup>172</sup> therefore a *constant* Jacobian matrix  $J_p$  can be assembled at 173 initialization, relating node coordinates to skinned point coor-174 dinates:

$$\mathbf{p} = \mathbf{J}_p \mathbf{x}, \quad \dot{\mathbf{p}} = \mathbf{J}_p \mathbf{v} \tag{2}$$

175 176 the contact surface of the object. The Principle of Virtual Work 224 be rewritten in terms of active nodes only: 177 implies that nodal forces  $\mathbf{f}$  are obtained from skin forces  $\mathbf{f}_{\mathbf{p}}$  as

<sup>130</sup> template to a set of feasible transformations. Iterative closest <sup>178</sup>  $\mathbf{f} = \mathbf{J}_p^T \mathbf{f}_p$ . Similarly, the generalized mass matrix for nodes M  $_{179}$  can be obtained at initialization based on the scalar masses  $M_p$ <sup>180</sup> of skinned particles:  $\mathbf{M} = \mathbf{J}_p^T \mathbf{M}_p \mathbf{J}_p$ . As shown in [6], differ-181 entiating Eq. (1) with respect to material coordinates produces 182 deformation gradients in the current configuration. By map-183 ping deformation gradients to strains (such as Cauchy, Green-184 Lagrange or corotational), and applying a constitutive law (such 185 as Hooke or Mooney-Rivlin), we can compute the elastic poten-186 tial energy density at any location. After spatial integration and 187 differentiation with respect to the degrees of freedom, forces 188 can be computed and propagated back to the nodes.

> We use different discretizations for visual surfaces, con-189 <sup>190</sup> tact surfaces, mass and elasticity (potential energy integration <sup>191</sup> points). Masses are precomputed using a dense volumetric ras-192 terization, where voxels are seen as point masses. Deforma-193 tion gradient samples (i.e. Gauss points) are distributed so as to <sup>194</sup> minimize the numerical integration error (see Sec. 5). For each 195 sample, volume moments are precomputed from the fine voxel <sup>196</sup> grid and associated with local material properties.

> The method is agnostic with respect to the way we solve <sup>198</sup> the equations of motion. We apply an implicit time integra-199 tion to maintain stability in case of high stiffness or large time 200 steps [40]. At each time step, we solve a linear equation system

$$\mathbf{A}\Delta\mathbf{v} = \mathbf{b} \tag{3}$$

where  $\Delta \mathbf{v}$  is the velocity change during the time step, matrix 202 A is a weighted sum of the mass and stiffness matrices, while <sup>203</sup> the right-hand term depends on the forces and velocities at the <sup>204</sup> beginning of the time step. The main part of the computation 205 time to set up the equation system is proportional to the num-206 ber integration points, while the time necessary to solve it is a 207 polynomial function of the number of nodes (note that A is a <sup>208</sup> sparse, positive-definite symmetric matrix).

#### 209 3. Adaptive Frame-based Simulation

Our first extension to the method presented in Sec. 2 is to 211 attach control nodes to others to reduce the number of inde-212 pendent DOFs. This amounts to adding an extra block to the 213 kinematic structure of the model, as shown in Fig. 3.

The independent state vectors are restricted to the active 214 where  $\mathcal{N}$  is the set of all control nodes, and  $\phi_i^j$  is the value 215 nodes. At each time step, the dynamics equation is solved to <sup>216</sup> update the positions and velocities of the active nodes, then the 217 changes are propagated to the passive nodes, then to the skin <sup>218</sup> points and the material integration points. The forces are prop- $_{219}$  agated the other way round. When a node *i* is passive, its matrix <sup>220</sup> is computed from active nodes using LBS as:

$$\mathbf{X}_{i}(t) = \sum_{j \in \mathcal{A}} \phi_{i}^{j} \mathbf{X}_{j}(t) \mathbf{X}_{i}^{j}(0)$$
(4)

where  $\mathcal{A}$  is the set of active nodes and  $\phi_i^j$  is the value of the 221 <sup>222</sup> shape function of node *j* at the origin of  $X_i$  in the reference, External forces can be applied directly to the nodes, or to 223 undeformed configuration. The point positions of Eq. (1) can



Figure 3: Kinematic structure of the simulation. Our adaptive scheme splits the control nodes into active (*i.e.* independent) nodes and passive (*i.e.* mapped) nodes.

$$\mathbf{p}_{i}(t) = \sum_{j \in \mathcal{A}} \psi_{i}^{j} \mathbf{X}_{j}(t) \mathbf{X}_{j}(0)^{-1} \mathbf{p}_{i}(0)$$

$$\psi_{i}^{j} = \phi_{i}^{j} + \sum_{k \in \mathcal{P}} \phi_{k}^{j} \phi_{i}^{k}$$
(6)

where  $\mathcal{P}$  is the set of passive nodes. These equations gen-225 226 eralize similarly to deformation gradients, to obtain a Jacobian 227 matrix  $\mathbf{J}_{p}$  in terms of active nodes alone. This easy composition of LBS is exploited in our node hierarchy (Sec. 4.2.2) 228 and our adaptive spatial integration scheme (Sec. 5). At any 229 time, an active node *i* can become passive. Since the coeffi-231 cients used in Eq. (4) are computed in the undeformed configu-<sup>232</sup> ration, the position  $\bar{\mathbf{X}}_i$  computed using this equation is different <sup>233</sup> from the current position  $X_i$ , and moving the frame to this po-234 sition would generate an artificial instantaneous displacement. 235 To avoid this, we compute the offset between the two configura-<sup>236</sup> tions  $\delta \mathbf{X}_i = \bar{\mathbf{X}}_i^{-1} \mathbf{X}_i$ , as illustrated in Fig. 2. The skinning of the 237 frame is then biased by this offset as long as the frame remains <sup>238</sup> passive, and its velocity is computed using the corresponding 239 Jacobian matrix:

$$\mathbf{X}_{i}(t) = \sum_{j \in \mathcal{A}} \psi_{i}^{j} \mathbf{X}_{j}(t) \mathbf{X}_{j}(0)^{-1} \mathbf{X}_{i}(0) \delta \mathbf{X}_{i}$$
(7)  
$$\mathbf{u}_{i}(t) = \mathbf{J}_{i} \mathbf{v}(t)$$
(8)

Our adaptivity criterion is based on comparing the velocity <sup>241</sup> of a passive node attached to nodes of A, with the velocity of <sup>242</sup> the same node moving independently; if the difference is below <sup>243</sup> a threshold the node should be passive, otherwise it should be <sup>244</sup> active.

#### 245 One-dimensional Example

A simple one-dimensional example is illustrated in Fig. 4. A bar is discretized using three control nodes and two integration points, and stretched horizontally by its weight, which applies the external forces 1/4, 1/2 and 1/4, from left to right



Figure 4: Refinement and simplification. Red and green arrows denote external and internal forces, respectively. Plain circles represent active nodes, while empty circles represent passive nodes attached to their parents, and crosses represent the positions of passive nodes interpolated from their parents positions. Dashed lines are used to denote forces divided up among the parent nodes. Rectangles denote integration points, where the stresses  $\sigma$  are computed. (a): A bar in reference state undergoes external forces and starts stretching. (b): In rest state, 3 active nodes. (c): With the middle node attached with an offset with respect to the interpolated position. (d): After replacing two integration points with one.

<sup>250</sup> respectively. For simplicity we assume unitary gravity, stiffness <sup>251</sup> and bar section, so that net forces are computed by simply sum-<sup>252</sup> ming up strain and force magnitudes. At the beginning of the <sup>253</sup> simulation, Fig. 4a, the bar is in reference configuration with <sup>254</sup> null stress, and the middle node is attached to the end nodes, <sup>255</sup> interpolated between the two. The left node is fixed, the ac-<sup>256</sup> celeration of the right node is 1, and the acceleration of the <sup>257</sup> interpolated node is thus 1/2. However, the acceleration of the <sup>259</sup> it is subject to gravity only. Due to this difference, we activate it <sup>260</sup> and the bar eventually converges to the equilibrium configura-<sup>261</sup> tion shown in Fig. 4b, with a non-uniform extension, as can be <sup>262</sup> visualized using the vertical lines regularly spaced in the mate-<sup>263</sup> rial domain.

<sup>264</sup> Once the center node is stable with respect to its parents, <sup>265</sup> we can simplify the model by attaching it to them, with offset <sup>266</sup>  $\delta \mathbf{X}$ . External and internal forces applied to the passive node, <sup>267</sup> which balance each other, are divided up among its parents, <sup>268</sup> which do not change the net force applied to the end node. The <sup>269</sup> equilibrium is thus maintained. The computation time is faster <sup>270</sup> since there are less unknown in the dynamics equation. How-<sup>271</sup> ever, computing the right-hand term remains expensive since <sup>272</sup> the same two integration points are used.

<sup>273</sup> Once the displacement field is simplified, any change of <sup>274</sup> strain due to the displacement of the two independent nodes <sup>275</sup> is uniform across the bar. We thus merge the two integration <sup>276</sup> points to save computation time, as shown in Fig. 4d. Sec-<sup>277</sup> tion 4 details node adaptivity, while the adaptivity of integration <sup>278</sup> points is presented in Section 5.

#### 279 4. Adaptive Kinematics

At each time step, our method partitions the nodes into two At each time step, our method partitions the nodes into two sets: the *active* nodes, denoted by A, are the currently independent DOFs from which the *passive* nodes, denoted by P, are mapped from the active nodes. We further define a subset  $AC \subset P$  to be composed of nodes candidate for activation. 285 Likewise, the deactivation candidate set is a subset  $\mathcal{PC} \subset \mathcal{A}$ . 333 active cases can be seen as a velocity correction due to kine-<sup>286</sup> To decide whether candidate nodes should become passive or <sup>334</sup> matic constraint forces:  $\mathbf{d} = dt \mathbf{M}^{-1} \mathbf{f}_{d}$  for some force vector  $\mathbf{f}_{d}$ , 287 active, we compare their velocities in each state (passive and 335 where the corresponding kinematic constraint maintains some 288 active) and change their status if the velocity difference crosses 336 frames dependent on others. Therefore, the constraint is holo- $_{209}$  a certain user-defined threshold  $\eta$  discussed below. At each time  $_{337}$  nomic and the associated constraint forces  $f_{\lambda}$  produce no instan-<sup>290</sup> step, we compare the velocities in the three following cases:

291	1.	with $\mathcal{A} \setminus \mathcal{PC}$ active and $\mathcal{P} \cup \mathcal{PC}$ passive (coarser resolu-
292		tion)

- 2. with  $\mathcal{A}$  active and  $\mathcal{P}$  passive (current resolution) 293
- 3. with  $\mathcal{A} \cup \mathcal{AC}$  active and  $\mathcal{P} \setminus \mathcal{AC}$  passive (finer resolution) 294
- 295

We avoid solving the three implicit integrations, noticing 296 297 that cases 1 and 3 are only used to compute the adaptivity cri-<sup>298</sup> terion. Instead of performing the implicit integration for case we use the solution given by 2 and we compute the veloci-299  $_{300}$  ties of the frames in  $\mathcal{PC}$  as if they were passive, using Eq. (8). 301 For case 3, we simply use an explicit integration for the addi- $_{302}$  tional nodes  $\mathcal{AC}$ , in linear time using a lumped mass matrix. In <sup>303</sup> practice, we only noticed small differences with a fully implicit <sup>304</sup> integration. At worse, overshooting due to explicit integration 305 temporary activates too many nodes.

Once every velocity difference has been computed and mea-306 <sup>307</sup> sured for candidate nodes, we integrate the dynamics forward 308 at current resolution (*i.e.* using system 2), then we update the <sup>309</sup> sets  $\mathcal{A}, \mathcal{P}, \mathcal{PC}, \mathcal{AC}$  and finally move on to the next time step.

#### 310 4.1. Velocity Metrics

For a candidate node *i*, the difference between its passive 311 <sup>312</sup> and active velocities is defined as:

$$\mathbf{d}_i = \mathbf{J}_i(\mathbf{v} + \Delta \mathbf{v}) - (\mathbf{u}_i + \Delta \mathbf{u}_i)$$
(9)

313 314 velocity updates computed by time integration, respectively in 357 era will produce lower measures, favoring deactivation. More  $_{315}$  the case where the candidate node is passive and active. Note  $_{358}$  precisely, if we call G<sub>i</sub> the kinematic mapping between node i 316 that for the activation criterion computed using explicit integra- 359 and the mesh vertices, obtained by considering mesh vertices as 317 tion (case 3), this reduces to the generalized velocity difference 360 material points in Eq. (1) and Eq. (2), and Z a diagonal matrix  $\mathbf{d}_i = \mathbf{J}_i \Delta \mathbf{v} - dt \mathbf{\tilde{M}}_i^{-1} \mathbf{f}_i$  where  $\mathbf{\tilde{M}}_i$  is the lumped mass matrix block 301 with positive values decreasing along with the distance between  $_{319}$  of node *i*,  $\mathbf{f}_i$  its net external force and *dt* is the time step, which  $_{362}$  mesh vertices and the camera, the criterion metric is then given <sub>320</sub> is a difference in *acceleration* up to dt. A measure of **d**<sub>i</sub> is then <sub>363</sub> by: 321 computed as:

$$\mu_i = \|\mathbf{d}_i\|_{\mathbf{W}_i}^2 := \frac{1}{2}\mathbf{d}_i^T \mathbf{W}_i \mathbf{d}_i$$
(10)

where  $\mathbf{W}_i$  is a positive-definite symmetric matrix defining 322  $_{323}$  the metric (some specific  $\mathbf{W}_i$  are shown below). The deac- $_{324}$  tivation (respectively activation) of a candidate node *i* occurs <sup>325</sup> whenever  $\mu_i \leq \eta$  (respectively  $\mu_i > \eta$ ), where  $\eta$  is a positive 326 user-defined threshold.

#### 327 4.1.1. Kinetic Energy

As the nodes are transitioning between passive and active 328 329 states, a velocity discontinuity may occur. In order to prevent 330 instabilities, a natural approach is to bound the associated ki-<sup>331</sup> netic energy discontinuity, as we now describe. The difference  $_{332}$  **d** = **J**(**v** +  $\Delta$ **v**) – (**u** +  $\Delta$ **u**) between velocities in the passive and

338 taneous mechanical work:

$$\mathbf{f}_{\lambda}^{T} \mathbf{J} (\mathbf{v} + \Delta \mathbf{v}) = 0 \tag{11}$$

This means that  $\mathbf{J}(\mathbf{v} + \Delta \mathbf{v})$  and **d** are **M**-orthogonal. It fol-339 340 lows that the kinetic energy difference between the active and 341 passive states is simply:

$$\|\mathbf{u} + \Delta \mathbf{u}\|_{\mathbf{M}}^2 - \|\mathbf{J}(\mathbf{v} + \Delta \mathbf{v})\|_{\mathbf{M}}^2 = \|\mathbf{d}\|_{\mathbf{M}}^2$$
(12)

The triangle inequality gives an upper bound on the total 342 343 change:

$$\|\mathbf{d}\|_{\mathbf{M}} = \left\|\sum_{i=1}^{k} \hat{\mathbf{d}}_{i}\right\|_{\mathbf{M}} \le \sum_{i=1}^{k} \left\|\hat{\mathbf{d}}_{i}\right\|_{\mathbf{M}} = \sum_{i=1}^{k} \|\mathbf{d}_{i}\|_{\mathbf{M}_{i}}$$
(13)

where  $\hat{\mathbf{d}}_i = (0, \dots, \mathbf{d}_i^T, 0, \dots)^T$  is a column vector whose <sup>345</sup> only non-zero entries are the ones corresponding to DoF *i*. If <sup>346</sup> we use  $\mathbf{W}_i = \mathbf{M}_i$  in Eq. (10), we effectively bound each  $\|\mathbf{d}_i\|_{\mathbf{M}_i}$  $_{347}$  in Eq. (13), hence the left-hand side  $\|\mathbf{d}\|_{\mathbf{M}}$ , and finally the to-<sup>348</sup> tal kinetic energy difference. The criterion threshold  $\eta$  can be 349 adapted so that the upper bound in Eq. (13) becomes a small 350 fraction of the current kinetic energy.

#### 351 4.1.2. Distance to Camera

For computer graphic applications, one is usually ready to 353 sacrifice precision for speed as long as the approximation is <sup>354</sup> not visible to the user. To this end, we can measure velocity 355 differences according to the distance to the camera of the assowhere  $J_i$  is the Jacobian of Eq. (8), and  $\Delta v, \Delta u_i$  are the 356 ciated visual mesh, so that motion happening far from the cam-

$$\mathbf{W}_i = \mathbf{G}_i^T \mathbf{Z} \mathbf{G}_i \tag{14}$$

In practice, we use a decreasing exponential for Z values 365 (1 on the camera near-plane, 0 on the camera far-plane), mim-366 icking the decreasing precision found in the depth buffer dur-<sup>367</sup> ing rendering. In our experiments, the exponential decrease 368 resulted in coarser models compared to a linear decrease as 369 the camera distance increased, without noticeable visual qual-370 ity degradation. The two metrics can also be combined by re-371 taining the minimum of their values: simplification is then fa-<sup>372</sup> vored far from the camera, where the distance metric is always 373 small, while the kinetic energy metric is used close to the cam-374 era, where the distance metric is always large. Of course, other 375 metrics may be used as well: for instance one may want to pe-376 nalize distance to a given region of interest.



Figure 5: Reference node hierarchy. From left to right: the first three levels, and the dependency graph.



Figure 6: Mechanical hierarchy. Left: active nodes 0, 1, 4, 5, 6 at a given time. Middle: Reference hierarchy; nodes 2, 3, 7 are activation candidates; nodes 4,5,6 are deactivation candidates. Right: the resulting two-level contracted graph to be used in the mechanical simulation.

#### 377 4.2. Adaptive Hierarchy

In principle, we could start with an unstructured fine node 378 379 discretization of the objects and at each time step, find the best <sup>380</sup> simplifications by considering all possible deactivation and ac-<sup>381</sup> tivation candidates. However, in order to avoid a quadratic <sup>382</sup> number of tests, we pre-compute a node hierarchy and define 383 candidate nodes to be the ones at the interface between passive 384 and active nodes in the hierarchy.

#### 385 4.2.1. Hierarchy Setup

386 <sub>387</sub> trated in Fig. 5. When building each level of the hierarchy,  $_{436}$  polynomials of degree *n* in the coordinates of **p**, and  $\bar{\mathbf{g}}_{e}$  is a vec-<sup>390</sup> the nodes already created at coarser levels. Shape functions are <sup>439</sup> ily fine rasterization. The approximation of Eq. (15) is exact if <sup>391</sup> computed for each level based on the position of inserted nodes  $_{440}$  n is the polynomial degree of g. Due to a possibly large num- $_{392}$  and stored in the fine voxel grid as in [7]. Given a node *j* at  $_{441}$  ber of polynomial factors, we limit our approximation to quartic <sup>393</sup> a given level *l*, weights  $\phi_i^i$  relative to its parents *i* are obtained <sup>442</sup> functions with respect to material coordinates, corresponding to  $_{394}$  through interpolation in the grid at level l-1. For each non-zero  $_{443}$  strain energies and forces when shape functions are linear and <sup>395</sup> weight, an edge is inserted into the dependency graph, resulting <sup>444</sup> the strain measure quadratic (*i.e.* Green-Lagrangian strain). <sup>396</sup> in a generalized hierarchy based on a Directed Acyclic Graph.

#### 397 4.2.2. Hierarchy Update

The candidates for activation  $\mathcal{AC}$  are the passive nodes with  $_{448}$  Fig. 7b. 398 <sup>399</sup> all parents active. Conversely, the candidates for deactivation  $\mathcal{PC}$  are the active nodes with all children passive, if any, except 400 401 for the root of the reference hierarchy. In particular, active leaf  $_{402}$  nodes are always in  $\mathcal{PC}$ . In the example shown in Fig. 6, nodes  $_{403}$  4, 5, 6, 1, and 0 shown in the character outline are active. As <sup>404</sup> such, they do not mechanically depend on their parents in the <sup>405</sup> reference hierarchy, and the mechanical dependency graph is 406 obtained by removing the corresponding edges from the refer-407 ence hierarchy. For edges in this two-levels graph, weights are <sup>408</sup> obtained by contracting the reference hierarchy using Eq. (5).

409 Similarly, two-level graphs can be obtained for the cases 1 and 410 3 discussed in the beginning of this section.

### 411 5. Adaptive Spatial Integration

We now describe how to adapt the spatial integration of 412 <sup>413</sup> elastic energy in order to further increase computational gains.

#### 414 5.1. Discretization

The spatial integration of energy and forces is numerically 416 computed using Gaussian quadrature, a weighted sum of values <sup>417</sup> computed at integration points. Exact quadrature rules are only 418 available for polyhedral domains with polynomial shape func-419 tions (e.g. tri-linear hexahedra). In meshless simulation, such 420 rules do not exist in general. However, in linear blend skinning 421 one can easily show that the deformation gradient is uniform 422 (respectively linear) in regions where the shape functions are 423 constant (respectively linear). As studied in [7], uniform shape 424 functions can be only obtained with one node, so linear shape 425 functions between nodes are the best choice for homogeneous 426 parts of the material, since the interpolation then corresponds 427 to the solution of static equilibrium. One integration point of 428 a certain degree (*i.e.* one elaston [4]) is sufficient to exactly in-429 tegrate polynomial functions of the deformation gradient there, 430 such as deformation energy in linear tetrahedra. We leverage 431 this property to optimize our distribution of integration points. <sup>432</sup> In a region  $\mathcal{V}e$  centered on point  $\bar{\mathbf{p}}_e$ , the integral of a function g 433 is given by:

$$\int_{\bar{\mathbf{p}}\in\mathcal{V}e}g\approx\mathbf{g}^{T}\int_{\bar{\mathbf{p}}\in\mathcal{V}e}(\bar{\mathbf{p}}-\bar{\mathbf{p}}_{e})^{(n)}=\mathbf{g}^{T}\bar{\mathbf{g}}_{e}$$
(15)

where  $\mathbf{g}$  is a vector containing g and its spatial derivatives 434 Our hierarchy is computed at initialization time, as illus- 435 up to degree n evaluated at  $\bar{\mathbf{p}}_{e}$ , while  $\mathbf{p}^{(n)}$  denotes a vector of we perform a Lloyd relaxation on a fine voxel grid to spread 437 tor of polynomials integrated across Ve which can be computed new control nodes as evenly as possible, taking into account 438 at initialization time by looping over the voxels of an arbitrar-

> Since the integration error is related to the linearity of shape 445 446 functions, we decompose the objects into regions of as linear as 447 possible shape functions at initial time, as shown in Fig. 7a and

> We compute the regions influenced by the same set of in-449 <sup>450</sup> dependent nodes, and we recursively split these regions until 451 a given linearity threshold is reached, based on the error of a <sup>452</sup> least squares linear fit of the shape functions. Let  $\phi_i(\bar{\mathbf{p}})$  be the <sup>453</sup> shape function of node *i* as defined in Eq. (1), and  $\mathbf{c}_i^{e^T} \bar{\mathbf{p}}^{(1)}$  its  $_{454}$  first order polynomial approximation in  $\mathcal{V}e$ . The linearity error 455 is given by:



Figure 7: Adaptive integration points in 1D. Disks denote control nodes while rectangles denote integration points.

$$\varepsilon(\mathbf{c}) = \int_{\mathcal{V}e} (\phi_i(\bar{\mathbf{p}}) - \mathbf{c}^T \bar{\mathbf{p}}^{(1)})^2$$
(16)

$$= \mathbf{c}^{T} A^{e} \mathbf{c} - 2 \mathbf{c}^{T} B_{i}^{e} + C_{i}^{e}$$
(17)

with: 
$$A^e = \int_{\mathcal{V}e} \bar{\mathbf{p}}^{(1)} \bar{\mathbf{p}}^{(1)T}$$
 (18)

$$B_i^e = \int_{\mathcal{V}e} \phi_i(\bar{\mathbf{p}})\bar{\mathbf{p}}^{(1)}$$
(19)

$$C_i^e = \int_{\mathcal{V}e} \phi_i(\bar{\mathbf{p}})^2 \tag{20}$$

We solve for the least squares coefficients  $\mathbf{c}_{i}^{e}$  minimizing  $\varepsilon$ : 457 C; =  $(A^e)^{-1}B_i^e$ . The region with largest error is split in two 458 until the target number of integration points or an upper bound 506 6.1. Validation 459 on the error is reached.

#### 460 5.2. Merging Integration Points

At run-time, the shape functions of the passive nodes can 46 462 be expressed as linear combinations of the shape functions of <sup>463</sup> the active nodes using Eq. (6). This allows us to merge integra-<sup>464</sup> tion points sharing the same set of active nodes (in  $A \cup AC$ ), as 465 shown in Fig. 7c. One can show that the linearity error in the <sup>466</sup> union of regions e and f is given by:

$$\varepsilon = \sum_{i} (C_{i}^{e} + C_{i}^{f}) - \sum_{i} (B_{i}^{e} + B_{i}^{f})^{T} (A^{e} + A^{f})^{-1} \sum_{i} (B_{i}^{e} + B_{i}^{f})$$

If this error is below a certain threshold, we can merge the 467 integration points. The new values of the shape function (at ori-469 gin) and its derivatives are:  $\mathbf{c}_i^n = (A^e + A^f)^{-1}(B_i^e + B_i^f)$ . For nu-<sup>470</sup> merical precision, the integration of Eq. (15) is centered on  $\bar{\mathbf{p}}_e$ . When merging e and f, we displace the precomputed integrals  $_{472}$   $\mathbf{\bar{g}}_{e}$  and  $\mathbf{\bar{g}}_{f}$  to a central position  $\mathbf{\bar{p}}_{n} = (\mathbf{\bar{p}}_{e} + \mathbf{\bar{p}}_{f})/2$  using simple 473 closed form polynomial expansions. Merging is fast because 474 the volume integrals of the new integration points are directly 475 computed based on those of the old ones, without integration 527 <sup>476</sup> across the voxels of the object volume. Splitting occurs when <sup>528</sup> following example scenes:

477 the children are not influenced by the same set of independent 478 nodes, due to a release of passive nodes. To speed up the adap-479 tivity process, we store the merging history in a graph, and dy-480 namically update the graph (instead of restarting from the finest 481 resolution). Only the leaves of the graph are considered in the 482 dynamics equation.

When curvature creates different local orientations at the in-483 484 tegration points, or when material laws are nonlinear, there may 485 be a small difference between the net forces computed using the 486 fine or the coarse integration points. Also, since Eq. (5) only <sub>487</sub> applies when rest states are considered, position offsets  $\delta X$  on 488 passive nodes create forces that are not taken into account by 489 coarse integration points. To maintain the force consistency be-<sup>490</sup> tween the different levels of details, we compute the difference <sup>491</sup> between the net forces applied by the coarse integration points 492 and the ones before adaptation. This force offset is associated <sup>493</sup> with the integration point and it is added to the elastic force it 494 applies to the nodes. Since net internal forces over the whole 495 object are necessarily null, so is the difference of the net forces 496 computed using different integration points, thus this force off-497 set influences the shape of the object but not its global trajec-498 tory. In three dimension, to maintain the force offset consistent 499 with object rotations, we project it from the basis of the defor-<sup>500</sup> mation gradient at the integration point to world coordinates.

#### 501 6. Results

We now report experimental results obtained with our method, 502 503 demonstrating its interest for computer graphics. We also pro-<sup>504</sup> pose an application of our technique to the deformable mesh 505 registration problem.

To measure the accuracy of our method, we performed some 507 508 standard tests on homogeneous Hookean beams under exten-509 sion and flexion (see Fig. 8). We obtain the same static equi-510 librium solutions using standard tetrahedral finite elements and 511 frame-based models (with/without kinematics/integration point 512 adaptation). In extension, when inertial forces are negligible 513 (low masses or static solving or high damping), our adaptive <sup>514</sup> model is fully coarsened as expected from the analytic solution 515 (one frame and one integration point are sufficient). In bend-516 ing, adaptivity is necessary to model non-linear variations of 517 the deformation gradient. At equilibrium, our model is simpli-<sup>518</sup> fied as expected. Fig. 9 shows the variation of the kinetic energy 519 (red curves). As expected, energy discontinuities remain lower 520 than the criterion threshold when adapting nodes and integra-<sup>521</sup> tion points (green and blue curves), allowing the user to control 522 maximum jumps in velocity. Because there is also no position 523 discontinuities (no popping) as guaranteed by construction, the 524 adaptive simulation in visually very close to the non-adaptive 525 one.

#### 526 6.2. Complex Scenes

We demonstrate the genericity of our method through the



Figure 8: Four cantilever beams at equilibrium with the same properties and loading (fixed on one side, and subject to gravity). Right: perspective side view of the same configuration, showing that our adaptivity framework produces results similar (up to the depth-buffer precision, hence the color changes) to the non-adaptive frame-based approach.



Figure 9: Kinetic Energy (red) Analysis with varying number of frames (green) and integration points (blue) over time (cantilever beam under flexion).

#### 529 6.2.1. Christmas Tree

A Christmas tree (Fig. 1) with a stiff trunk and more flexible branches, with rigid ornaments is subject to gravity. Initially, only one node is used to represent the tree. As the ornament falls, the branches bend and nodes are automatically active until the static equilibrium is reached and the nodes become passive again. The final, bent configuration is again represented using only one control node.

#### 537 6.2.2. Elephant Seal

A simple animation skeleton is converted to control nodes to
 animate an elephant seal (Fig. 10) using key-frames. Adaptive,
 secondary motions are automatically handled by our method as more nodes are added into the hierarchy.



Figure 10: 40 adaptive, elastic frames (*green: active, red: passive*) adding secondary motion on a (deliberately short) kinematic skeleton corresponding to 12 (blue) frames.

#### 542 6.2.3. Bouncing Ball

A ball is bouncing on the floor with unilateral contacts (Fig. 11). As the ball falls, only one node is needed to animate it. On impact, contact constraint forces produce deformations and the nodes are active accordingly. On its way up, the ball recovers that our method allows simplifications in non-equilibrium states.



Figure 11: A falling deformable ball with unilateral contacts.

#### 550 6.2.4. Elastic Mushroom Field

In Fig. 12a, simplification allows all the mushrooms to be tatached to one single control frame until a shoe crushes some tatached to one single control frame until a shoe crushes some to shoe them. Local nodes are then activated to respond to shoe to shoe to secondary contacts. They are deactivated as the shoe goes away. In this example of multi-body adaptivity, the to node configuration has little importance and simply corretron node configuration has little importance and simply correto a global affine transformation of all the mechanical bodies. The second level of the hierarchy consists in one node the per body, and the remaining levels are restricted to each object the (*i.e.* no cross-object node influence, though our method allows this).



(a) Crushing elastic mushrooms

(b) 18 Armadillos falling in a bowl

Figure 12: Selected pictures of complex scenes where only a subset of the available frames and integration points are active.

#### 562 6.2.5. Deformable Ball Stack

Eight deformable balls (Fig. 13) are dropped into a glass. From left to right: (a) A unique node is necessary to simulate all balls falling under gravity, at the same speed. (b) While colliding, nodes are activated to simulate deformations. (c) Once for stabilized, the deformed balls are simplified to one node. (d) Removing the glass, some nodes are re-activated to allow the balls to fall apart. (e) Once the balls are separated they are <sup>570</sup> freely falling with air damping, and one node is sufficient to <sup>571</sup> simulate all of them.



Figure 13: Eight deformable balls stacking up in a glass, which is eventually removed.

#### 572 6.2.6. Armadillo Salad

A set of Armadillos (Fig. 12b) is dropped into a bowl, demon-574 strating the scalability and robustness of our method in a diffi-575 cult (self-)contacting situation.

#### 576 6.3. Performance

In the various scenarios described above, our technique alrote lows a significant reduction of both kinematic DOFs and integration points, as presented in Table 2. Speedups are subset stantial, even when collision handling is time consuming. It is worth noting that, for a fair comparison with the non-adaptive set case, our examples exhibit large, global and dynamical deformations.

In order to evaluate the gain of adaptivity regarding the 584 585 scene complexity, we throw armadillos in a bowl, at various resolutions. The speedups presented in Table 1 show that scenes 587 resulting in larger systems give better speedups since the complexity of solving the system increases along with the number 588 589 of DOFs. The algorithmic complexity of solving deformable object dynamics generally depends on three factors: the num-590 ber of DOFs, the computation of elastic forces and, in the case of iterative solvers, the conditioning of the system. By using 592 fewer integration points, our method is able to compute elastic 593 forces in a much faster way. In the case of badly conditioned 594 595 systems, as for instance tightly mechanically coupled system 596 (e.g. stacks), iterative methods need a large number of itera-597 tions and thus the number of DOFs becomes critical. The de-<sup>598</sup> pendency on the number of DOFs is even larger when using di-<sup>599</sup> rect solvers. Thus, our method is particularly interesting in such 600 cases and allow for significant speedups compared to the non- $_{601}$  adaptive case. For instance:  $6.25 \times$  when the balls are stacked 602 into the glass (see Fig. 13c).

We noticed that the overhead due to adaptivity is moderate compared to the overall computational time (typically between 5% and 10%), since adaptivity is incremental for both nodes and integration points between two consecutive time steps. The der dense voxel grid is visited only once at initialization to compute shape functions, masses, and integration data. Note that the cost of our adaptivity scheme is independent from the method to compute shape functions (they could be based on harmonic cordinates, natural neighbor interpolants, etc).

Nb	Max Nodes / Integration Points per Armadillo							
Armadillos								
	10 / 49	100 / 1509	250 / 3953					
1	×1.75	×3.3	×12					
18	×1.5	×3	×3.1					

Table 1: Speedups for a salad of one and 18 armadillos at various maximal resolutions (including collision timing)

#### 612 6.4. Application to Mesh Registration

613 Using our adaptive scheme, the number of DOFs can be 614 progressively increased as needed during the registration pro-615 cess, as shown in Fig. 14. Our approach is the following: we 616 start with a fully coarsened model and apply the deformable 617 registration procedure. Springs between the deformable sur-618 face and corresponding points on the target surface are created 619 at each iteration. For a given spring stiffness, nodes become 620 active as needed to produce the required, increasingly local de-621 formations, minimizing both external and internal energy. Af-622 ter equilibrium is reached, our adaptive kinematic model comes 623 back to its fully coarsened state. When this happens, we pro-624 gressively increase the registration spring stiffness, then again 625 let the adaptive model deform accordingly. We iterate this pro-626 cess until the distance to the target surface becomes smaller 627 than a user-defined threshold. By using our adaptive kinemat-628 ics, the deformation field seamlessly transitions from global to 629 local transformations as needed, implementing the coarse-to-630 fine registration strategy. In addition, parts of the mesh that 631 are already registered will not be refined again as other parts 632 continue to undergo deformations, if these parts belong to sep-633 arate branches of the node hierarchy. Consider for instance 634 the node hierarchy presented in Fig. 5: if the legs are already 635 registered (nodes 3, 4, 7, 8, 9, 10), the arm registration (nodes 636 1, 2, 5, 6, 11, 12) will not trigger leg node activation since these 637 nodes belong to separate subgraphs, thus avoiding unneces-638 sary computations. A comparison between adaptive and non-639 adaptive registration is presented in Fig. 15: our adaptive kine-640 matics produce better results even though the registration poten-641 tial forces are the same in both cases. In this example, the over-642 all computational time was about 3 minutes. Since most of it 643 was spent on closest point search, we did not notice significant 644 computational gains between the adaptive and non-adaptive cases.

#### 645 7. Conclusion and Perspectives

We introduced a novel method for the run-time adaptivity 647 of elastic models. Our method requires few pre-processing (a 648 few seconds) contrary to existing model reduction techniques 649 based on modal analysis and system training. Nodes are sim-650 plified as soon as their velocities can be described by nodes at 651 coarser levels of details, otherwise they are made independent. 652 Linear interpolation is particularly suited for linear materials 653 and affine deformations as it provides the static solution; there-654 fore no refinement occurs except if inertia produces large ve-655 locity gradients. In non-linear deformation such as bending and 656 twisting, new nodes are active to approximate the solution in 657 terms of velocity. Using frames as kinematic primitives allows

Seene	Timing	#Steps (dt)	#Frames			#Integration Points				Speedup	
Scene	including collisions		total	min	max	mean	total	min	max	mean	including collisions
Christmas Tree (Fig. 1)	5-270 ms/frame	380 (0.04s)	36	1	31	9	124	124	124	124	×1.5
Cantilever Beam (Fig. 8)	<1-110 ms/frame	370 (0.5s)	15	1	15	1.8	164	3	164	20	×2
Mushroom Field (Fig. 12a)	75-200 ms/frame	200 (0.1s)	156	1	11	5.4	251	78	88	84	×2.1
Armadillo Salad (Fig. 12b)	650-1,200 ms/frame	1,556 (0.01s)	1,800	18	1,784	365	27,162	108	27,086	5,011	×3
Ball Stack (Fig. 13)	100-250 ms/frame	20 (0.1s)	50	1	38	11.5	407	70	360	178	×1.7

Table 2: Adaptivity performances and timings.



Figure 14: Deformable mesh registration of an adaptive deformable model *(blue)* to a target mesh *(red)*. Left to right: while initially fully coarsened, the deformable model is progressively refined, automatically producing increasingly local deformations.



Figure 15: Comparison of deformable registration results for adaptive *(top)* and non-adaptive *(bottom)* kinematics. Our coarse-to-fine scheme produces a tighter fit, given the same registration potential forces.

<sup>658</sup> simplifications in deformed configurations based on local coor-<sup>659</sup> dinates, which is not possible in traditional Finite Element or <sup>660</sup> particle-based techniques. Various distance metrics can be eas-<sup>661</sup> ily implemented to tune the adaptivity criterion depending on <sup>662</sup> the simulation context (*e.g.* physical, visual precision). Reduc-

<sup>663</sup> ing the number of independent DOFs speeds up the simulation,
<sup>664</sup> although the factor depends on the choice of the solver (*e.g.* it<sup>665</sup> erative/direct solver, collision response method), and on the
<sup>666</sup> simulation scenario (*e.g.* presence of steady states, local/global,
<sup>667</sup> linear/non-linear deformations, mass distributions). In addition
<sup>668</sup> to kinematic adaptivity, we presented a method to merge inte<sup>669</sup> gration points to speed up the computations even more of elas<sup>670</sup> tic internal forces. Force offsets are used to remove disconti<sup>671</sup> nuities between the levels of detail. Finally, we presented an
<sup>672</sup> application of this scheme to the deformable mesh registration
<sup>673</sup> problem, for which a coarse-to-fine deformation strategy can be
<sup>674</sup> very easily implemented.

In future work, we will address the question of stiffness discrotinuities and the design of scenario-dependent frame hierarcrotic chies. We will also perform a more in-depth analysis and evalucross ation of our adaptive scheme in the context of mesh registration. Finally, the presented technique is likely to be generalizable to non-frame kinematic DOFs, which could provide a basis for a fully generic adaptivity framework.

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